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PERFORMANCE OF TRANSONIC FAN STAGE
WITH WEIGHT FLOW PER UNIT ANNULUS AREA
OF 208 KILOGRAMS PER SECOND PER
SQUARE METER (42.6 (LB/SEC)/FT<sup>2</sup>)

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# PERFORMANCE OF TRANSONIC FAN STAGE WITH WEIGHT FLOW PER UNIT ANNULUS AREA OF 208 KILOGRAMS PER SECOND PER SQUARE METER (42. 6 (LB/SEC)/FT<sup>2</sup>)

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## SUMMARY

Aerodynamic performance was obtained for a 50-centimeter-diameter single-stage axial-flow transonic compressor designed for a high weight flow per unit annulus area of 208 kilograms per second per square meter at a rotor-blade tip speed of 425 meters per second. Performance data were taken over the stable operating flow range of the stage at equivalent rotative speeds from 50 to 100 percent of design speed.

Peak efficiency for both rotor and stage occurred at an equivalent weight flow of 30.2 kilograms per second as compared with the design value of 31.0. Peak efficiency values of 0.83 and 0.79 were obtained for the rotor and stage, respectively.

The stall margin for the stage was 23 percent, based on equivalent weight flow and total-pressure ratio at peak efficiency and stall.

Rotor blade-element performance indicated that the outer 50 percent of the blade was choked at design weight flow and that high rotor losses occurred in the region between the blade damper and the hub.

Stator losses at the overall equivalent weight flow corresponding to stage peak efficiency are approximately equal to design values across the entire blade span except in the regions of the hub and behind the rotor damper.

#### INTRODUCTION

The NASA Lewis Research Center is engaged in a research program on axial-flow fans and compressors for advanced air-breathing engines. The program is directed primarily toward reducing the size and weight of the fans and compressors while maintaining a high level of performance.

In support of this program experimental studies are being conducted to investigate the effect of weight flow per unit annulus area on efficiency and stall margin. A series of three transonic compressor stages were designed for weight flows per unit annulus area values of 178, 198, and 208 kilograms per second per square meter. All three stages have a design pressure ratio of 1.57, and all three stages use the same flow path geometry. Two of the stages in this series, having design weight flow per unit annulus area of 178 and 198 kilograms per second per square meter, were tested earlier, and the results are presented in references 1 and 2.

This report presents the aerodynamic design and experimental performance of a single-stage axial-flow transonic compressor designed for a weight flow per unit annulus area of 208 kilograms per second per square meter. This stage is designated "stage 17-12" (rotor 17 - stator 12). The overall performance of both rotor and stage and the blade-element performance of both rotor and stator are presented. The data are presented over the stage stable operating flow range at rotative speeds from 50 to 100 percent of design speed. The design rotor-blade tip speed is 425 meters per second. Surveys of the flow conditions were taken at 11 radial positions. The tests were conducted in the single-stage compressor facility at Lewis.

#### AERODYNAMIC DESIGN

Three computer programs were used in the design of this compressor stage: the streamline analysis program, the blade geometry program, and the blade coordinate program. These programs are described in detail in references 3 and 4; only a brief description of each is presented in this report.

The streamline analysis program was used to calculate the flow field parameters at several axial locations, including planes approximating the blade leading and trailing edges for both the rotor and stator. The weight flow, rotative speed, flow path geometry, and radial distributions of total pressure and temperature are inputs in this program. The program accounts for both streamline curvature and entropy gradients; boundary-layer blockage factors are also included.

The distributions of velocity, total pressure, and total temperature calculated in the streamline analysis program are used in the blade geometry program to compute blade geometry parameters. The total loss for this rotor blade was based on the experimental rotor loss data presented in reference 3. The profile loss was then estimated by subtracting a calculated shock loss from the total loss. The shock loss calculation was based on the method presented in reference 5. Because of the stringent design requirements for this rotor, the forward portion of the transonic compressor blade over most of the blade span had considerable camber, resulting in high shock loss values. Thus,

the design profile losses over most of the blade span so computed were unrealistically small. Profile loss for the stator was based on the data presented in reference 6.

The blade geometry parameters are used in the blade coordinate program (ref. 7) to compute blade elements on conical surfaces passing through the blade. In this program the blade elements are then stacked on a line passing through their centers of gravity, and Cartesian blade coordinates, which are used directly in fabrication, are computed.

The overall design parameters for stage 17-12 are listed in table I, and the flow path is shown in figure 1. This stage was designed for an overall pressure ratio of 1.57 at a weight flow of 31.0 kilograms per second  $(208 \, (kg/sec)/m^2)$  of annulus area). The design tip speed was 425 meters per second. The rotor and stator were designed for a tip solidity of 1.3. The rotor had 43 blades with an aspect ratio of 2.4, and the stator had 48 blades with an aspect ratio of 2.3.

The blade-element design parameters for rotor 17 are presented in table II. This rotor was designed for a radially constant total pressure ratio of 1.60. The stator blade-element design parameters are given in table III. The blade geometry is presented in table IV for rotor 17 and in table V for stator 12. Both the rotor and stator have multiple-circular-arc blade shapes.

The symbols and equations are given in appendixes A and B. The abbreviations and units used for the tabular data are presented in appendix C.

#### APPARATUS AND PROCEDURE

#### Compressor Test Facility

The compressor stage was tested in the Lewis single-stage compressor facility (described in detail in ref. 4). A schematic diagram of the facility is shown in figure 2. Atmospheric air enters the test facility through an inlet located on the roof of the building and flows through the flow measuring orifice and into the plenum chamber upstream of the test stage. The air then passes through the experimental compressor stage into the collector and is exhausted to the atmosphere.

#### Test Stage

Photographs of the rotor and stator are shown in figures 3 and 4. The rotor blades have vibration dampers located at about 50 percent span. The maximum thickness of the damper is 0.215 centimeter. The nonrotating radial tip clearance of the rotor was a

nominal 0.05 centimeter at ambient conditions. The axial spacing between the rotor hub trailing edge and the stator hub leading edge was 2.80 centimeters.

#### Instrumentation

The compressor weight flow was determined from measurements on a calibrated thin-plate orifice that was 38.9 centimeters in diameter. The temperature at the orifice was determined from an average of two Chromel Alumel thermocouples. Orifices pressures were measured by calibrated transducers.

Radial surveys of the flow were made upstream of the rotor, between the rotor and stator, and downstream of the stator. Photographs of the survey probes are shown in figure 5. Total pressure, total temperature, and flow angle were measured with the combination probe (fig. 5(a)), and the static pressure was measured with an 8° C-shaped wedge probe (fig. 5(b)). Each probe was positioned with a null-balancing, stream-directional sensitive control system that automatically alined the probe to the direction of flow. The thermocouple material was iron/constantan. The probes were calibrated in an air tunnel. A combination probe and wedge static probe were used at each of the three measuring stations.

Inner- and outer-wall static-pressure taps were located at the same axial stations as the survey probes. The circumferential locations of both types of survey probes along with inner- and outer-wall static pressure taps are shown in figure 6. The combination probes downstream of the stator (station 3) were circumferentially traversed one stator blade passage  $(7.5^{\circ})$  counterclockwise from the nominal values shown.

An electronic speed counter, in conjunction with a magnetic pickup, was used to measure rotative speed (in rpm).

The estimated errors of the data based on inherent accuracies of the instrumentation and recording system are as follows:

Flow rate, kg/sec±0.3
Rotative speed, rpm
Flow angle, deg
Temperature, K
Rotor inlet total pressure, $N/cm^2$
Rotor outlet total pressure, $N/cm^2$ ±0.10
Stator outlet total pressure, $N/cm^2$
Rotor inlet static pressure, $N/cm^2$
Rotor outlet static pressure, N/cm <sup>2</sup> ±0.07
Stator outlet static pressure, N/cm <sup>2</sup>

At each measuring station the integrated weight flow is computed based on the radial survey data. An indication of the consistency of the data can be observed by comparing integrated weight flows at each of the measuring stations to the orifice weight flow in table VI.

#### Test Procedure

The stage survey data were taken over a range of weight flows from maximum flow to the near-stall conditions. At 70, 90, and 100 percent of design speed, radial surveys were taken over the whole flow range of the compressor. At 50, 60, and 80 percent of design speed, radial surveys were taken for the near-stall weight flow only. Data were recorded at 11 radial positions for each speed and weight flow.

At each radial position the combination probe behind the stator was circumferentially traversed to nine different locations across the stator gap. The wedge probe was set at midgap because previous studies showed that the static pressure across the stator gap was constant. Values of pressure, temperature, and flow angle were recorded at each circumferential position. At the last circumferential position values of pressure, temperature, and flow angle were also recorded at stations 1 and 2. All probes were then traversed to the next radial position, and the circumferential traverse procedure was repeated.

At each of the six rotative speeds the back pressure on the stage was increased by closing the sleeve valve in the collector until a stalled condition was detected by a sudden drop in stage outlet total pressure. This pressure was measured by a probe located at midpassage and was recorded on an X-Y plotter. Stall was corroborated by large increases in the measured blade stresses on both rotor and stator along with a sudden increase in noise level.

### Calculation Procedure

Because of the physical construction of the C-shaped static-pressure wedges, it was not possible to obtain static-pressure measurements at 5, 10, and 95 percent of span. The static pressure at 95 percent span was obtained by assuming a linear variation in static pressure between the values at the inner wall and the probe measurement at 90 percent span. A similar variation was assumed between the static-pressure measurements at the outer wall and the 30 percent span to obtain the static pressure at 5 and 10 percent span.

At each radial position, averaged values of the nine circumferential measurements of pressure, temperature, and flow angle downstream of the stator (station 3) were ob-

tained. The nine values of total temperature were mass averaged to obtain the stator outlet total temperature presented. The nine values of total pressure were energy averaged. The measured values of pressure, temperature, and flow angle were used to calculate axial and tangential velocities at each circumferential position. The flow angles presented for each radial position are calculated based on the circumferential massaveraged angular momentum and the average axial velocity. To obtain the overall performance, the radial values of total temperature were mass averaged, and the values of total pressure were energy averaged.

The data, measured at the three measuring stations, have been translated to the blade leading and trailing edges by the method presented in reference 3.

The weight flow at stall was obtained in the following manner: During operation at the near-stall condition, the collector valve was slowly closed in small increments. At each increment the weight flow was obtained. The weight flow obtained just before stall occurred is called the stall weight flow. The pressure ratio at stall was obtained by extrapolating the total pressure obtained from the survey data to the stall weight flow.

Orifice weight flow, total pressures, static pressures, and temperatures were all corrected to standard sea-level conditions based on the rotor-inlet conditions.

#### RESULTS AND DISCUSSION

The results from this investigation are presented in three main sections. The overall performances for the rotor and the stage are presented first. Radial distributions of several performance parameters are then presented for the rotor and stator. Finally, the blade-element data are presented for both the rotor and stator. The data presented are computer plotted; occasionally, a data point falls outside the range of parameters shown in the figure and is omitted.

All of the plotted data and some additional performance parameters are presented in tabular form. The overall performance data are presented in table VI (p. 25). The blade-element data are presented for the rotor in tables VII (p. 27) and for the stator in tables VIII (p. 47). The definitions and units used for the tabular data are presented in appendix C.

#### Overall Performance

The overall performance for rotor 17 and for stage 17-12 are presented in figures 7 and 8. For both of these computer plotted figures, data are presented for speeds from 50 to 100 percent design speed. For the 50, 60, and 80 percent of design speeds, the

overall performance is presented for the near-stall condition only. For the 70, 90, and 100 percent of design speeds, data are presented at several weight flows from choke to the near-stall conditions. Design point values are shown as solid symbols on both figures. The stall lines (dashed lines) shown in figures 7 and 8 were determined using the method discussed in the section Calculation Procedure.

In figures 7 and 8 averaged values of total pressure ratio, total temperature ratio, and temperature rise efficiency are plotted as functions of equivalent weight flow. At a near design weight flow of 30.9 kilograms per second (207.4 (kg/sec)/m<sup>2</sup> of annulus area), the stage experimental overall temperature-rise efficiency of 0.78 was 7 percentage points lower than the value based on design losses of 0.85. The experimental stage pressure ratio of 1.45 was lower than the design value of 1.57. Peak efficiency for the stage at the design tip speed of 425 meters per second was 0.79 and occurred at an equivalent weight flow of 30.2 kilograms per second. Stage pressure ratio at the peak efficiency point was 1.53.

The rotor experimental overall temperature-rise efficiency of 0.81 at design speed and near the design weight flow of 30.9 kilograms per second was 8 percentage points lower than the design value of 0.89. The rotor total pressure ratio was 1.49, 0.11 lower than the design value of 1.60. Peak efficiency for the rotor was 0.83, and it occurred at the equivalent weight flow of 30.2 kilograms per second. Sources of the relatively high losses in the rotor will be discussed in the blade-element performance sections that follow.

Stall margin for the stage was 23 percent based on equivalent weight flow and total pressure ratio at which peak efficiency occurred as compared with the values at stall.

#### Radial Distributions

The radial distributions of selected flow and performance parameters for both rotor and stator are shown in figures 9 and 10. The results are presented for three weight flows at design speed. The data shown represent the flow conditions at near stall, peak efficiency, and choke. The design values are shown by solid symbols. In this section performance results at the weight flow corresponding to peak efficiency are compared with design.

Rotor. - The total pressure ratio (fig. 9) is greater than design in the blade tip region and less than design in the region between the damper and the hub. The total-temperature ratio distribution agrees well with design except in the blade tip region where higher values were recorded. The temperature-rise efficiency is less than design across the entire blade span, particularly in the damper region where efficiency is as much as 20 points lower than design. The total loss coefficient distribution shows that

the losses are greater than design across the entire blade, especially in the region between the damper and the hub.

Large values of total loss in the damper region are due to both high shock and high profile losses. Design requirements necessitated an excessively high blade curvature over the leading edge portion of the blade suction surface where the flow is supersonic, resulting in high shock losses. Profile losses in the damper region (table VII) are higher than those observed on several rotors reported in reference 8. These higher profile losses may be attributed to the higher flow rate per unit annulus area and possible deviations in streamlines between peak efficiency and design conditions, resulting in large damper wakes.

Stator. - The total loss coefficient distribution (fig. 10) shows the losses are approximately that of design with exceptions at the hub and rotor damper regions. The loss distribution in the damper region is attributed to a streamline shift through the stator. From the overall performance (figs. 7 and 8) the stator performance at the weight flow corresponding to peak efficiency agrees well with design.

Deviation angles were less than design across the entire blade except in the hub and tip regions.

# Variation with Incidence Angle

The variations of selected rotor and stator blade-element flow and performance parameters with incidence angle are presented in figures 11 and 12. The data are given for 70, 90, and 100 percent of rotor design speed at blade elements located at 5, 10, 30, 45, 70, 90, and 95 percent of blade span as measured from the rotor tip trailing edge. Design values are shown by solid symbols. The incidence angle curves are presented primarily for future use in comparing the performance of these blades with others. Only a few brief observations will be made from the curves at present.

Rotor-blade. - The rotor blades were designed for zero incidence angle on the blade suction surface. Minimum loss values, over the range of incidence angles tested, were defined across the entire rotor-blade passage except in the hub region. At design speed the rotor-blade suction-surface incidence angles corresponding to minimum losses were within  $\pm 1^{\circ}$  of the design value over the entire span. The high losses recorded in the damper region are attributed to the large camber over the supersonic portion of the multiple-circular-arc blade, as was previously discussed. For all rotor blade elements at design incidence angle (zero degrees on blade suction surface) the experimental values of blade loading, as indicated by the diffusion factor, were lower than design values while the losses, as shown by the total loss coefficient, were greater than design. As design incidence is approached in the region of the blade between the tip and the dampers, blade

loading deteriorates rapidly, while meridional velocity ratio shows a sharp increase. Apparently the blade passage becomes choked over the outer 50 percent of the span at design incidence. Efficiency at the rotor tip is 8 percentage points less than the design value. Exceptionally high losses are noted in the damper region, resulting in efficiencies approximately 12 percentage points less than design at 45 percent of blade span at design incidence.

In summary it appears that the high measured losses for this rotor at the design condition are caused by a choked-flow condition existing in the region of the blade from the tip to the dampers, excessively high damper losses, and the large camber over the supersonic portion of the multiple-circular-arc blade.

Stator blade. - Minimum losses were defined at each radial station over the entire stator blade. Minimum values of total loss coefficient approximate design values except in the region behind the rotor damper and at the hub. Minimum loss occurred at incidence angles less than design values for all elements with the exceptions of the tip and rotor damper regions. Minimum loss incidence angles for all elements, with exceptions at 30 and 90 percent of blade span, occurred at an overall stage weight flow corresponding to stage peak efficiency. Stator deviation angles are within 20 of the design values at the design incidence angles of zero degree with exceptions at 70 percent of blade span and at the hub.

#### REMARKS

The transonic compressor stage presented in this report was part of a series of three stages in which the primary variable was weight flow. The stage reported herein was designated to have the highest weight flow of the three. In an attempt to isolate the effect of weight flow, the following design parameters were held constant for this series of rotors: rotor tip speed, flow path geometry, rotor blade loss, rotor solidity, and flow area allowances. The design constraints placed on this series of stages resulted in a penalty to this high weight flow stage; total-pressure ratio and efficiency were significantly lower than design values. The design constraints required that there be appreciably more turning in the forward portion of the rotor blade passage of the high weight flow stage than in the other two rotors. This turning apparently resulted in high local Mach numbers and considerably higher than design loss levels. If the design restrictions were eased to redistribute the turning through the blade passage, a high weight flow stage having reduced losses and better overall performance would probably result.

#### SUMMARY OF RESULTS

This report presents the aerodynamic design and the overall and blade-element performance of a 50-centimeter-diameter single-stage axial-flow transonic compressor designed for a high weight flow per unit annulus area value of 208 kilograms per second per square meter. Radial surveys of the flow conditions at the rotor inlet, rotor outlet, and stator outlet were made over the stable operating flow range of the stage equivalent rotative speeds from 50 to 100 percent of design speed. Flow and performance parameters were calculated across a number of selected blade elements. The following principal results were obtained:

- 1. At the design tip speed of 425 meters per second, peak efficiency for both rotor and stage occurred at an equivalent weight flow of 30.2 kilograms per second as compared with the design value of 31.0 kilograms per second (208  $(kg/sec)/m^2$ ) of annulus area.
- 2. For the stage the peak efficiency of 0.79 was 6 percentage points lower than the design value of 0.85. At the near design weight flow of 30.9 kilograms per second, the overall efficiency of 0.78 was 7 percentage points less than design.
- 3. For the rotor the peak efficiency of 0.83 was 6 percentage points lower than the design value of 0.89. At the near design weight flow of 30.9 kilograms per second, the overall efficiency of 0.81 was 8 percentage points less than design.
- 4. The stall margin for the stage was 23 percent, based on equivalent weight flow and total pressure ratios at peak efficiency and stall.
- 5. The experimental radial distribution of total loss coefficient for the rotor shows that high losses occurred in the region between the rotor damper and the hub. These high losses are attributed to both the damper and to the large camber of the supersonic portion of the multiple-circular-arc blade.
- 6. Stator losses at the overall equivalent weight flow corresponding to stage peak efficiency are approximately equal to design except in the regions of the hub and rotor damper.
- 7. Rotor blade-element performance indicates that the blade passage over the outer half of the rotor was choked at the overall design equivalent weight flow of 31.0 kilograms per second.
- 8. Minimum values of total loss coefficient for the stator approximate design values except in the regions of the hub and behind the rotor damper.

Lewis Research Center,

National Aeronautics and Space Administration, Cleveland, Ohio, July 11, 1973, 501-24.

# APPENDIX A

# SYMBOLS

A <sub>an</sub>	annulus area at rotor leading edge, 0.147 m <sup>2</sup>
${f A_f}$	frontal area at rotor leading edge, 0.198 m <sup>2</sup>
$C_{\mathfrak{p}}$	specific heat at constant pressure, 1004 J/(kg)(K)
c	aerodynamic chord, cm
D	diffusion factor
g	acceleration of gravity, 9.8 m/sec <sup>2</sup>
<sup>i</sup> mc	mean incidence angle, angle between inlet air direction and line tangent to blade mean camber line at leading edge, deg
<sup>i</sup> ss	suction surface incidence angle, angle between inlet air direction and line tangent to blade suction surface at leading edge, deg
J	mechanical equivalent of heat
N	rotative speed, rpm
P	total pressure, N/cm <sup>2</sup>
p	static pressure, N/cm <sup>2</sup>
r	radius, cm
SM	stall margin
T	total temperature, K
U	wheel speed, m/sec
v	air velocity, m/sec
W	weight flow, kg/sec
Z	axial distance references from rotor blade hub leading edge, cm
$^{lpha}{ m c}$	cone angle, deg
$\alpha_{\mathbf{s}}$	slope of streamline, deg
β	air angle, angle between air velocity and axial direction, deg
$eta_{f c}^{f '}$	relative meridional air angle based on cone angle, arctan(tan $\beta_{\rm m}^{\rm t}$ cos $\alpha_{\rm c}/{\cos\alpha_{\rm s}}$ ), deg
γ	ratio of specific heats

- ratio of rotor inlet total pressure to standard pressure of 10.13 N/m $^2$ . δ δΟ deviation angle, angle between exit air direction and tangent to blade mean camber line at trailing edge, deg θ ratio of rotor inlet total temperature to standard temperature of 288.2 K efficiency η angle between the blade mean camber line and the meridional plane, deg  $\kappa_{\rm mc}$ κ<sub>SS</sub> angle between the blade suction surface camber line at the leading edge and the meridional plane, deg solidity, ratio of chord to spacing σ  $\overline{\omega}$ total loss coefficient  $\overline{\omega}_{\mathbf{p}}$ profile loss coefficient shock loss coefficient Subscripts: ad adiabatic (temperature rise) id ideal LE blade leading edge meridional direction m mom momentum rise polytropic p TE blade trailing edge axial direction  $\mathbf{z}$ θ tangential direction 1 instrumentation plane upstream of rotor 2 instrumentation plane between rotor and stator instrumentation plane downstream of stator
- relative to blade

Superscript:

#### APPENDIX B

## **EQUATIONS**

Performance parameters are defined as follows:

Suction-surface incidence angle -

$$i_{SS} = (\beta_C^{\dagger})_{I.E} - \kappa_{SS}$$
 (B1)

Mean incidence angle -

$$i_{mc} = (\beta'_c)_{LE} - (\kappa_{mc})_{LE}$$
 (B2)

Deviation angle -

$$\delta^{O} = \left(\beta_{C}^{\prime}\right)_{TE} - \left(\kappa_{mc}\right)_{TE} \tag{B3}$$

Diffusion factor -

$$D = 1 - \frac{V_{TE}^{\dagger}}{V_{LE}^{\dagger}} + \left| \frac{\left( \mathbf{r} V_{\theta} \right)_{TE} - \left( \mathbf{r} V_{\theta} \right)_{LE}}{\left( \mathbf{r}_{TE} + \mathbf{r}_{LE} \right) \sigma(V_{LE}^{\dagger})} \right|$$
(B4)

Total loss coefficient -

$$\overline{\omega} = \frac{\left(P'_{id}\right)_{TE} - \left(P'\right)_{TE}}{\left(P'\right)_{LE} - \left(p\right)_{LE}}$$
(B5)

Profile loss coefficient -

$$\overline{\omega}_{p} = \overline{\omega} - \overline{\omega}_{s}$$
 (B6)

Total loss parameter -

$$\frac{\overline{\omega} \cos\left(\beta_{\rm m}'\right)_{\rm TE}}{2\sigma} \tag{B7}$$

Profile loss parameter -

$$\frac{\overline{\omega}_{p} \cos(\beta'_{m})_{TE}}{2\sigma}$$
 (B8)

Adiabatic (temperature-rise) efficiency -

$$\eta_{ad} = \frac{\left(\frac{P_{TE}}{P_{LE}}\right)^{(\gamma-1)/1} - 1}{\frac{T_{TE}}{T_{LE}} - 1}$$
(B9)

Momentum-rise efficiency -

$$\eta_{\text{mom}} = \frac{\left(\frac{P_{\text{TE}}}{P_{\text{LE}}}\right)^{(\gamma-1)/1} - 1}{\left(\frac{UV_{\theta}}{T_{\text{LE}}} - \left(\frac{UV_{\theta}}{T_{\text{LE}}}\right)\right)_{\text{LE}}} \tag{B10}$$

Equivalent weight flow -

$$\frac{\mathbf{W}\sqrt[4]{\theta}}{\delta} \tag{B11}$$

Equivalent rotative speed -

$$\frac{N}{\sqrt{\theta}}$$
 (B12)

Weight flow per unit annulus area -

$$\left(\frac{\mathbf{W}\sqrt{\theta}}{\delta}\right)/\mathbf{A}_{an}$$
 (B13)

Weight flow per unit frontal area -

$$\left(\frac{W\sqrt{\theta}}{\delta}\right)/A_{an}$$
 (B14)

Head-rise coefficient -

$$\frac{gJC_{p}T_{LE}}{U_{tip}^{2}}\left[\left(\frac{P_{TE}}{P_{LE}}\right)^{(\gamma-1)/\gamma}-1\right]$$
(B15)

Flow coefficient -

$$\left(\frac{V_{z}}{U_{tip}}\right)_{LE}$$
(B16)

Stall margin -

$$SM = \left[ \frac{\left(\frac{P_{TE}}{P_{LE}}\right)_{stall}}{\left(\frac{P_{TE}}{P_{LE}}\right)_{ref}} \times \frac{\left(\frac{W\sqrt{\theta}}{\delta}\right)_{ref}}{\left(\frac{W\sqrt{\theta}}{\delta}\right)_{stall}} - 1 \right] 100$$
(B17)

Polytropic efficiency -

$$\eta_{p} = \exp \left[ \frac{\left( \frac{P_{TE}}{P_{LE}} \right)^{(\gamma - 1)/\gamma}}{\frac{T_{TE}}{T_{LE}}} \right]$$
(B18)

#### APPENDIX C

#### DEFINITIONS AND UNITS USED IN TABLES

ABS absolute

AERO CHORD aerodynamic chord, cm

AREA RATIO ratio of actual flow area to critical area (where local Mach

number is one)

BETAM meridional air angle, deg

CONE ANGLE angle between axial direction and conical surface representing

blade element, deg

DELTA INC difference between mean camber blade angle and suction-

surface blade angle at leading edge, deg

DEV deviation angle (defined by eq. (B3)), deg

D-FACT diffusion factor (defined by eq. (B4))

EFF adiabatic efficiency (defined by eq. (B9))

IN inlet (leading edge of blade)

INCIDENCE incidence angle (suction surface defined by eq. (B1) and mean

defined by eq. (B2)), deg

KIC angle between blade mean camber line at leading edge and me-

ridional plane, deg

KOC angle between blade mean camber line at trailing edge and me-

ridional plane, deg

KTC angle between blade mean camber line at transition point and

meridional plane, deg

LOSS COEFF loss coefficient (total defined by eq. (B5) and profile defined by

eq. (B6))

LOSS PARAM loss parameter (total defined by eq. (B7) and profile defined by

eq. (B8))

MERID meridional

MERID VEL R meridional velocity ratio

OUT outlet (trailing edge of blade)

PERCENT SPAN percent of blade span from tip at rotor outlet

PHISS suction-surface camber ahead of assumed shock location, deg

PRESS pressure, N/cm<sup>2</sup>

PROF profile

RADII radius, cm

REL relative to blade

RI inlet radius (leading edge of blade), cm

RO outlet radius (trailing edge of blade), cm

RP radial position

RPM equivalent rotative speed, rpm

SETTING ANGLE angle between aerodynamic chord and meridional plane, deg

SOLIDITY ratio of aerodynamic chord to blade spacing

SPEED speed, m/sec

SS suction surface

STREAMLINE SLOPE slope of streamline, deg

TANG tangential

TEMP temperature, K

TI thickness of blade at leading edge, cm

TM thickness of blade at maximum thickness, cm

TO thickness of blade at trailing edge, cm

TOT total

TOTAL CAMBER difference between inlet and outlet blade mean camber lines,

deg

VEL velocity, m/sec

WT FLOW equivalent weight flow, kg/sec

X FACTOR ratio of suction-surface camber ahead of assumed shock loca-

tion of multiple-circular-arc blade section to that of double-

circular-arc blade section

ZIC axial distance to blade leading edge from inlet, cm

ZMC axial distance to blade maximum thickness point from inlet, cm

ZOC axial distance to blade trailing edge from inlet, cm

ZTC axial distance to transition point from inlet, cm

#### REFERENCES

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# TABLE I. - DESIGN OVERALL PARAMETERS

# FOR STAGE 17-12

ROTOR TOTAL PRESSURE RATIO,1.601
STAGE TOTAL PRESSURE RATIO
ROTOR TOTAL TEMPERATURE RATIO
STAGE TOTAL TEMPERATURE RATIO
ROTOR ADIABATIC EFFICIENCY
STAGE ADIABATIC EFFICIENCY
ROTOR POLYTROPIC EFFICIENCY
STAGE POLYTROPIC EFFICIENCY
ROTOR HEAD RISE COEFFICIENT
STAGE HEAD RISE COEFFICIENT0.221
FLOH COEFFICIENT
MT FLOW PER UNIT FRONTAL AREA155.182
MT FLOW PER UNIT ANNULUS AREA208.018
WT FLOW 30.958
RPM16100.000
TIP SPEED 424,858

TABLE II. - DESIGN BLADE-ELEMENT PARAMETERS FOR ROTOR 17

RP TIP 1 2 3 4 5 6 7 8 9 10 11 HUB	RADI IN 25.199 24.615 2 24.044 2 21.719 1 19.935 1 19.633 1 19.025 1 18.719 1 16.837 1 14.158 1 13.446 1 2.700	0UT 24.785 24.251 23.716 21.579 19.976 19.708 19.441 19.174 18.907 17.303 15.166 14.631	ABS IN 0. -0. 0. 0. 0. 0. 0. 0.	BETAM OUT 40.0 38.7 37.9 37.4 38.2 38.4 38.7 38.1 40.7 43.9 45.1	REL IN 65.4 64.3 63.3 59.6 57.0 56.6 55.2 55.4 52.8 49.1 48.0 46.7	BETAM 0UT 60.9 60.0 59.0 54.3 49.7 48.8 47.9 46.9 45.9 38.6 24.6 19.9 14.6		TEMP RATIO 1.192 1.183 1.176 1.156 1.156 1.156 1.155 1.155 1.153 1.153 1.154	TOTAL IN 10.13 10.13 10.13 10.13 10.13 10.13 10.13 10.13 10.13 10.13	PRESS RATIO 1.601 1.601 1.601 1.601 1.601 1.601 1.601 1.601 1.601 1.601 1.601
RP TIP 1 23 4 5 6 7 8 9 10 11 HUS	ABS 194.7 199.9 204.2 215.1 217.9 218.0 217.9 215.3 207.0 204.4 201.6	VEL 0UT 207.2 207.0 207.3 212.2 217.9 219.0 220.2 221.5 222.8 232.1 249.8 255.9 263.0	RELL IN 467.3 460.6 453.9 424.7 400.6 392.1 387.9 383.5 356.3 315.2 294.1	VEL OUT 326.0 322.6 317.9 289.1 264.7 250.4 252.3 248.2 225.1 197.0 187.2	MERII 1N 194.7 199.9 204.2 215.1 217.9 218.0 217.9 215.3 207.0 204.4 201.6	VEL OUT 158.8 161.4 163.6 168.6 171.1 171.5 172.0 172.4 172.8 175.9 179.9 180.6 181.2	TANO IN 0. -0. 0. 0. 0. 0.	VEL OUT 133.1 129.5 127.3 128.9 134.9 136.2 137.6 139.1 140.6 151.4 173.3 181.4	WHEEL 10 424.9 415.0 405.4 366.2 336.1 331.0 325.9 320.8 315.6 283.9 283.7 226.7 214.1	SPEED OUT 417.9 408.9 399.9 363.8 332.3 327.8 323.3 318.8 291.7 255.7 246.7 237.7
RP TIP 1 2 3 4 5 6 7 8 9 10 11 HUB	ABS M IN 0.592 0.609 0.623 0.659 0.669 0.669 0.669 0.668 0.652 0.614	ACH NO 0UT 0.576 0.578 0.580 0.599 0.621 0.625 0.625 0.623 0.633 0.759	REL M IN 1.421 1.403 1.385 1.301 1.226 1.203 1.190 1.176 1.092 0.965 0.931 0.896	ACH NO OUT 0.906 0.900 0.890 0.750 0.739 0.717 0.706 0.643 0.553 0.5540	MERID M IN 0.592 0.609 0.623 0.659 0.668 0.669 0.669 0.668 0.663 0.632 0.624	ACH NO OUT 0.441 0.450 0.458 0.476 0.487 0.488 0.490 0.491 0.502 0.502 0.523	STREAML II 1N -5.68 -5.04 -4.33 -0.79 2.33 2.33 2.46 4.04 4.64 8.62 15.51 17.69 20.16	NE SLOPE OUT -8.80 -7.62 -6.47 -2.14 0.96 1.48 2.53 3.06 6.41 11.51 12.93 14.41		PEAK SS MACH NO 1.584 1.571 1.562 1.553 1.553 1.553 1.554 1.556 1.597 1.675 1.633
R₽		INCI			D-FACT					PARAM

TABLE III. - DESIGN BLADE-ELEMENT PARAMETERS FOR STATOR 12

RP T1P 1 2 3 4 5 6 7 8 9 10 11 HUB	RAD I IN 24.394 2 23.903 2 23.438 2 21.542 2 20.098 2 19.615 1 19.331 1 17.676 1 15.736 1 14.643 1	OUT 24.384 23.882 23.427 21.604 20.233 20.005 19.778 19.550 19.323 17.973 16.216	ABS 1N 35.1 33.6 33.6 34.5 34.5 34.9 35.4 36.9 40.9 42.4	BETAM OUT 0. 0. 0. 0. 0. 0. 0.	REL IN 35.1 34.1 33.6 34.5 34.5 34.7 34.9 35.4 36.9 39.8 40.9 42.4	BETAM OUT 0. -0. 0. 0. 0. 0. 0. 0.	TOTAL IN 343.7 340.9 333.4 333.2 333.1 332.8 332.1 332.8 332.1 332.3 333.4	TEMP RATIO 1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000 1.000	TOTAL IN 16.22 16.22 16.22 16.22 16.22 16.22 16.22 16.22 16.22 16.22 16.22	PRESS RATIO 0.972 0.978 0.987 0.985 0.985 0.985 0.984 0.984 0.982 0.973 0.965
RP TIP 1 2 3 4 5 6 7 8 9 10 11 HUB	ABS IN 235.9 234.2 233.1 235.4 236.5 237.3 238.1 239.0 240.0 247.0 265.7 272.6	VEL 0UT 184.2 186.8 188.7 191.9 192.7 192.9 193.2 193.4 195.5 196.9 196.2	REL IN 235.9 234.2 233.1 235.4 236.5 237.3 239.0 240.0 247.0 260.9 265.7 272.6	VEL 0UT 184.2 186.8 188.7 191.9 192.6 192.7 193.2 193.4 195.5 197.2 196.2	MER II IN 193.0 193.8 194.3 194.4 194.9 195.2 195.4 195.7 197.6 200.8 201.3	VEL 0UT 184.2 186.8 188.7 191.9 192.6 192.7 193.2 193.4 195.5 197.2 196.2	TANO IN 135.6 128.8 129.1 134.1 135.2 137.6 139.0 148.2 167.0 174.0 183.9	O. O	WHEEL IN 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP TIP 1 2 3 4 5 6 7 8 9 10 11 HUB	ABS M IN 0.662 0.660 0.659 0.664 0.678 0.680 0.683 0.687 0.709 0.753	ACH NO 0.508 0.518 0.525 0.538 0.542 0.543 0.544 0.551 0.555 0.555 0.552	REL M IN 0.662 0.669 0.665 0.664 0.675 0.680 0.683 0.687 0.709 0.753 0.768	ACH NO 0UT 0.508 0.518 0.525 0.538 0.542 0.543 0.544 0.551 0.555 0.555	MERID M IN 0.542 0.546 0.553 0.555 0.5558 0.5559 0.560 0.568 0.579 0.588	NO OUT 0.508 0.518 0.525 0.538 0.541 0.543 0.544 0.554 0.556 0.555 0.555	STREAML1 IN -1.15 -0.80 -0.47 1.05 2.41 2.67 2.93 3.19 3.47 5.35 8.64 9.64	OUT -0.05 0.02 0.10 0.74 1.36 1.47 1.59 1.71 1.83 2.63 3.92		PEAK SS MACH NO 0.923 0.906 0.895 0.890 0.903 0.910 0.914 0.919 0.950 1.017 1.042 1.077
				0.552	0.505	0.552	11.00	4.65	0.975	1.077

TABLE IV. - BLADE GEOMETRY FOR ROTOR 17

RP TIP 1 23 4 5 6 7 8 9 10 11 HUB	PERCENT SPAN 0. 5. 10. 30. 45. 48. 50. 55. 70. 90. 95.	RI 25.199 24.615 24.044 21.719 19.935 19.635 19.330 19.025 18.719 16.837 14.158	R0 24.785 24.251 23.716 21.579 19.976 19.708 19.441 19.174 18.907 17.303 15.166 14.631	BLAI KIC 62.61 61.29 60.05 52.10 51.55 51.00 50.46 49.92 46.62 42.13 40.96	DE ANGLE KTC 61.26 60.07 58.79 52.67 47.65 46.77 45.88 44.98 44.05 37.38 20.47	55 K0C 56.06 55.54 54.91 51.81 47.44 46.59 44.59 43.54 36.27 22.17 16.66 10.26	DELTA INC 2.53 - 2.80 3.06 4.14 4.96 5.10 5.24 5.37 5.50 6.28 7.14 7.28	CONE ANGLE -10.696 -9.067 -7.870 -2.944 0.778 2.051 2.051 2.692 3.339 7.494 15.844 17.956
	BLADE	E THICKN			XIAL DII			
RP TIP 1 2 3 4 5 6 7 8 9 10 11 HUB	TI 0.051 0.051 0.051 0.051 0.051 0.051 0.051 0.051 0.051	TM 0.152 0.163 0.163 0.216 0.249 0.255 0.266 0.271 0.306 0.356 0.370	TO 0.051 0.051 0.051 0.051 0.051 0.051 0.051 0.051 0.051	Z1C 1.093 1.047 1.085 0.676 0.653 0.629 0.580 0.414 0.153 0.080	ZMC 2.147 2.148 2.150 2.150 2.149 2.149 2.148 2.148 2.148 2.148 2.119 2.096	ZTC 2.552 2.518 2.481 2.283 2.080 2.042 2.002 1.961 1.918 1.616 1.099 0.790	ZOC 3.285 3.326 3.369 3.544 3.690 3.715 3.741 3.767 3.794 4.198 4.255 4.311	
RP TIP 1 2 3 4 5 6 7 8 9 10 11 HUB	AERO CHORD 4.639 4.627 4.615 4.613 4.614 4.615 4.616 4.636 4.750 4.760 4.824	SETTING ANGLE 60.91 59.72 58.50 53.22 48.66 47.82 46.96 46.08 45.17 38.67 26.80 22.83 18.30		50LIDITY 1.300 1.326 1.356 1.493 1.619 1.667 1.692 1.718 1.902 2.255 2.374 2.521	X FACTOR 0.705 0.748 0.803 1.024 1.223 1.243 1.225 1.278 1.278 1.278 1.295 1.474 1.866 1.945	PHISS 4.83 4.94 5.19 7.26 9.17 9.51 10.50 10.50 13.59 18.86 20.13 21.40	AREA RAT10 1.041 1.040 1.040 1.040 1.040 1.040 1.040 1.039 1.025	

TABLE V. - BLADE GEOMETRY FOR STATOR 12

RP TIP 1 2 3 4 5 6 7 8 9 10 11 HUB	5. 10. 30. 45. 48. 50. 53. 55. 70. 90.	RI 24.394 23.903 23.438 21.542 20.098 19.856 19.615 19.373 19.131 17.676	R0 24.384 23.882 23.427 21.604 20.233 20.005 19.550 19.550 19.323 17.973 16.216 15.785	BLAI KIC 28.69 27.72 27.12 27.17 28.12 28.54 28.77 29.00 30.52 33.59 34.72 36.33	DE ANGLE KTC 24.28 23.70 23.36 23.65 24.73 24.92 25.12 25.34 26.69 26.69 31.63	ES K0C -9.14 -8.52 -8.11 -7.59 -7.38 -7.37 -7.37 -7.37 -7.36 -7.70 -7.91	DELTA INC 6.41 6.42 6.43 6.41 6.41 6.41 6.40 6.38 6.32 6.27	CONE ANGLE -0.150 -0.310 -0.152 0.915 1.994 2.196 2.404 2.620 2.843 4.394 7.111 7.903 8.900
	BLADE	THICKN	ESSES	A	KIAL DII	MENSION	ıs ·	
RP	T!	TM	TO	ZIC	ZMC	ZTC	ZOC	
TIP 1	0.051 0.051	0.279	0.051	7.051 7.042	8.902 8.904		10.933	
2	0.051	0.279	0.051	7.036	8.906		10.933	
3	0.051 0.051	0.279	0.051	7.035	8.907		10.933	
3 4 5 6 7	0.051	0.279	0.051	7.042 7.044	8.906 8.905	8.070	10.932	
6	0.051	0.279	0.051	7.045	8.905	8.059	10.932	
8	0.051 0.051	0.279 0.279	0.051	7.046 7.048	8.904 8.903		10.931	
9	0.051	0.279	0.051	7.060	8.900	8.017	10.929	
10 11	0.051 0.051	0.279 0.279	0.051	7.085 7.095	8.893 8.890		10.925	
HUB	0.051	0.279	0.051	7.110	8.886		10.922	
RP	AERO CHORD	SETTING	CAMBER	SOLIDITY	X FACTOR	PH155	AREA RATIO	
TIP	4.053	14.05	37.83	1.270	0.600	8.81	1,144	
1 2	4.053 4.053	13.51 13.18	36.24 35.23	1.296 1.321	0.600 0.600	8.23 7.83	1.137	•
3	4.053	13.14	34.68	1.435	0.600	7.29	1.117	
4	4.055 4.056	13.59 13.69	35.50 35.70	1.536 1.555	0.600 0.600	7.21 7.21	1.102	
5 6	4.056	13.79	35.91	1.573	0.600	7.21	1.097	
7 8	4.057 4.058		36.13 36.37	1.593 1.612	0.600	7.22 7.23	1.094	
9	4.058	14.74	37.88	1.742	0.600	7.26	1.071	
10	4.082		41.16	1.952	0.600	7.55		
11 HUB	4.088 4.098		42.42 44.24	2.015	0.600 0.600	7.70 7.95	1.038 1.030	

## TABLE VI. - OVERALL PERFORMANCE FOR STAGE 17-12

(a) 100 Percent of design speed

Parameter	Reading number							
	1143	1144	1145	1146	1147	1148	1179	
ROTOR TOTAL PRESSURE RATIO STAGE TOTAL PRESSURE RATIO ROTOR TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO STAGE TEMP. RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENCY ROTOR HEAD RISE COEFFICIENT STAGE HEAD RISE COEFFICIENT FLOW COEFFICIENT HT FLOW PER UNIT FRONTAL AREA HT FLOW PER UNIT ANNULUS AREA HT FLOW AT ROTOR INLET HT FLOW AT ROTOR INLET HT FLOW AT ROTOR OUTLET HT FLOW AT STATOR OUTLET ROTATIVE SPEED PERCENT OF DESIGN SPEED	1.435 1.363 1.137 1.137 0.796 0.694 0.782 0.174 0.148 0.443 155.40 208.30 31.04 31.90 32.40 16113.4	1.574 1.534 1.168 1.164 0.825 0.791 0.227 0.221 0.207 0.427 151.60 203.21 30.24 30.37 31.31 16138.4	1.623 1.574 1.182 1.180 0.817 0.767 0.221 0.402 145.11 194.51 28.95 29.98 29.56 16126.4	1.654 1.573 1.189 1.189 0.794 0.241 0.221 0.369 136.73 183.28 27.28 27.31 28.59 28.52 16128.4	1.624 1.553 1.195 1.195 0.762 0.693 0.761 0.237 0.214 0.343 129.27 173.28 25.79 25.79 25.78 27.18 27.57	1.618 1.541 1.196 0.750 0.675 0.747 0.236 0.211 0.332 125.69 25.07 25.09 26.51 16103.5	1.490 1.453 1.148 1.145 0.814 0.779 0.791 0.193 0.180 0.439 154.74 207.42 30.87 30.97 30.97 31.52 31.52 31.52	

(b) 90 Percent of design speed

Parameter	Reading number							
	1151	1152	1153	1154	1155	1156		
ROTOR TOTAL PRESSURE RATIO STAGE TOTAL PRESSURE RATIO ROTOR TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO ROTOR TEMP. RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENCY ROTOR HEAD RISE COEFFICIENT STAGE HEAD RISE COEFFICIENT HT FLOM PER UNIT FRONTAL AREA HT FLOM PER UNIT FRONTAL AREA HT FLOM AT ROTOR INLET HT FLOM AT ROTOR OUTLET HT FLOM AT ROTOR OUTLET HT FLOM AT STATOR OUTLET ROTATIVE SPEED PERCENT OF DESIGN SPEED	1.337 1.295 1.103 1.102 0.839 0.754 0.827 0.171 0.152 0.450 145.78 195.42 29.08 29.22 30.13 30.13	1.421 1.392 1.123 1.121 0.860 0.821 0.857 0.299 0.196 0.430 141,17 189,24 28,16 28,27 29,22 28,67 14485,7 90,0	1.462 1.426 1.136 1.133 0.840 0.227 0.211 0.399 133.25 26.58 26.68 27.78 27.15	1.473 1.432 1.145 1.142 0.804 0.762 0.806 0.231 0.213 0.367 124.93 24.93 24.93 24.93 24.93 14501.1	1,472 1,426 1,150 1,149 0,777 0,778 0,231 0,211 0,241 117,35 25,41 25,44 24,58 14503.5	1.472 1.413 1.160 1.158 0.752 0.658 0.726 0.231 0.205 0.306 107.43 144.00 21.43 21.43 21.43 21.43 21.43		

(c) 80 Percent of design speed

Parameter	Reading number
	1157
ROTOR TOTAL PRESSURE RATIO STACE TOTAL PRESSURE RATIO ROTOR TOTAL TEMPERATURE RATIO STACE TOTAL TEMPERATURE RATIO STACE TOTAL TEMPERATURE RATIO ROTOR TEMP. RISE EFFICIENCY STACE TEMP. RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENCY ROTOR HEAD RISE COEFFICIENT STACE HEAD RISE COEFFICIENT HTFLOM COEFFICIENT HTFLOM PER UNIT FRONTAL AREA HTFLOM PER UNIT FRONTAL AREA HTFLOM AT ROTOR UNLET HTFLOM AT ROTOR OUTLET HTFLOM AT ROTOR OUTLET HTFLOM AT STATOR OUTLET ROTATIVE SPEED PERCENT OF DESIGN SPEED	1.363 1.323 1.125 1.123 0.742 0.676 0.742 0.231 0.208 0.305 96.54 129.41 19.26 19.20 20.26 20.62 12891.2

## TABLE VI. - Concluded. OVERALL PERFORMANCE FOR

#### STAGE 17-12

# (d) 70 Percent of design speed

Parameter	Reading number					
	1173	1174	1175	1176		
ROTOR TOTAL PRESSURE RATIO STAGE TOTAL PRESSURE RATIO ROTOR TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO ROTOR TEMP. RISE EFFICIENCY STAGE TEMP. RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENCY ROTOR HEAD RISE COEFFICIENT STAGE HEAD RISE COEFFICIENT HIT FLOW PER UNIT FRONTAL AREA HIT FLOW PER UNIT FRONTAL AREA HIT FLOW AT ORTIFICE HIT FLOW AT ROTOR INLET HIT FLOW AT ROTOR OUTLET HIT FLOW AT SATOR OUTLET ROTATIVE SPEED PERCENT OF DESIGN SPEED	1.181 1.167 1.055 1.055 0.883 0.819 0.857 0.159 0.147 0.446 118.90 159.38 23.72 23.82 24.18 24.11 11281.9	1.225 1.210 1.068 1.067 0.871 0.859 0.193 0.193 0.408 110.33 147.90 22.01 22.07 22.49 22.30 11278.6	1.257 1.259 1.083 1.082 0.810 0.768 0.804 0.220 0.206 0.347 96.07 128.75 19.17 19.14 19.73 19.52 11283.5	1.267 1.241 1.094 1.092 0.744 0.690 0.744 0.250 0.209 0.302 84.33 113.05 16.79 17.18 17.82 11244.9 69.8		

# (e) 60 Percent of design speed

Parameter	Reading number
	1177
ROTOR TOTAL PRESSURE RATIO STAGE TOTAL PRESSURE RATIO ROTOR TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO ROTOR TEMP. RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENCY ROTOR HEAD RISE COEFFICIENT STAGE HEAD RISE COEFFICIENT FLOH COEFFICIENT HI FLOW PER UNIT FRONTAL AREA HIT FLOW PER UNIT ANNULUS AREA HIT FLOW AT ROTOR INLET HIT FLOW AT ROTOR OUTLET HIT FLOW AT STATOR OUTLET ROTATIVE SPEED PERCENT OF DESIGN SPEED	1.193 1.174 1.069 1.067 0.752 0.695 0.754 0.232 0.295 71.20 95.45 14.17 14.53 14.98 9607.0 59.7

## (f) 50 Percent of design speed

Parameter	Reading number
	1178
ROTOR TOTAL PRESSURE RATIO STAGE TOTAL PRESSURE RATIO ROTOR TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO STAGE TOTAL TEMPERATURE RATIO ROTOR TEMP. RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENCY ROTOR MOMENTUM RISE EFFICIENT STAGE MEAD RISE COEFFICIENT FLOM COEFFICIENT HIT FLOM PER UNIT FRONTAL AREA HIT FLOM PER UNIT ANNULUS AREA HIT FLOM AT ROTOR INLET HIT FLOM AT ROTOR UNLET HIT FLOM AT ROTOR OUTLET HIT FLOM AT STATOR OUTLET ROTATIVE SPEED PERCENT OF DESIGN SPEED	1.134 1.122 1.048 1.047 0.760 0.704 0.761 0.234 0.213 0.293 79.40 11.82 11.77 12.15 12.46 8064.8

TABLE VII. - BLADE-ELEMENT DATA AT BLADE EDGES FOR ROTOR 17

# (a) 100 Percent of design speed; reading number 1143

RP 1 2 3 4 5 6 7 8 9 10	RADII IN OUT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS BETAM IN OUT -0.3 29. -0.3 28. -0.3 27. -0.3 31. -0.3 32. -0.3 33. -0.3 33. -0.3 34. -0.3 35. -0.3 35.	IN OUT 9 64.2 57.7 6 63.0 56.6 8 59.9 53.2 5 57.1 47.5 2 56.7 47.8 6 56.2 49.7 3 55.7 51.1 2 55.2 49.3 8 52.4 38.1 4 48.3 23.0	TOTAL TEMP IN RAT [0 288.8 1.154 288.7 1.147 288.1 1.129 289.1 1.137 287.9 1.137 287.9 1.125 288.0 1.128 287.8 1.138 287.8 1.138	TOTAL PRESS [N RAT[0] 10.08 1.444 10.13 1.443 10.15 1.428 10.14 1.350 10.14 1.314 10.14 1.327 10.14 1.505
RP 1 2 5 4 5 6 7 8 9 10 11	ABS VEL IN OUT 201.2 218.4 207.0 221.1 212.7 220.8 217.8 232.1 218.5 226.7 219.1 214.0 219.5 203.9 219.8 209.3 219.5 243.1 213.6 276.9 210.1 272.6	REL VEL IN OUT 461.8 354.4 456.4 352.6 424.4 325.9 401.5 292. 397.7 285. 393.7 278. 385.5 265. 359.8 259. 321.2 245.2 310.3 235.	5 207.0 194.1 9 212.7 195.2 7 217.8 197.9 7 218.5 191.9 7 219.1 180.3 6 219.5 170.4 6 219.8 173.0 7 219.5 204.3 2 213.6 225.8	TANG VEL IN OUT -0.9 109.0 -1.0 105.9 -1.0 103.1 -1.0 121.3 -1.0 112.0 -1.0 117.8 -1.0 131.7 -0.9 160.4 -0.9 162.3	HHEEL SPEED IN OUT 414.8 408.7 405.7 400.2 366.3 364.0 336.3 337.0 331.2 332.5 326.0 327.9 320.9 323.4 315.8 318.9 284.2 292.0 259.1 256.1 227.4 247.5
RP 1	ABS MACH NO IN OUT - 0.612 0.619	REL MACH NO IN OUT 1,406 1.00	IN OUT 5 0.612 0.537		MERID PEAK SS VEL R MACH NO 0.941 1.570
2 3 4 5 6 7 8 9 10	0.632 0.630 0.651 0.635 0.668 0.651 0.671 0.651 0.673 0.614 0.674 0.584 0.675 0.600 0.674 0.705 0.654 0.812 0.643 0.798	1.392 1.00 1.299 0.93 1.232 0.84 1.221 0.82 1.209 0.80 1.196 0.77 1.184 0.76 1.105 0.75 0.984 0.71 0.949 0.68	7 0.651 0.561 0.668 0.569 0.671 0.551 0.673 0.517 0.674 0.488 0 0.675 0.496 5 0.674 0.593 0 0.654 0.662		0.938 1.561 0.918 1.563 0.908 1.558 0.878 1.557 0.776 1.557 0.787 1.557 0.931 1.592 1.057 1.684 1.042 1.670

# TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE

## **EDGES FOR ROTOR 17**

# (b) 100 Percent of design speed; reading number 1144

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN OUT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS BETAM IN OUT -0.3 39.8 -0.3 36.4 -0.3 36.1 -0.3 38.5 -0.3 42.6 -0.3 42.6 -0.3 43.6 -0.3 40.7 -0.2 42.0	REL BETAM IN OUT 65.0 56.5 63.8 55.9 60.8 53.0 58.2 47.5 57.3 48.1 56.9 49.1 56.5 48.3 53.9 38.3 50.0 24.7 49.0 21.6	TOTAL TEMP IN RATIO 288.8 1.210 288.6 1.188 288.1 1.165 288.0 1.162 288.2 1.161 287.9 1.161 287.9 1.163 287.9 1.154 287.8 1.150 287.9 1.151	TOTAL PRESS IN RATIO 10.06 1.642 10.13 1.588 10.14 1.557 10.14 1.491 10.14 1.533 10.14 1.573 10.14 1.573
RP 1 2 3 4 5 6 7 8 9 10	ABS VEL IN OUT 194.2 227.2 200.8 225.1 205.8 219.5 209.0 228.1 209.9 224.9 210.2 210.5 212.7 208.6 233.9 201.3 256.2 198.2 256.8	REL VEL 1N OUT 459.6 316.7 454.5 323.1 421.3 294.5 396.8 264.4 393.4 258.7 389.1 250.6 384.8 238.8 380.9 231.7 353.8 225.9 313.4 213.7 302.4 205.2	MERID VEL IN OUT 194.2 174.7 200.8 181.1 205.8 177.4 209.0 178.5 209.9 174.4 210.2 167.5 210.1 156.3 210.5 154.1 208.6 177.2 201.3 194.2 198.2 190.7	TANG VEL IN OUT -0.9 145.3 -1.0 133.7 -0.9 129.3 -0.9 142.1 -0.9 142.0 -0.9 141.9 -1.0 143.5 -1.0 146.7 -0.9 152.6 -0.8 167.2 -0.8 172.0	WHEEL SPEED IN OUT 415.6 409.5 406.8 401.2 366.7 364.4 337.1 331.7 333.0 326.5 328.4 0.316.5 319.7 284.8 292.7 239.4 256.5 227.5 247.5
RP 1 2 3 4 5 6 7 8 9	ABS MACH NO IN OUT 0.590 0.630 0.611 0.630 0.628 0.620 0.639 0.648 0.642 0.637 0.643 0.622 0.643 0.599 0.644 0.601 0.638 0.668 0.614 0.740 0.604 0.742	REL MACH NO IN OUT 1.395 0.878 1.384 0.904 1.286 0.832 1.213 0.750 1.203 0.753 1.190 0.710 1.177 0.674 1.165 0.654 1.082 0.645 0.956 0.617 0.921 0.592	MERID MACH NO IN OUT 0.590 0.484 0.611 0.507 0.628 0.501 0.639 0.507 0.642 0.494 0.643 0.441 0.644 0.435 0.638 0.506 0.614 0.561 0.604 0.551		MERID PEAK SS VEL R MACH NO 0.899 1.588 0.902 1.578 0.862 1.579 0.854 1.579 0.797 1.580 0.744 1.582 0.732 1.584 0.850 1.626 0.965 1.691 0.962 1.673
RP 12345678910	PERCENT INC SPAN MEAN 5.00 3.5 10.00 5.3 45.00 6.1 47.50 6.2 50.00 6.3 52.50 6.3 55.00 70.00 7.3 90.00 8.1 95.00 8.8	IDENCE SS 0.7 0.9 0.5 0.9 1.2 1.1 1.1 1.1 1.1 2.5 1.1 4.8 1.1 2.0 1.1 1.1 2.0 1.1 4.7	D-FACT EFF  0.430 0.725 0.398 0.797 0.404 0.856 0.445 0.859 0.453 0.859 0.466 0.798 0.491 0.750 0.505 0.733 0.477 0.842 0.441 0.921 0.447 0.891	LOSS COEFF TOT PROF 0.225 0.126 0.156 0.061 0.110 0.030 0.113 0.044 0.136 0.098 0.164 0.098 0.204 0.140 0.222 0.159 0.142 0.082 0.084 0.028 0.122 0.075	LOSS PARAM TOT PROF 0.047 0.026 0.032 0.013 0.022 0.006 0.024 0.009 0.028 0.014 0.033 0.020 0.039 0.027 0.043 0.031 0.029 0.017 0.017 0.006 0.024 0.015

TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE

# **EDGES FOR ROTOR 17**

# (c) 100 Percent of design speed; reading number 1145

# TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE

## **EDGES FOR ROTOR 17**

# (d) 100 Percent of design speed; reading number 1146

RP 1 2 3 4 5 6 7 8 9 10	RADII IN 01 24.615 24.2 24.044 23. 21.720 21.1 19.934 19. 19.632 19. 19.025 19. 19.025 19. 18.720 18.6 16.838 17. 14.158 15.	UT	BETAM OUT 45.5 43.9 42.4 45.7 46.7 48.4 50.2 52.1 47.6 44.0	REL IN 68.1 67.0 64.4 62.2 61.8 61.4 61.0 60.6 58.3 54.4	BETAM OUT 56.0 55.0 53.2 49.7 49.4 48.8 48.9 23.8 20.3	TOTAL IN 288.8 288.7 288.1 288.0 287.8 288.0 287.8 287.8 287.8	TEMP RATIO 1.235 1.188 1.182 1.183 1.184 1.184 1.186 1.171 1.161 1.159	TOTAL IN 10.06 10.11 10.14 10.15 10.15 10.15 10.15 10.15	PRESS RAT10 1.744 1.725 1.647 1.593 1.593 1.582 1.560 1.553 1.603
RP 1 2 3 4 5 6 7 8 9 10 11	167.5 236 172.4 233 176.1 219 178.1 219 178.4 211 178.4 211 178.5 216 176.4 226 171.6 251	REL UT IN 4.2 449.3 2.3 441.4 9.2 407.1 9.1 381.5 8.2 377.7 7.8 372.7 5.5 368.0 6.2 364.0 3.0 335.6 1.9 295.1 4.9 283.5	VEL 0UT 293.1 291.7 270.0 237.4 229.9 219.9 219.9 199.6 197.5 195.0 192.3	MERII IN 167.5 172.4 176.1 178.1 178.7 178.4 178.5 176.4 171.6 169.1	O VEL OUT 164.1 167.4 161.9 153.6 149.6 144.7 137.9 153.6 178.5 180.3	TAN 1N -0.8 -0.8 -0.8 -0.8 -0.8 -0.8 -0.8 -0.8	VEL 0UT 167.1 161.0 147.8 156.2 158.8 162.8 165.6 170.6 168.5 177.8 180.2	WHEEL IN 416.1 405.5 366.2 336.6 332.0 326.5 321.1 316.4 284.7 239.4 226.9	SPEED OUT 410.0 400.0 363.8 357.3 353.2 528.4 323.6 519.6 292.5 256.4 246.9
	ABS MACH	NO REL M	ACH NO	MERID M	ACH NO			MEDIA	0514 00
RP 1 2 3 4 5 6 7 8 9 10 11	0.504 0.6 0.520 0.6 0.538 0.6 0.540 0.6 0.540 0.6 0.539 0.6 0.539 0.6 0.533 0.6	UT IN 644 1.352 643 1.350 613 1.250 614 1.154 611 1.142 610 1.128 603 1.113 604 1.00 645 1.015 723 0.891 733 0.855	0UT 0.806 0.807 0.755 0.666 0.644 0.616 0.587 0.558 0.559 0.559	1N 0.504 0.520 0.532 0.538 0.540 0.540 0.539 0.533 0.518 0.510	0.451 0.453 0.453 0.453 0.419 0.405 0.386 0.3871 0.435 0.518				PEAK SS MACH NO 1.653 1.637 1.651 1.664 1.671 1.672 1.678 1.745 1.713

TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE

# **EDGES FOR ROTOR 17**

# (e) 100 Percent of design speed; reading number 1147

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN 0UT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS BETAM IN OUT -0.3 45.8 -0.3 45.8 -0.3 48.9 -0.3 52.1 -0.3 52.1 -0.3 55.4 -0.3 55.4 -0.3 49.4 -0.2 44.8	64.0 50.4 63.7 50.2 63.4 50.2 62.9 50.2 62.6 49.7 60.1 39.5 56.1 23.8	TOTAL TEMP IN RATIO 289.4 1.250 289.5 1.231 288.0 1.193 288.0 1.191 287.6 1.190 287.7 1.190 287.7 1.190 287.7 1.174 287.6 1.160 287.6 1.157	TOTAL PRESS IN RATIO 10.06 1.736 10.10 1.728 10.14 1.622 10.15 1.567 10.15 1.567 10.15 1.537 10.15 1.550 10.15 1.619 10.14 1.602
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL IN OUT 154.9 236.3 157.6 235.1 162.7 215.9 164.8 217.2 164.3 215.2 164.7 213.9 164.6 214.0 163.8 225.7 161.4 252.3 159.4 254.2	REL VEL 1N OUT 444.3 281.4 435.7 283.3 401.8 261.6 375.9 224.9 371.5 216.7 366.4 206.8 362.0 197.5 357.5 188.1 329.0 190.5 289.0 197.3 277.9 192.6	MERID VEL IN OUT 154.9 157.4 157.6 163.8 162.7 153.2 164.8 143.2 164.8 138.8 164.3 132.3 164.7 126.5 164.6 121.6 163.8 147.0 161.4 180.5 159.4 180.3	TANG VEL IN OUT -0.8 176.3 -0.7 168.7 -0.8 152.1 -0.8 164.4 -0.8 167.1 -0.8 169.7 -0.7 172.5 -0.8 176.1 -0.7 171.2 -0.7 176.3 -0.7 179.2	WHEEL SPEED OUT 415.7 409.5 405.4 399.9 366.6 364.3 337.1 337.8 322.2 333.5 326.8 328.7 321.7 324.2 316.5 319.7 284.6 292.4 239.0 265.1 227.0 247.0
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.464 0.645 0.473 0.647 0.489 0.602 0.496 0.606 0.496 0.601 0.496 0.597 0.496 0.597 0.496 0.537 0.486 0.724 0.480 0.731	REL MACH NO IN OUT 1.331 0.768 1.306 0.780 1.209 0.729 1.132 0.628 1.119 0.605 1.104 0.577 1.090 0.551 1.077 0.525 0.991 0.538 0.870 0.566 0.836 0.554	MERID MACH NO OUT 0.464 0.429 0.473 0.451 0.489 0.427 0.496 0.387 0.496 0.353 0.496 0.353 0.496 0.353 0.496 0.518 0.480 0.519		MERID PEAK SS VEL R MACH NO 1.016 1.683 1.039 1.675 0.942 1.693 0.869 1.710 0.842 1.715 0.805 1.720 0.768 1.724 0.739 1.729 0.897 1.789 1.118 1.722 1.131 1.696
RP 1 2 3 4 5 6 7 8 9 10 11	PERCENT INCI SPAN MEAN 5.00 8.1 10.00 8.6 30.00 10.7 45.00 11.9 47.50 12.1 50.00 12.4 52.50 12.5 55.00 12.7 70.00 13.6 90.00 14.1	DENCE SS 5.3 0.4 5.5 -0.3 6.6 2.3 6.9 3.0 7.0 3.6 7.1 4.6 7.1 5.6 7.2 6.2 7.3 3.2 7.0 1.5 7.0 3.7	D-FACT EFF  0.516  0.682 0.492  0.734 0.476  0.766 0.538  0.737 0.555  0.727 0.575  0.713 0.596  0.698 0.618  0.688 0.560  0.767 0.458  0.924	LOSS COEFF TOT PROF 0.307 0.197 0.250 0.145 0.212 0.118 0.255 0.169 0.268 0.183 0.286 0.202 0.305 0.222 0.305 0.222 0.324 0.170 0.097 0.046	LOSS PARAM TOT PROF 0.065 0.041 0.053 0.031 0.042 0.023 0.050 0.033 0.052 0.036 0.055 0.039 0.058 0.042 0.060 0.045 0.052 0.034 0.052 0.034

# TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE

# **EDGES FOR ROTOR 17**

# (f) 100 Percent of design speed; reading number 1148

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN 0UT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS SETAM IN OUT -0.3 49.8 -0.3 47.0 -0.3 50.6 -0.3 51.7 -0.3 55.3 -0.3 56.7 -0.3 49.5 -0.2 44.3	REL BETAM IN OUT 70.3 56.8 69.5 55.0 66.9 54.3 64.8 50.4 64.2 50.4 63.8 50.9 63.4 50.4 60.9 39.6 56.7 23.6 55.6 20.3	TOTAL TEMP  IN RATIO 289.4 1.252 289.3 1.233 288.1 1.197 287.8 1.194 287.6 1.193 287.1 1.190 287.9 1.190 287.6 1.173 287.6 1.173 287.6 1.158	TOTAL PRESS IN RATIO 10.06 1.717 10.09 1.719 10.14 1.618 10.15 1.568 10.15 1.557 10.15 1.536 10.15 1.527 10.15 1.527 10.15 1.621 10.14 1.605
RP 1 2 5 4 5 6 7 8 9 10 11	ABS VEL IN OUT 148.6 233.7 151.5 234.3 156.6 216.2 158.4 218.5 158.3 215.4 157.9 212.3 158.3 212.6 158.6 224.9 157.4 253.0 155.7 255.0	REL VEL 1N OUT 441.7 275.6 433.1 278.6 399.3 255.7 372.3 217.9 367.9 210.8 363.4 202.4 358.0 191.4 353.7 183.2 325.8 189.3 286.5 197.5 275.5 193.2	MERID VEL IN OUT 148.6 150.8 151.5 159.9 156.6 149.1 158.4 138.7 158.3 134.3 158.4 129.1 157.9 120.8 158.3 116.9 158.6 145.9 157.4 181.0 155.7 181.1	TANG VEL IN OUT -0.7 178.5 -0.7 171.3 -0.7 156.6 -0.7 168.8 -0.7 172.4 -0.7 172.4 -0.7 177.6 -0.7 177.6 -0.7 176.8 -0.6 179.4	WHEEL SPEED IN OUT 415.2 409.1 405.0 399.5 366.6 364.2 336.2 336.2 332.2 320.6 323.1 315.6 318.8 283.9 291.7 238.7 255.7 226.6 246.6
RP 1 2 5 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.444 0.637 0.453 0.644 0.470 0.601 0.476 0.609 0.476 0.601 0.475 0.593 0.476 0.593 0.476 0.593 0.477 0.635 0.473 0.726 0.468 0.734	REL MACH NO IN OUT 1.320 0.751 1.296 0.766 1.199 0.711 1.119 0.658 1.106 0.588 1.093 0.565 1.078 0.534 1.063 0.511 0.980 0.534 0.861 0.567 0.828 0.556	MERID MACH NO IN OUT 0.444 0.411 0.453 0.440 0.470 0.415 0.476 0.375 0.476 0.375 0.477 0.360 0.475 0.337 0.476 0.326 0.477 0.412 0.473 0.520 0.468 0.521		MERID PEAK SS VEL R MACH NO 1.015 1.699 1.056 1.691 0.952 1.711 0.876 1.729 0.848 1.736 0.815 1.741 0.765 1.746 0.738 1.750 0.92C 1.794 1.150 1.725 1.164 1.697
RP 1 2 3 4 5 6 7 8 9 10	PERCENT INCI SPAN MEAN 5.00 8.9 10.00 9.3 30.00 11.5 45.00 12.7 47.50 13.0 50.00 13.2 52.50 13.4 55.00 14.3 90.00 14.7 95.00 14.9	DENCE DEV SS 6.1 1.2 6.3 -0.0 7.3 2.5 7.8 3.0 7.9 3.9 7.9 4.8 8.0 6.3 8.0 6.8 8.0 3.2 7.6 1.2 7.6 3.4	D-FACT EFF  0.528  0.664 0.502  0.718 0.491  0.747 0.555  0.722 0.569  0.709 0.586  0.701 0.611  0.685 0.629  0.675 0.560  0.762 0.453  0.925 0.442  0.918	LOSS COEFF TOT PROF 0.327 0.214 0.269 0.162 0.234 0.138 0.276 0.188 0.292 0.204 0.303 0.215 0.322 0.236 0.337 0.252 0.263 0.179 0.097 0.047 0.113 0.072	LOSS PARAM TOT PROF 0.068 0.044 0.057 0.034 0.046 0.027 0.054 0.037 0.057 0.039 0.058 0.044 0.063 0.044 0.063 0.047 0.053 0.036 0.020 0.009 0.022 0.014

# TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE

# **EDGES FOR ROTOR 17**

# (g) 100 Percent of design speed; reading number 1179

RP 1 2 3 4 5 6 7 8 9 10 11	24.615 24.	OUT IN 252 0.0	OUT 34.4 33.2 31.5 35.5 36.3 37.3 38.6 39.6 36.1 39.6	1N 64.3 63.1 60.1 57.3 56.8 56.3 55.9 55.4 52.7 48.6	BETAM 001 57.4 56.3 53.1 46.7 47.2 49.8 51.2 48.6 37.9 23.0 20.5	TOTA 1N 288.7 288.6 288.1 288.0 288.1 288.0 288.7 287.9	L TEMP RATIO 1.172 1.162 1.142 1.151 1.149 1.143 1.140 1.143 1.143	TOTAL IN 10.07 10.13 10.14 10.14 10.14 10.14 10.15 10.14	PRESS RAT10 1.525 1.517 1.490 1.502 1.467 1.403 1.367 1.389 1.520
RP 1 2 3 4 5 6 7 8 9 10 11	199.4 22 205.1 22 211.1 21 216.3 21 217.0 22 217.3 21 217.5 20 217.6 21 216.6 21 210.2 26	CL RELL IN 100.6 460.6 454.0 9.7 423.0 153.3 400.2 17.5 396.0 15.0 387.9 0.8 383.3 19.2 357.1 166.9 307.3	VEL 0UT 337.7 334.2 312.0 277.4 269.9 261.3 252.9 245.5 243.5 230.1 219.5	MERII 1N 199.4 205.1 211.1 216.3 217.0 217.3 217.5 217.6 216.6 210.2 207.2	VEL 0UT 182.1 185.4 187.3 190.1 183.4 158.6 162.3 192.2 211.8 205.6	TAN IN 0.0 0.0 0.0 0.0 0.0 0.0 0.0	G VEL 0UT 124.6 121.5 114.7 135.4 134.5 128.5 126.7 134.5 142.3 165.9 170.1	HEELL [N 415.1 405.0 366.6 336.7 331.3 326.3 321.2 315.6 283.9 238.7 226.9	SPEED 0UT 409.0 399.5 364.3 337.4 332.6 328.2 323.7 318.8 291.7 255.8 246.9
RP 1 23 4 5 67 8 9 10	0.607 0.626 0.646 0.663 0.665 0.667 0.667 0.667 0.664 0.643 0.643 0.	NO REL I DUT IN 621 1.401 627 1.385 627 1.294 667 1.224 664 1.203 577 1.190 600 1.176 690 1.095 784 0.973 777 0.939	MACH NO OUT 0.950 0.945 0.891 0.793 0.771 0.744 0.719 0.699 0.702 0.671 0.639	MERID M 0.607 0.626 0.646 0.665 0.667 0.667 0.667 0.663 0.633	ACH NO .OUT 0.512 0.524 0.535 0.543 0.524 0.480 0.462 0.554 0.618 0.599			MERID VEL R 0.915 0.904 0.887 0.876 0.776 0.729 0.746 0.888 1.008	PEAK SS MACH NO 1,571 1,558 1,559 1,557 1,558 1,558 1,556 1,594 1,675
RP 1 2 3 4 5 6 7	PERCENT SPAN 5.00 10.00 30.00 45.00 47.50 50.00	INCIDENCE MEAN SS 2.9 0.1 2.9 -0.1 4.6 0.5 5.2 0.2 5.3 0.2 5.4 0.1	DEV 1.7 1.3 1.3 -0.7 0.7 4.2	D-FACT 0.368 0.362 0.353 0.411 0.422 0.432	0.744 0.779 0.852 0.817 0.774	LOSS C TOT 0.179 0.150 0.098 0.136 0.168 0.206	PROF 0.083 0.059 0.020 0.069	LOSS P TOT 0.036 0.031 0.020 0.029 0.035	PROF 0.017 0.012 0.004 0.015

TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE

# **EDGES FOR ROTOR 17**

(h) 90 Percent of design speed; reading number 1151

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN OUT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS BETAM IN OUT -0.3 25.0 -0.3 24.4 -0.3 28.0 -0.3 28.5 -0.3 28.5 -0.3 29.7 -0.3 30.8 -0.3 29.9 -0.2 33.9	64.2 57.0 62.9 56.3 59.8 53.0 57.1 46.8 56.6 46.1 56.1 46.4 55.6 47.0 55.2 45.8 52.3 37.1 48.3 24.0	TOTAL TEMP IN RATIO 288.7 1.109 288.6 1.103 288.0 1.095 288.1 1.104 288.0 1.105 287.6 1.102 288.3 1.102 288.1 1.104 287.9 1.101 287.9 1.108 287.8 1.111	TOTAL PRESS IN RATIO 10.06 1.331 10.12 1.318 10.13 1.307 10.14 1.339 10.14 1.337 10.14 1.291 10.15 1.358 10.15 1.358 10.15 1.392 10.14 1.408
RP 1234567891011	ABS VEL IN OUT 181.1 202.5 186.7 202.3 192.3 201.7 196.8 215.1 197.4 215.5 197.0 209.8 198.0 203.8 197.8 205.3 197.8 227.3 192.3 248.5 189.6 252.3	REL VEL IN OUT 415.8 336.8 410.4 305.6 362.0 277.5 358.4 272.9 353.6 266.7 350.6 259.4 346.3 252.8 323.5 247.0 288.9 225.8 279.1 219.2	MERID VEL IN OUT 181.1 185.6 186.7 184.2 192.3 185.7 196.8 189.8 197.4 189.3 197.0 187.8 198.0 177.0 197.8 176.4 197.8 197.1 192.3 206.3 189.6 205.7	TANG VEL IN OUT -0.9 85.5 -0.9 83.3 -0.9 101.1 -0.9 102.9 -0.9 101.2 -0.9 101.2 -0.9 105.0 -0.9 113.3 -0.8 138.4 -0.8 146.1	NHEEL SPEED IN OUT 373.4 367.9 364.6 359.7 329.6 327.5 302.9 303.5 298.3 299.4 292.8 294.5 288.5 290.8 285.1 262.1 214.8 230.1 204.0 222.0
RP 1 23 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.547 0.583 0.566 0.584 0.584 0.586 0.599 0.624 0.601 0.624 0.602 0.590 0.602 0.590 0.602 0.594 0.602 0.664 0.584 0.730 0.576 0.742	REL MACH NO IN OUT 1.257 0.970 1.243 0.959 1.162 0.887 1.101 0.806 1.091 0.775 1.067 0.751 1.054 0.732 0.985 0.722 0.878 0.663	MERID MACH NO 1N 0UT 0.547 0.529 0.566 0.552 0.599 0.551 0.601 0.550 0.600 0.534 0.602 0.512 0.602 0.512 0.602 0.516 0.584 0.606		MERID PEAK SS VEL R MACH NO 1.013 1.430 0.987 1.418 0.955 1.440 0.964 1.457 0.959 1.457 0.933 1.459 0.894 1.462 0.892 1.465 0.996 1.516 1.073 1.500
	0.576 0.742	0.847 0.644	0.576 0.604		1.085 1.486

#### **EDGES FOR ROTOR 17**

# (i) 90 Percent of design speed; reading number 1152

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN 0UT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.938 17.302 14.158 15.166 13.447 14.630	ABS BETAM IN OUT -0.3 33.5 -0.3 31.6 -0.3 31.8 -0.3 33.2 -0.3 34.8 -0.3 36.2 -0.3 37.2 -0.3 35.6 -0.2 39.9	63.9 55.6 60.9 52.9 58.2 47.6 57.8 46.5 57.4 45.7 56.9 45.0 53.8 37.6 50.1 24.0	TOTAL TEMP IN RATIO 288.8 1.144 288.8 1.133 288.0 1.118 288.0 1.119 287.9 1.121 288.1 1.123 287.8 1.123 287.8 1.123 287.9 1.115 287.8 1.119 287.8 1.119	TOTAL PRESS IN RATIO 10.07 1.453 10.11 1.445 10.14 1.408 10.14 1.419 10.14 1.414 10.15 1.393 10.15 1.385 10.14 1.407 10.13 1.448
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL IN OUT 174.7 204.3 179.2 203.8 183.9 198.3 187.8 207.4 188.5 209.1 188.5 209.2 188.7 204.6 187.2 217.2 180.5 236.1 177.8 243.3	REL VEL 1N 0UT 412.9 306.8 407.5 307.2 377.6 279.2 356.8 258.7 352.9 254.1 349.6 246.0 345.6 237.2 341.8 230.6 317.2 222.9 281.1 201.3 271.0 197.9	MERID VEL IN OUT 174.7 170.4 179.2 173.6 183.9 168.6 187.8 174.5 188.2 175.0 188.5 171.8 188.7 165.7 188.7 162.9 187.2 176.7 180.5 183.8 177.8 186.7	TANG VEL IN OUT -0.8 112.7 -0.9 106.8 -0.8 104.4 -0.9 112.1 -0.9 114.5 -0.9 121.2 -0.8 125.8 -0.8 126.4 -0.7 148.1 -0.7 156.1	WHEEL SPEED IN OUT 373.3 367.8 365.2 360.2 329.0 326.9 302.4 303.0 297.6 298.8 293.6 295.3 288.7 291.0 284.1 287.0 255.3 262.3 214.8 230.1 203.8 221.7
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.527 0.579 0.541 0.581 0.557 0.569 0.570 0.597 0.571 0.602 0.572 0.602 0.572 0.589 0.573 0.587 0.568 0.628 0.546 0.687 0.538 0.709	REL MACH NO IN OUT 1.245 0.870 1.231 0.875 1.144 0.801 1.082 0.744 1.070 0.737 1.061 0.707 1.049 0.681 1.037 0.662 0.962 0.645 0.851 0.585 0.820 0.577	MERID MACH NO		MERID PEAK SS VEL R MACH NO 0.976 1.447 0.969 1.441 0.917 1.462 0.929 1.482 0.930 1.485 0.912 1.492 0.878 1.496 0.863 1.503 0.943 1.531 1.018 1.506 1.050 1.487
RP 1 2 3 4 5 6 7 8 9 10	PERCENT INCI SPAN MEAN 5.00 3.5 10.00 3.7 30.00 5.4 45.00 6.1 47.50 6.2 50.00 6.4 52.50 6.5 55.00 6.5 70.00 7.3 90.00 8.1 95.00 8.3	DENCE DEV SS 0.7 0.6 0.7 0.6 1.3 1.0 1.2 0.1 1.1 -0.1 1.2 0.1 1.1 1.5 1.0 1.2 1.1 1.5 1.0 1.7	D-FACT EFF  0.360 0.781 0.343 0.834 0.354 0.871 0.373 0.890 0.380 0.883 0.400 0.861 0.418 0.810 0.432 0.794 0.404 0.894 0.405 0.904 0.397 0.927	LOSS COEFF TOT PROF 0.150 0.101 0.109 0.063 0.083 0.045 0.077 0.043 0.084 0.051 0.102 0.069 0.142 0.110 0.156 0.124 0.086 0.058 0.097 0.083 0.080 0.07C	LOSS PARAM TOT PROF 0.021 0.021 0.023 0.013 0.017 0.009 0.016 0.009 0.018 0.011 0.021 0.014 0.029 0.025 0.032 0.012 0.018 0.012 0.020 0.017 0.016 0.014

### EDGES FOR ROTOR 17

# (j) 90 Percent of design speed; reading number 1153

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN 0UT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS BETAM IN OUT -0.3 37.8 -0.3 37.1 -0.3 35.7 -0.3 36.7 -0.3 38.8 -0.3 41.1 -0.3 42.4 -0.3 42.3 -0.2 41.8	59.5 46.7 59.1 46.8 58.7 46.5 56.3 38.6 52.7 23.1	TOTAL TEMP IN RATIO 289.0 1.169 289.0 1.154 288.0 1.131 288.0 1.128 287.8 1.128 287.9 1.131 287.8 1.132 287.8 1.126 287.8 1.126 287.8 1.125	TOTAL PRESS IN RATIO 10.06 1.525 10.10 1.516 10.14 1.461 10.15 1.444 10.14 1.442 10.15 1.418 10.15 1.406 10.15 1.466 10.15 1.468 10.15 1.468
RP 1 2 3 4 5 6 7 8 9 11	ABS VEL 1N OUT 162.1 208.1 166.1 207.9 171.3 197.4 173.5 201.1 173.8 201.9 173.6 203.2 173.0 198.9 173.2 197.6 170.7 207.5 164.5 233.9 161.7 236.5	REL VEL 1N OUT 407.9 291.4 401.1 286.8 372.2 265.9 349.5 245.9 345.6 237.1 341.9 231.0 336.9 219.2 333.3 211.7 307.8 196.3 271.2 189.5 261.2 186.2	MERID VEL IN OUT 162.1 164.5 166.1 165.8 171.3 160.2 173.5 161.2 173.8 159.4 173.0 149.9 173.2 145.8 170.7 153.3 164.5 174.4 161.7 175.1	TANG VEL 1N OUT -0.8 127.4 -0.8 125.4 -0.8 125.3 -0.8 120.2 -0.8 123.7 -0.8 127.2 -0.8 130.7 -0.8 133.3 -0.8 139.7 -0.7 155.9 -0.7 159.1	MHEEL SPEED IN OUT 375.5 368.0 364.3 359.4 329.6 327.5 302.6 303.2 297.9 299.1 293.7 295.4 288.3 290.6 284.0 286.8 255.3 262.3 214.9 230.2 204.5 222.5
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.487 0.584 0.499 0.587 0.517 0.563 0.525 0.577 0.524 0.581 0.522 0.567 0.523 0.563 0.515 0.595 0.496 0.678 0.487 0.686	REL MACH NO IN OUT 1.225 0.818 1.206 0.810 1.123 0.758 1.055 0.697 1.044 0.678 1.032 0.660 1.017 0.625 1.007 0.603 0.929 0.563 0.817 0.549 0.786 0.540	MERID MACH NO (N OUT) 0.487 0.461 0.499 0.468 0.517 0.457 0.524 0.456 0.524 0.453 0.522 0.428 0.523 0.416 0.515 0.440 0.496 0.508		MERID PEAK SS VEL R MACH NO 1.015 1.484 0.999 1.476 0.936 1.506 0.929 1.534 0.918 1.550 0.912 1.550 0.867 1.558 0.842 1.567 0.898 1.555 1.060 1.519 1.082 1.503
RP 1 2 3 4 5 6 7 8 9 10	PERCENT INCI SPAN MEAN 5.00 5.1 10.00 5.3 30.00 7.2 45.00 8.1 47.50 8.3 50.00 8.5 52.50 8.7 55.00 8.8 70.00 9.8 90.00 10.7 95.00 11.1	DENCE DEV \$S 2.3 -0.0 2.3 -0.3 3.0 1.1 3.2 1.2 3.3 1.1 3.3 2.3 3.3 2.9 3.5 2.3 3.6 0.7 3.8 3.0	D-FACT EFF  0.403 0.759 0.400 0.819 0.390 0.872 0.409 0.871 0.424 0.849 0.437 0.849 0.465 0.798 0.483 0.774 0.484 0.820 0.434 0.925 0.421 0.920	LOSS COEFF TOT PROF 0.190 0.138 0.137 0.089 0.093 0.050 0.100 0.061 0.106 0.068 0.122 0.083 0.166 0.128 0.189 0.150 0.165 0.137 0.087 0.074 0.096 0.087	LOSS PARAM TOT PROF 0.040 0.029 0.029 0.019 0.019 0.010 0.020 0.012 0.022 0.014 0.025 0.017 0.034 0.026 0.038 0.030 0.034 0.028 0.018 0.015 0.019 0.017

# TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE EDGES FOR ROTOR 17

### (k) 90 Percent of design speed; reading number 1154

RP 1 2 3 4 5 6 7 8 9 10	RADII IN OUT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS. BETAM IN OUT -0.3 41.5 -0.3 40.2 -0.3 38.5 -0.3 45.7 -0.3 45.5 -0.3 47.3 -0.3 45.9 -0.2 42.8	REL BETAM IN OUT 68.3 55.6 67.4 54.2 64.5 53.6 62.4 49.7 62.1 48.9 61.7 48.4 61.4 48.4 61.1 47.9 58.6 58.3 54.7 23.3 53.6 20.4	TOTAL TEMP IN RATIO 289.7 1.181 289.3 1.169 287.9 1.140 287.7 1.140 287.7 1.142 287.7 1.142 287.7 1.142 287.7 1.142 287.7 1.152 287.7 1.152	TOTAL PRESS IN RATIO 10.07 1.546 10.11 1.548 10.14 1.442 10.14 1.439 10.14 1.431 10.15 1.414 10.15 1.424 10.15 1.482 10.15 1.482 10.15 1.489
RP 1 2 3 4 5 6 7 8 9 11 11	ABS VEL 1N OUT 148.9 209.9 152.2 211.2 157.5 194.1 158.6 196.6 158.5 197.1 158.4 196.7 157.8 194.2 157.5 194.1 156.2 207.1 152.8 232.3 151.0 233.2	REL VEL IN OUT 403.2 278.2 396.2 275.9 365.6 256.2 342.7 223.1 338.4 216.6 334.4 207.6 329.7 198.2 325.5 188.7 300.1 183.7 264.3 187.5 254.5 182.4	MERID VEL 1N OUT 148.9 157.2 152.2 161.4 157.5 151.9 158.6 144.3 158.5 142.4 157.8 131.6 157.5 126.6 156.2 144.1 152.8 172.2 151.0 170.9	TANG VEL 1N OUT -0.7 139.0 -0.7 136.3 -0.7 120.8 -0.7 135.6 -0.7 140.3 -0.7 142.8 -0.7 147.1 -0.7 148.7 -0.7 156.0 -0.6 158.6	WHEEL SPEED IN OUT 374.0 368.5 365.1 360.1 329.2 327.1 305.1 309.4 299.4 295.7 295.4 288.8 291.0 284.1 287.0 255.6 262.6 215.0 250.3 204.3 222.3
ŖР	ABS MACH NO	REL MACH NO	MERID MACH NO		MERID PEAK SS
1 2 3 4 5 6 7 8 9 10	IN OUT 0.445 0.595 0.456 0.593 0.473 0.550 0.477 0.558 0.477 0.558 0.476 0.558 0.474 0.550 0.474 0.550 0.469 0.591 0.459 0.675 0.453 0.675	1.205 0.776 1.186 0.774 1.099 0.727 1.031 0.634 1.018 0.615 1.005 0.589 0.991 0.562 0.979 0.535 0.902 0.525 0.794 0.543 0.764 0.528	IN OUT 0.445 0.458 0.456 0.455 0.473 0.451 0.477 0.404 0.476 0.391 0.474 0.373 0.474 0.359 0.469 0.412 0.459 0.498 0.453 0.495		VEL R MACH NC 1.056 1.527 1.060 1.527 0.965 1.554 0.910 1.598 0.899 1.607 0.871 1.618 0.804 1.611 0.923 1.583 1.126 1.532 1.132 1.511

TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE

#### **EDGES FOR ROTOR 17**

(1) 90 Percent of design speed; reading number 1155

	(1) 00 1	Crecin or act.	D		
RP 1 2 3 4 5 6 7 8 9 10	RADII IN OUT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS BETAM (N OUT -0.3 44.8 -0.3 43.1 -0.3 45.0 -0.3 50.3 -0.3 51.9 -0.3 53.5 -0.3 53.5 -0.2 42.4 -0.2 43.3	REL BETAM IN OUT 70.1 56.7 69.1 55.3 66.3 54.5 64.4 50.0 64.0 49.5 63.7 49.8 63.4 49.8 63.0 49.2 60.5 38.5 56.3 23.7 55.2 20.8	TOTAL TEMP IN RATIO 290.3 1.180 289.7 1.172 287.9 1.150 287.6 1.151 287.5 1.150 287.5 1.149 287.6 1.149 287.5 1.139 287.5 1.127 287.5 1.126	TOTAL PRESS IN RATIO 10.06 1.538 10.11 1.536 10.14 1.445 10.15 1.440 10.15 1.414 10.15 1.408 10.15 1.431 10.15 1.431 10.15 1.431 10.15 1.430 10.14 1.470
RP 1 2 3 4 5 6 7 8 9 10	ABS VEL IN OUT 135.6 206.5 139.8 207.4 144.8 191.9 145.6 196.8 145.4 193.9 145.3 192.1 145.2 192.3 145.2 206.3 143.8 230.6 142.1 230.7	REL VEL IN OUT 398.4 266.7 391.7 266.0 361.0 241.7 336.4 205.9 332.2 198.6 328.5 191.8 324.2 183.8 319.9 175.2 294.7 178.1 259.1 185.8 249.1 179.5	MERID VEL IN OUT 135.6 146.4 139.8 151.5 144.8 140.3 145.6 132.5 145.4 129.1 145.5 123.9 145.3 118.6 145.2 114.4 145.2 139.5 143.8 170.1 142.1 167.8	TANG VEL IN OUT -0.7 145.6 -0.7 141.6 -0.7 145.6 -0.7 145.6 -0.7 145.6 -0.7 145.1 -0.7 154.6 -0.7 152.1 -0.6 155.6 -0.6 158.2	NHEEL SPEED 1N OUT 373.9 368.4 365.2 360.2 329.9 327.8 302.6 303.2 298.0 299.2 293.8 295.8 295.8 295.8 262.9 214.9 230.2 204.0 221.9
RP 1 23 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.404 0.575 0.417 0.580 0.436 0.556 0.436 0.556 0.436 0.542 0.436 0.542 0.435 0.543 0.435 0.548 0.435 0.588 0.431 0.667 0.426 0.668	REL MACH NO IN OUT 1.185 0.742 1.168 0.744 1.081 0.682 1.008 0.561 0.985 0.542 0.972 0.519 0.959 0.495 0.883 0.508 0.7746 0.537 0.7446 0.519	MERID MACH NO UT 0.404 0.408 0.417 0.424 0.436 0.366 0.436 0.350 0.436 0.355 0.436 0.325 0.435 0.325 0.435 0.328 0.431 0.486		MERID PEAK SS VEL R MACH NO 1.080 1.570 1.084 1.565 0.969 1.606 0.910 1.656 0.881 1.653 0.851 1.658 0.816 1.651 0.788 1.642 0.961 1.606 1.183 1.544 1.181 1.519
RP 1 2 3 4 5 6 7 8 9 10	PERCENT INCI SPAN MEAN 5.00 8.6 10.00 8.9 30.00 10.9 45.00 12.3 47.50 12.5 50.00 12.7 52.50 12.9 55.00 13.1 70.00 13.9 90.00 14.3	DENCE SS 5.8 1.1 5.9 0.3 6.8 2.7 7.3 2.5 7.4 2.9 7.5 4.2 7.6 5.2 7.6 5.7 7.7 2.1 7.2 1.3 7.2 3.9	D-FACT EFF  0.468 0.729 0.454 0.759 0.452 0.763 0.522 0.737 0.539 0.729 0.553 0.712 0.572 0.696 0.594 0.689 0.534 0.777 0.421 0.947 0.419 0.926	LOSS COEFF TOT PROF 0.232 0.169 0.204 0.145 0.199 0.143 0.243 0.187 0.254 0.199 0.273 0.220 0.292 0.242 0.303 0.257 0.236 0.205 0.066 0.054 0.096 0.088	LOSS PARAM TOT PROF 0.048 0.035 0.043 0.030 0.039 0.028 0.048 0.037 0.050 0.039 0.053 0.043 0.056 0.046 0.058 0.049 0.049 0.042 0.013 0.011 0.019 0.017

# TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE EDGES FOR ROTOR 17

# (m) 90 Percent of design speed; reading number 1156

RP 1 2 3 4 5 6 7 8 9 10	RADII 1N OUT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS BETAM OUT -0.3 54.6 -0.3 49.6 -0.3 52.7 -0.3 55.3 -0.3 56.6 -0.3 48.9 -0.2 43.6	REL BETAM 1N OUT 72.6 59.5 71.6 57.2 68.7 55.3 66.5 50.3 66.1 50.5 65.7 50.8 65.4 50.6 62.6 37.6 58.2 24.1 57.2 20.9	TOTAL TEMP IN RATIO 291.1 1.201 290.3 1.190 287.8 1.165 287.4 1.161 287.5 1.160 287.4 1.157 287.3 1.155 287.3 1.155 287.3 1.127	TOTAL PRESS IN RAT10 10.05 1.525 10.10 1.521 10.14 1.463 10.15 1.445 10.15 1.430 10.15 1.418 10.15 1.410 10.15 1.445 10.15 1.425
RP 1 25 45 67 8 9 9 11	ABS VEL IN OUT 117.3 204.8 122.1 204.1 128.8 193.1 129.8 196.3 130.3 193.9 130.7 191.7 130.9 191.1 133.0 208.6 133.5 228.7 131.8 230.1	REL VEL 1N OUT 392.2 233.7 385.8 243.5 354.8 219.9 329.5 190.7 325.9 186.1 321.8 179.9 317.8 172.8 313.8 165.5 288.8 173.2 253.5 184.0 243.6 178.2	MERID VEL 1N OUT 117.3 118.8 122.1 131.9 128.8 125.2 129.8 121.7 130.1 118.9 130.3 114.4 130.7 109.3 130.9 105.1 133.0 137.2 133.5 167.9 131.8 166.5	TANG VEL 1N OUT -0.6 166.9 -0.6 155.7 -0.6 147.0 -0.6 156.1 -0.6 156.5 -0.6 157.5 -0.6 157.1 -0.6 155.2 -0.6 158.8	WHEEL SPEED 1N OUT 373.6 368.1 360.4 330.0 327.9 302.3 302.9 298.2 299.3 299.1 291.4 284.7 287.5 255.7 262.8 215.0 230.3 204.3 222.3
RP 1 2 3 4 5 6 7 8 9 10	ABS MACH NO IN OUT 0.544 0.564 0.565 0.584 0.541 0.588 0.557 0.389 0.545 0.391 0.548 0.597 0.599 0.665	REL MACH NO IN OUT 1.160 0.643 1.144 0.674 1.059 0.616 0.984 0.537 0.973 0.524 0.961 0.506 0.949 0.486 0.938 0.466 0.863 0.493 0.758 0.531 0.728 0.515	MERID MACH NO IN OUT 0.347 0.327 0.362 0.365 0.384 0.351 0.388 0.343 0.389 0.334 0.389 0.332 0.390 0.307 0.391 0.296 0.397 0.391 0.399 0.485 0.394 0.482	·	MERID PEAK SS VEL R MACH NO 1.013 1.636 1.081 1.631 0.972 1.673 0.938 1.711 0.913 1.706 0.878 1.697 0.836 1.689 0.803 1.682 1.032 1.633 1.258 1.561 1.263 1.537

#### **EDGES FOR ROTOR 17**

# (n) 80 Percent of design speed; reading number 1157

RP 1 2 3 4 5 6 7 8 9 10	RADII IN OUT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS BETAM 1N OUT -0.3 54.6 -0.3 47.9 -0.3 49.3 -0.3 49.8 -0.3 50.9 -0.3 53.7 -0.3 53.7 -0.3 48.0 -0.2 42.7 -0.2 43.5	REL BETAM IN 0UT 72.7 57.8 68.9 55.2 66.9 49.5 66.6 49.0 66.2 49.3 65.8 49.1 65.8 48.7 62.7 36.9 58.5 23.7	TOTAL TEMP IN RATIO 289.9 1.161 289.1 1.150 287.9 1.128 287.7 1.124 287.9 1.123 287.8 1.121 287.8 1.120 287.7 1.113 287.7 1.102 287.7 1.100	TOTAL PRESS IN RATIO 10.09 1.393 10.13 1.387 10.13 1.355 10.14 1.354 10.14 1.351 10.14 1.352 10.14 1.350 10.14 1.350 10.14 1.350 10.14 1.384 10.13 1.367
RP 1 23 4 5 6 7 8 9 111	ABS VEL IN OUT 103.4 180.5 107.9 178.4 113.5 170.9 114.9 177.2 115.3 176.7 115.4 174.1 115.6 172.8 115.7 172.4 117.4 187.1 117.4 204.5 116.0 204.0	REL VEL IN OUT 348.1 208.2 343.1 220.1 315.0 200.4 293.0 177.7 289.7 173.9 285.9 168.2 282.1 161.6 278.3 154.6 256.0 156.5 224.7 164.1 215.9 158.6	MERID VEL IN OUT 103.4 104.7 107.9 117.2 113.5 114.5 114.9 115.4 115.3 114.0 115.4 109.7 115.6 105.8 115.7 102.0 117.4 125.1 117.4 150.2 116.0 147.9	TANG VEL IN OUT -0.5 147.1 -0.5 134.4 -0.5 126.9 -0.5 134.4 -0.5 135.9 -0.5 136.7 -0.5 138.9 -0.5 138.8 -0.5 140.5	NHEEL SPEED IN OUT 331.9 327.0 325.2 320.7 293.3 291.4 269.0 269.5 266.2 261.1 262.6 256.8 258.8 252.5 255.1 226.9 233.2 191.2 204.8 181.6 197.6
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.306 0.503 0.320 0.500 0.338 0.484 0.342 0.504 0.343 0.495 0.344 0.492 0.344 0.492 0.344 0.490 0.349 0.536 0.349 0.593 0.345 0.592	REL MACH NO IN OUT 1.030 0.580 1.017 0.617 0.937 0.568 0.872 0.505 0.862 0.497 0.839 0.460 0.828 0.440 0.762 0.449 0.669 0.476 0.643 0.460	MERID MACH NO IN OUT 0.506 0.292 0.320 0.329 0.358 0.324 0.342 0.343 0.324 0.343 0.344 0.301 0.344 0.290 0.349 0.359 0.349 0.436 0.345 0.429		MERID PEAK SS VEL R MACH NO 1.013 1.559 1.087 1.562 1.008 1.553 1.004 1.513 0.989 1.513 0.991 1.505 0.915 1.496 0.882 1.487 1.066 1.445 1.280 1.385 1.275 1.362
RP 1 2 3 4 5 6 7 8 9 10 11	PERCENT INC SPAN MEAN 5.00 11.3 10.00 11.5 30.00 13.5 45.00 14.8 47.50 15.0 50.00 15.2 52.50 15.4 55.00 16.2 90.00 16.6 95.00 16.8	10.0 3.7 10.0 4.5 10.0 5.1 9.9 0.6 9.4 1.3	D-FACT EFF  0.561  0.615  0.503  0.653  0.499  0.710  0.536  0.730  0.542  0.728  0.551  0.714  0.571  0.707  0.591  0.698  0.534  0.791  0.412  0.956  0.409  0.931	LOSS COEFF TOT PROF 0.351 0.311 0.306 0.267 0.254 0.225 0.258 0.241 0.263 0.248 0.279 0.265 0.289 0.277 0.302 0.292 0.230 0.227 0.056 0.056 0.094 0.094	LOSS PARAM TOT PROF 0.067 0.059 0.060 0.052 0.049 0.043 0.052 0.048 0.053 0.049 0.055 0.052 0.056 0.054 0.058 0.056 0.048 0.048 0.011 0.011 0.018 0.018

#### EDGES FOR ROTOR 17

# (o) 70 Percent of design speed; reading number 1173

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN 0UT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS BETAM 1N 0UT 0.0 21.0 0.0 20.5 -0.0 21.1 0.0 22.9 0.0 24.6 0.0 26.3 0.0 27.8 0.0 27.5 -0.0 32.2 -0.0 33.8	63.4 56.4 60.3 53.0 57.7 49.1 57.2 48.1 56.8 46.6 56.8 45.7 55.9 44.6 53.2 37.4 49.1 24.1	TOTAL TEMP 1N RAT10 288.6 1.054 288.4 1.051 288.1 1.048 288.2 1.051 288.1 1.055 288.2 1.055 288.2 1.058 287.9 1.057 287.9 1.068	TOTAL PRESS IN RATIO 10.08 1.165 10.13 1.161 10.13 1.164 10.14 1.170 10.14 1.169 10.14 1.169 10.14 1.169 10.14 1.25 10.14 1.25 10.14 1.239
RP 1 2 5 4 5 6 7 8 9 10 11	ABS VEL IN OUT 138.1 158.2 142.1 159.3 146.6 159.6 149.8 163.4 150.1 164.9 148.3 165.4 147.9 164.7 149.9 167.0 148.9 179.4 145.8 197.9 144.0 204.6	REL VEL IN 0UT 322.1 273.5 317.4 269.4 295.5 247.2 280.1 229.7 277.3 227.0 270.6 218.7 267.1 211.5 267.4 207.6 248.7 200.3 222.8 183.4 214.3 179.9	MERID VEL IN OUT 138.1 147.8 142.1 149.2 146.6 148.8 149.8 150.5 150.1 151.6 148.3 150.3 147.9 147.7 149.9 147.8 148.9 159.1 145.8 167.4 144.0 170.0	TANG VEL IN OUT 0.0 56.6 0.0 55.7 -0.0 57.6 0.0 63.7 0.0 65.1 0.0 68.9 0.0 77.8 0.0 82.9 -0.0 113.8	WHEEL SPEED IN OUT 291.0 286.7 283.8 279.9 256.6 254.9 236.7 237.2 233.2 254.1 226.3 227.6 222.5 224.3 221.4 223.6 199.2 204.7 168.5 180.5 158.7 172.6
RP 1 25 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.412 0.462 0.425 0.466 0.439 0.468 0.449 0.479 0.450 0.484 0.444 0.485 0.443 0.482 0.444 0.488 0.446 0.527 0.437 0.582 0.431 0.603	REL MACH NO IN OUT 0.962 0.799 0.949 0.725 0.840 0.674 0.831 0.666 0.811 0.641 0.800 0.619 0.801 0.530 0.642 0.530	MERID MACH NO IN OUT 0.412 0.431 0.425 0.437 0.439 0.442 0.442 0.444 0.444 0.444 0.443 0.432 0.449 0.432 0.446 0.667 0.437 0.493 0.431 0.501		MERID PEAK SS VEL R MACH NO 1.070 1.218 1.050 1.204 1.015 1.207 1.005 1.199 1.010 1.196 1.014 1.175 0.999 1.171 0.986 1.180 1.068 1.171 1.148 1.160 1.180 1.137
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 3.1 10.00 3.2 30.00 4.8 45.00 5.6 47.50 5.7 50.00 5.8 52.50 6.0 55.00 6.0 70.00 6.7 90.00 7.2 95.00 7.1	DENCE DEV SS 0.3 1.7 0.1 1.4 0.7 1.2 0.6 1.6 0.6 1.6 0.6 1.1 0.5 1.1 0.4 1.1 0.1 1.8 -0.2 2.2	D-FACT EFF  0.217 0.821 0.216 0.851 0.228 0.921 0.250 0.907 0.253 0.902 0.268 0.865 0.289 0.809 0.309 0.786 0.283 0.910 0.286 0.904 0.277 0.931	LOSS COEFF TOT PROF 0.070 0.069 0.057 0.056 0.031 0.031 0.042 0.042 0.045 0.045 0.067 0.067 0.101 0.101 0.118 0.118 0.056 0.056 0.082 0.082 0.065 0.065	LOSS PARAM TOT PROF 0.014 0.014 0.012 0.011 0.006 0.006 0.008 0.008 0.009 0.009 0.014 0.014 0.021 0.021 0.024 0.024 0.012 0.012 0.017 0.017 0.013 0.013

TABLE VII. - Continued. BLADE-ELEMENT DATA AT BLADE

# EDGE FOR ROTOR 17

# (p) 70 Percent of design speed; reading number 1174

			7		•	-			
RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN 0U 24.615 24.2 24.044 23.7 21.720 21.5 19.934 19.9 19.632 19.7 19.329 19.4 19.025 19.1 18.720 18.9 16.838 17.3 14.158 15.1 13.447 14.6	16 0.0 80 0.0 75 0.0 08 0.0 41 0.0 74 0.0 08 0.0 02 0.0 66 0.0	28.0	RELL IN 66.3 65.2 62.3 59.6 59.6 59.8 58.4 55.0 51.1	BETAM 0UT 57.0 56.2 53.1 49.6 48.5 46.7 45.9 38.2 22.2	TOTAL IN 288.5 288.4 288.2 288.1 287.8 288.3 287.9 288.2 288.0 287.9	- TEMP RATIO 1.074 1.070 1.063 1.063 1.064 1.066 1.069 1.067 1.072	TOTAL IN 10.10 10.13 10.13 10.13 10.14 10.14 10.14 10.14 10.14	1.200 1.217 1.254
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL IN OU 128.0 156.135.8 156.135.8 157.36.0 156.6 156.0 156.134.7 167.135.2 190.128.2 193	REL T IN .5 318.1 .4 312.4 .0 290.2 .7 272.6 .9 268.8 .7 266.3 .1 262.1 .9 259.4 .8 240.2 .5 211.7	VEL OUT	MER II IN 128.0 130.8 135.0 136.5 136.1 136.6 135.8 136.0 134.7 130.2 128.2	VEL 0UT 136.6 138.0 136.5 134.8 136.2 130.1 127.4 137.6 151.3 151.7	IN 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	VEL OUT 76.3 73.4 73.6 80.4 85.3 88.1 91.7 96.0 115.8 119.9	WHEEL IN 291.2 283.7 256.9 235.9 231.8 228.6 224.2 220.9 198.9 166.9 158.8	SPEED 0UT 286.9 279.8 255.2 7 229.9 226.0 223.2 204.4 178.8 172.8
RP	ABS MACH IN OU 0.381 0.4 0.390 0.4 0.403 0.4 0.407 0.4 0.408 0.4 0.406 0.4 0.406 0.4 0.406 0.4 0.389 0.5 0.382 0.5	IT IN 152 0.948 0.948 0.955 0.966 0.867 0.803 67 0.795 0.775 0.775 89 0.717 158 0.632	ACH NO OUT 0.726 0.720 0.661 0.692 0.577 0.550 0.531 0.510 0.480	MERID M IN 0.381 0.390 0.403 0.408 0.406 0.406 0.406 0.406 0.389 0.389	ACH NO OUT 0.395 0.400 0.597 0.391 0.392 0.396 0.378 0.369 0.444			MERID VEL R 1.067 1.055 1.011 0.985 0.998 0.958 0.937 1.021 1.163	PEAK SS MACH NO 1,258 1,245 1,242 1,226 1,226 1,216 1,209 1,206 1,191 1,161
RP 1 2 3 4 5 6 7 8 9 10	SPAN 5.00 10.00 30.00 45.00 47.50 50.00 52.50 55.00	INCIDENCE IEAN SS 4.8 2.0 5.0 2.0 6.8 2.7 7.9 2.9 8.0 2.9 8.2 2.9 8.4 3.0 9.3 3.1 0.1 3.0 0.4 3.1		D-FACT 0.300 0.291 0.300 0.329 0.335 0.350 0.377 0.398 0.377 0.398	0.815 0.855 0.910 0.878 0.874 0.865 0.775 0.858 0.930 0.932	LOSS C TOT 0.099 0.075 0.048 0.071 0.076 0.128 0.153 0.108 0.071 0.074	OEFF PROF 0.096 0.074 0.071 0.076 0.086 0.128 0.153 0.108	LOSS P TOT 0.020 0.015 0.010 0.014 0.015 0.020 0.031 0.022 0.015	PROF 0.020 0.015 0.015 0.015 0.018 0.026 0.031 0.022 0.015

#### **EDGES FOR ROTOR 17**

# (q) 70 Percent of design speed; reading number 1175

RP 1 2 3 4 5 6 7 8 9 10	RADII IN OUT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS BETAM IN OUT 0.0 39. 0.0 38. 0.0 39. 0.0 41. 0.0 41. 0.0 45. 0.0 47. 0.0 44. 0.0 41.	68.5 56.8 66.0 54.4 66.0 54.4 63.6 4.0 49.3 63.7 48.5 63.4 47.9 63.0 48.1 62.7 47.4 60.2 37.7 65.6 1 23.1	TOTAL TEMP IN RATIO 288.4 1.097 288.5 1.090 288.2 1.082 288.1 1.082 287.9 1.082 288.0 1.083 288.0 1.083 288.0 1.080 288.0 1.077 288.0 1.076	TOTAL PRESS IN RATIO 10.10 1.273 10.13 1.269 10.14 1.249 10.14 1.243 10.14 1.234 10.14 1.231 10.14 1.231 10.15 1.278 10.13 1.278
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL IN OUT 107.9 153.0 111.5 153.8 114.0 148.6 114.9 154.5 114.5 153.9 114.2 151.3 114.4 151.9 114.2 163.6 112.4 182.3 111.1 183.2	REL VEL IN OUT 310.2 223.3 304.7 221.0 280.4 197.2 262.3 177.7 258.8 173.6 255.4 166.1 251.9 158.8 249.1 151.8 229.6 148.5 201.6 148.2 194.0 144.0	MERID VEL IN OUT 107.9 118.2 111.5 121.1 114.0 114.9 115.8 114.5 111.3 114.2 106.0 114.4 102.8 114.2 117.5 112.4 136.3 111.1 135.2	TANG VEL IN OUT 0.0 97.1 0.0 94.9 0.0 101.6 0.0 103.0 0.0 106.0 0.0 111.8 0.0 113.9 0.0 121.1	WHEEL SPEED IN OUT 290.9 286.6 283.6 279.8 256.2 254.5 235.9 236.3 232.1 233.0 228.3 229.6 224.5 226.3 221.3 223.5 199.2 204.7 167.4 179.5 159.1 173.1
RP 1 23 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.320 0.437 0.331 0.441 0.359 0.428 0.342 0.444 0.341 0.445 0.341 0.445 0.339 0.435 0.340 0.437 0.359 0.475 0.334 0.531 0.534	REL MACH NO IN OUT 0.921 0.638 0.905 0.634 0.833 0.567 0.780 0.512 0.770 0.500 0.759 0.478 0.749 0.457 0.683 0.429 0.599 0.432 0.577 0.420	MERID MACH NO IN OUT 0.520 0.558 0.351 0.347 0.359 0.351 0.341 0.352 0.341 0.321 0.359 0.305 0.340 0.296 0.359 0.340 0.354 0.597 0.330 0.394		MERID PEAK SS VEL R MACH NO 1.095 1.331 1.086 1.314 1.008 1.301 1.005 1.277 0.972 1.270 0.928 1.264 0.899 1.259 1.029 1.234 1.213 1.188 1.217 1.171
RP 1 2 3 4 5 6 7 8 9 10	PERCENT INCI SPAN MEAN 5.00 8.2 10.00 8.4 30.00 10.6 45.00 11.9 47.50 12.2 50.00 12.4 52.50 12.6 55.00 12.8 70.00 13.6 90.00 14.2	DENCE SS 5.4 2.4 5.3 1.8 6.4 2.5 7.0 1.9 7.1 1.9 7.1 2.3 7.2 3.5 7.3 3.8 7.4 1.4 7.0 0.7 7.1 3.2	0.402 0.949	LOSS COEFF TOT PROF 0.187 0.182 0.148 0.145 0.141 0.140 0.157 0.157 0.163 0.163 0.183 0.183 0.213 0.213 0.228 0.228 0.176 0.176 0.061 0.061 0.084 0.084	LOSS PARAM TOT PROF 0.037 0.036 0.030 0.029 0.028 0.027 0.032 0.032 0.033 0.033 0.037 0.037 0.042 0.045 0.045 0.037 0.037 0.012 0.012 0.017 0.017

#### **EDGES FOR ROTOR 17**

# (r) 70 Percent of design speed; reading number 1176

RP 1 2 3 4 5 6 7 8 9 10 11	•	ABS BETAM IN OUT 0.0 59.9 0.0 52.5 0.0 45.2 0.0 46.8 0.0 47.4 0.0 49.1 0.0 53.5 0.0 49.2 0.0 43.2 0.0 43.8	REL BETAM IN OUT 72.7 61.5 71.7 59.1 69.2 55.0 67.1 49.6 66.7 48.6	TOTAL TEMP IN RATIO 288.6 1.132 288.4 1.120 288.1 1.093 288.0 1.090 287.9 1.090 288.0 1.090 288.0 1.090 288.0 1.090 288.0 1.086 288.0 1.077 287.9 1.076	TOTAL PRESS IN RATIO 10.11 1.297 10.13 1.283 10.13 1.263 10.14 1.262 10.14 1.254 10.14 1.254 10.14 1.254 10.14 1.280 10.13 1.272
RP 1 23 ~ 567 8 9 10 1 1	ABS VEL IN 0UT 90.7 160.4 94.1 154.8 97.5 148.2 99.2 153.4 99.3 154.1 99.4 152.8 99.7 150.4 99.7 150.0 100.3 161.2 100.2 176.9 99.0 177.9	REL VEL IN OUT 305.1 168.6 299.3 183.5 274.0 182.3 254.8 162.1 251.5 157.8 248.4 151.3 245.4 142.9 241.6 135.3 222.0 133.1 193.9 141.0 186.1 137.3	MERID VEL IN OUT 90.7 80.4 94.1 94.2 97.5 104.5 99.2 105.1 99.3 104.2 99.4 100.0 99.7 89.2 100.3 105.3 100.2 129.0 99.0 128.5	TANG VEL IN OUT 0.0 138.8 0.0 105.1 0.0 111.8 0.0 113.5 0.0 117.9 -0.0 120.6 0.0 122.0 0.0 121.0 0.0 123.1	WHEEL SPEED IN OUT 291.3 287.0 284.2 280.3 256.1 254.5 234.8 235.2 231.1 232.0 227.7 229.0 224.3 226.0 220.1 222.3 198.0 203.5 166.1 177.9 157.7 171.5
RP 1 2 5 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.268 0.452 0.279 0.438 0.289 0.424 0.294 0.440 0.295 0.4438 0.296 0.431 0.296 0.430 0.297 0.464 0.297 0.514 0.293 0.518	REL MACH NO IN OUT 0.902 0.475 0.886 0.519 0.812 0.522 0.756 0.465 0.746 0.453 0.737 0.454 0.728 0.410 0.716 0.388 0.658 0.383 0.575 0.410 0.552 0.400	MERID MACH NO IN OUT 0.268 0.226 0.279 0.266 0.289 0.299 0.295 0.295 0.295 0.295 0.296 0.296 0.296 0.296 0.296 0.297 0.503 0.297 0.375 0.293 0.374		MERID PEAK SS VEL R MACH NO 0.886 1.400 1.001 1.381 1.072 1.354 1.060 1.321 1.050 1.314 1.006 1.309 0.937 1.502 0.895 1.291 1.050 1.258 1.288 1.198 1.298 1.178
RP 1 2 3 4 5 6 7 8 9 10	PERCENT INC SPAN MEAN 5.00 11.3 10.00 11.5 30.00 15.0 47.50 15.0 50.00 15.4 52.50 15.4 55.00 15.7 70.00 16.6 90.00 16.9 95.00 17.2	10.3 1.4 9.8 1.4	D-FACT EFF  0.618 0.587 0.537 0.615 0.463 0.742 0.499 0.764 0.510 0.763 0.531 0.747 0.560 0.725 0.586 0.707 0.547 0.781	LOSS COEFF TOT PROF 0.381 0.372 0.339 0.333 0.210 0.209 0.212 0.212 0.217 0.217 0.236 0.236 0.261 0.261 0.286 0.286 0.237 0.237 0.069 0.069 0.083 0.083	LOSS PARAM TOT PROF 0.068 0.067 0.064 0.063 0.040 0.040 0.042 0.042 0.044 0.044 0.047 0.047 0.050 0.055 0.055 0.055 0.049 0.049 0.014 0.014 0.016 0.016

### EDGES FOR ROTOR 17

### (s) 60 Percent of design speed; reading number 1177

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN 0UT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS.BETAM IN OUT 0.0 61.1 0.0 53.4 0.0 44.4 0.0 46.7 0.0 48.4 0.0 50.7 -0.0 52.9 0.0 49.4 0.0 43.7	69.5 54.6 67.6 49.5 67.2 48.5 66.8 48.4 66.6 48.9 66.2 48.8 63.7 38.3 59.5 23.6	TOTAL TEMP IN RATIO 288.4 1.097 288.3 1.087 288.1 1.066 288.1 1.065 288.1 1.065 288.1 1.065 288.1 1.065 288.1 1.065 288.1 1.065 288.1 1.065 288.1 1.065 288.1 1.065	TOTAL PRESS IN RATIO 10.11 1.216 10.13 1.204 10.13 1.189 10.14 1.189 10.14 1.179 10.13 1.175 10.14 1.182 10.13 1.202 10.13 1.202 10.13 1.202
RP 1 25 4 5 6 7 8 9 111	ABS VEL 1N OUT 74.5 137.6 78.1 132.0 81.4 127.1 82.7 131.2 82.9 131.8 83.1 130.4 83.1 128.9 83.2 128.2 84.0 137.0 83.7 151.6 82.4 152.9	REL VEL 1N OUT 258.7 140.5 254.4 154.4 233.0 156.8 217.1 140.2 214.0 136.3 211.0 130.4 208.9 124.0 206.2 117.4 189.3 113.5 164.6 120.9 158.6 118.3	MERID VEL IN OUT 74.5 66.6 78.1 78.6 81.4 90.7 82.7 91.0 82.9 90.3 83.1 81.6 83.1 81.6 83.2 77.3 84.0 89.1 83.7 110.8 82.4 110.6	TANG VEL IN OUT 0.0 120.4 0.0 106.0 0.0 89.0 0.0 94.5 0.0 96.0 0.0 97.5 0.0 99.8 -0.0 102.2 0.0 104.1 0.0 105.5	WHEEL SPEED IN OUT 247.7 244.1 242.2 238.9 218.3 216.9 200.7 201.1 197.3 198.0 194.0 195.1 191.7 193.2 188.6 190.5 169.7 174.4 141.8 151.9 135.5 147.4
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.220 0.392 0.231 0.377 0.241 0.366 0.244 0.379 0.245 0.381 0.246 0.376 0.246 0.372 0.246 0.370 0.248 0.396 0.247 0.442 0.244 0.446	REL MACH NO IN OUT 0.764 0.400 0.752 0.441 0.689 0.452 0.642 0.405 0.624 0.377 0.618 0.358 0.610 0.339 0.560 0.328 0.469 0.345	MERID MACH NO IN OUT 0.220 0.190 0.231 0.225 0.241 0.262 0.244 0.261 0.246 0.250 0.246 0.223 0.246 0.223 0.248 0.223 0.244 0.323 0.244 0.323		MERID PEAK SS VEL R MACH NO 0.894 1.199 1.006 1.183 1.114 1.157 1.101 1.134 1.090 1.125 1.041 1.117 0.982 1.117 0.929 1.111 1.060 1.080 1.324 1.025 1.343 1.015
RP 1 2 3 4 5 6 7 8 9 10	PERCENT INCI SPAN MEAN 5.00 11.8 10.00 12.0 30.00 14.1 45.00 15.5 47.50 15.7 50.00 16.3 70.00 17.1 90.00 17.5 95.00 18.0	9.0 6.1	D-FACT EFF  0.631 0.592 0.546 0.623 0.455 0.761 0.489 0.774 0.500 0.774 0.521 0.759 0.548 0.739 0.576 0.723 0.547 0.771 0.410 0.950 0.400 0.936	LOSS COEFF TOT PROF 0.373 0.373 0.324 0.324 0.189 0.189 0.198 0.198 0.202 0.202 0.220 0.220 0.242 0.242 0.263 0.263 0.248 0.248 0.063 0.063 0.087 0.087	LOSS PARAM TOT PROF 0.067 0.067 0.061 0.061 0.037 0.037 0.040 0.040 0.041 0.041 0.044 0.044 0.047 0.047 0.050 0.050 0.051 0.051 0.013 0.013 0.017 0.017

### EDGES FOR ROTOR 17

# (t) 50 Percent of design speed; reading number 1178

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RP 1 2 3 4 5 6 7 8 9 10 11	RAD!! 1N OUT 24.615 24.252 24.044 23.716 21.720 21.580 19.934 19.975 19.632 19.708 19.329 19.441 19.025 19.174 18.720 18.908 16.838 17.302 14.158 15.166 13.447 14.630	ABS BETAM IN OUT 0.0 59.8 0.0 44.6 0.0 46.1 0.0 46.8 0.0 48.8 0.0 51.3 0.0 53.2 0.0 49.2 0.0 43.1 0.0 43.9	REL BETAM 1N OUT 73.6 61.3 72.5 59.1 69.9 54.7 67.9 49.3 67.6 48.7 67.3 48.5 66.9 49.0 66.6 48.6 64.1 37.5 59.9 24.0 59.0 20.9	TOTAL TEMP IN RATIO 288.5 1.061 288.2 1.047 288.1 1.046 288.0 1.046 288.0 1.046 288.0 1.045 288.0 1.040 288.0 1.040	TOTAL PRESS IN RATIO 10.11 1.150 10.13 1.132 10.13 1.133 10.13 1.125 10.13 1.125 10.13 1.125 10.13 1.128 10.13 1.128 10.13 1.136
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL IN OUT 60.9 114.8 64.0 110.6 67.0 106.8 68.3 110.7 68.3 110.5 68.5 109.8 68.6 108.0 68.5 107.9 69.0 115.9 69.3 126.9 68.2 127.7	REL VEL IN OUT 216.3 120.1 213.0 131.4 195.4 131.7 181.8 117.6 179.4 114.6 177.1 109.1 174.7 102.8 172.2 97.6 157.9 95.4 138.1 101.5 132.6 98.5	MERID VEL IN OUT 60.9 57.7 64.0 67.4 67.0 76.1 68.3 76.7 68.5 72.7 68.5 72.4 68.6 67.5 68.6 67.5 69.0 75.7 69.3 92.7 68.2 92.0	TANG VEL IN OUT 0.0 99.2 0.0 87.7 0.0 74.9 0.0 79.7 0.0 80.5 0.0 82.6 0.0 84.3 0.0 86.4 0.0 87.8 0.0 86.7 0.0 88.6	WHEEL SPEED IN OUT 207.5 204.5 203.2 200.4 183.6 182.4 168.5 168.8 165.9 166.6 163.4 164.3 160.6 161.9 158.0 159.6 142.0 145.9 119.5 128.0 113.7 123.7
RP 234567891011	ABS MACH NO IN OUT 0.179 0.350 0.189 0.319 0.198 0.310 0.201 0.321 0.202 0.319 0.203 0.313 0.202 0.313 0.204 0.337 0.205 0.371 0.201 0.373	REL MACH NO IN OUT 0.637 0.345 0.628 0.378 0.576 0.382 0.530 0.331 0.523 0.317 0.515 0.298 0.508 0.283 0.466 0.278 0.408 0.297 0.391 0.288	MERID MACH NO IN OUT 0.179 0.166 0.189 0.194 0.198 0.221 0.201 0.223 0.202 0.202 0.202 0.202 0.187 0.204 0.220 0.205 0.271 0.201 0.269		MERID PEAK SS VEL R MACH NO 0.948 1.009 1.054 0.997 1.135 0.977 1.124 0.943 1.057 0.944 0.984 0.937 0.943 0.933 1.096 0.906 1.338 0.865 1.348 0.852
RP 1 2 3 4 5 6 7 8 9 10	PERCENT INC SPAN MEAN 5.00 12.2 10.00 12.4 30.00 14.5 45.00 15.9 47.50 16.1 50.00 16.3 52.50 16.4 55.00 16.7 70.00 17.5 90.00 17.9 95.00 18.3	IDENCE SS 9.4 5.6 9.3 4.1 10.4 2.9 10.9 1.8 11.0 2.2 11.1 4.4 11.2 5.0 11.2 1.2 10.8 1.7 11.0 4.0	D-FACT EFF  0.617 0.601 0.534 0.631 0.454 0.768 0.489 0.781 0.524 0.766 0.555 0.739 0.580 0.728 0.544 0.787 0.409 0.952 0.404 0.937	LOSS COEFF TOT PROF 0.356 0.356 0.310 0.310 0.178 0.178 0.181 0.181 0.192 0.192 0.210 0.210 0.241 0.241 0.257 0.257 0.229 0.229 0.061 0.061 0.085 0.085	LOSS PARAM TOT PROF 0.065 0.065 0.059 0.059 0.034 0.034 0.036 0.036 0.039 0.039 0.042 0.042 0.047 0.049 0.048 0.048 0.012 0.012 0.017 0.017

### TABLE VIII. - BLADE-ELEMENT DATA AT BLADE EDGES FOR STATOR 12

# (a) 100 Percent of design speed; reading number 1143

RP 1 23 4 5 6 7 8 9 10 11	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS. BETAM IN OUT 25.2 0.6 24.1 -1.4 24.1 -3.1 27.6 -2.8 28.4 -3.4 29.0 -3.8 29.8 -4.1 30.7 -3.5 30.7 -3.5 31.5 3.7	24.1 -1.4 24.1 -3.1 27.6 -2.8 28.4 -5.4 29.0 -3.8 29.8 -4.1 30.7 -3.9 28.7 -3.7 30.2 -0.6	TOTAL TEMP IN RAT10 333.3 0.997 331.1 0.997 325.4 0.998 328.1 0.992 327.2 0.992 325.1 0.997 323.9 1.000 324.8 1.000 325.3 0.998 327.4 1.000 327.5 0.999	TOTAL PRESS IN RATIO 14.55 0.934 14.62 0.967 14.46 0.960 14.50 0.951 13.69 0.980 13.32 1.006 13.45 0.999 14.53 0.958 15.45 0.920 15.26 0.868
RP 1 23 4 5 6 7 8 9 10 11	ABS VEL IN OUT 259.7 242.4 262.1 254.6 253.3 248.5 260.4 247.5 252.2 244.5 235.7 243.1 222.8 243.2 268.6 266.3 307.1 314.1 297.6 286.2	REL VEL IN OUT 259.7 242.4 262.1 254.6 253.3 248.5 260.4 247.5 252.2 244.5 235.7 243.1 222.8 243.2 228.1 245.8 268.6 266.3 307.1 314.1 297.6 286.2	MERID VEL IN OUT 235.0 242.4 239.2 254.5 231.2 248.2 230.8 247.2 221.9 244.1 206.1 242.6 193.3 242.6 193.3 242.6 196.2 245.2 235.6 265.8 265.3 314.1 253.7 285.6	TANG VEL IN OUT 110.6 2.7 107.1 -6.4 103.3 -12.0 119.9 -14.3 114.4 -16.1 110.9 -17.2 116.4 -16.9 128.9 -17.2 154.6 -3.2	WHEEL SPEED IN OUT 0.
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.748 0.695 0.759 0.736 0.737 0.725 0.757 0.719 0.732 0.710 0.682 0.707 0.643 0.707 0.658 0.714 0.788 0.781 0.915 0.940 0.882 0.844	REL MACH NO IN OUT 0.748 0.695 0.759 0.736 0.737 0.725 0.752 0.710 0.682 0.707 0.643 0.707 0.658 0.714 0.788 0.781 0.915 0.940 0.882 0.844	MERID MACH NO 1N OUT 0.677 0.695 0.692 0.736 0.673 0.722 0.671 0.719 0.644 0.709 0.596 0.705 0.557 0.705 0.566 0.713 0.691 0.779 0.790 0.939 0.752 0.842		MERID PEAK SS VEL R MACH NO 1.032 0.748 1.064 0.759 1.073 0.737 1.071 0.782 1.100 0.786 1.177 0.748 1.255 0.725 1.250 0.761 1.128 0.788 1.184 0.915 1.126 0.882
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 -2.5 10.00 -3.0 30.00 -3.1 45.00 -0.5 47.50 0.1 50.00 0.5 52.50 1.1 55.00 1.7	DENCE DEV SS -8.9 9.1 -9.4 6.7 -9.5 4.5 -6.9 4.6 -6.3 4.0 -5.9 3.6 -5.3 3.3 -4.7 3.4	D-FACT EFF  0.227 0. 0.193 0. 0.179 0. 0.215 0. 0.201 0. 0.144 0. 0.088 0. 0.103 0.	LOSS COEFF TOT PROF 0.214 0.214 0.104 0.104 0.130 0.130 0.186 0.186 0.165 0.165 0.073 0.073 -0.023 -0.023 0.003 0.003	LOSS PARAM TOT PROF 0.083 0.083 0.039 0.039 0.045 0.045 0.061 0.061 0.053 0.053 0.023 0.023 -0.007 -0.007 0.001 0.001

# EDGES FOR STATOR 12

# (b) 100 Percent of design speed; reading number 1144

RP 1 2 3 4 5 6 7 8 9 10	RAD IN 23.904 23.437 21.542 20.096 19.855 19.614 19.373 19.131 17.676 15.735 15.253	OUT 23.884 23.426 21.603 20.234 20.005 19.779 19.550 19.324 17.973 16.215	IN 34.8 31.7 32.2 34.7 35.4 36.6 39.0 40.0 36.9 36.4	BETAM OUT 1.4 0.6 -2.4 -1.6 -2.1 -2.6 -2.7 -3.1 -3.0 -2.7 0.9	REL 1N 34.8 31.7 32.2 34.7 35.4 36.6 39.0 40.0 36.9 36.4 37.6	BETAM OUT 1.4 0.6 -2.4 -1.6 -2.1 -2.6 -2.7 -3.1 -3.0 -2.7 0.9	TOTA' 1N 349.4 342.9 335.7 334.8 334.4 334.3 334.7 332.4 331.0	TEMP RAT10 0.992 0.998 0.997 0.995 0.997 0.995 0.995 1.000	TOTAL IN 16.52 16.52 16.09 15.79 15.49 15.12 15.05 15.56 15.76	PRESS RATIO 0.967 0.979 0.984 0.968 0.975 0.994 0.994 0.958 0.940
RP 1 2 3 4 5 6 7 8 9	IN 258.4 257.1 243.2 248.3 243.5 236.1 225.8 225.5	VEL 0UT 205.9 209.6 191.9 187.7 182.8 180.6 178.9 191.2	REL IN 258.4 257.1 243.2 248.3 243.5 236.1 225.8 225.5 249.0	VEL 0UT 205.9 209.6 199.8 191.9 187.7 182.8 180.6 178.9 191.2	MERII IN 212.2 218.7 205.9 204.2 198.6 189.6 175.6 172.7 199.2	0UT 205.8 209.6 199.6 191.8 187.6 182.6 180.4 178.7	IN 147.4 135.3 129.5 141.2 140.9 140.7 142.0 145.0 149.4	G VEL 0UT 5.1 2.3 -8.4 -5.3 -7.0 -8.4 -8.5 -9.7	IN 0. 0. 0. 0. 0.	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
10	249.0 271.6 270.1	203.9	271.6	203.9 191.7	218.7 213.8	203.7 191.7	161.1	-9.7 3.0	0. 0.	0. 0.
10	271.6 270.1	203.9	271.6	203.9	218.7	191.7			0. MERID	
RP 1 2 3 4 5 6 7 8 9 10	271.6 270.1 ABS M 0.725 0.729 0.693 0.710 0.695 0.673 0.641 0.639 0.715 0.790	203.9 191.7 NACH NO 0UT 0.569 0.584 0.562 0.539 0.527 0.512 0.506 0.501 0.539 0.576	271.6 270.1 REL M 0.725 0.729 0.710 0.695 0.641 0.639 0.715 0.790	203.9 191.7 IACH NO OUT 0.569 0.584 0.562 0.539 0.527 0.512 0.506 0.501 0.539	218.7 213.8 MERID M 0.595 0.620 0.587 0.584 0.567 0.540 0.498 0.490 0.572 0.636	191.7 ACH NO 0UT 0.569 0.584 0.561 0.539 0.526 0.512 0.501 0.539 0.577 0.540		3.0	0. MERII VEL 1 0.97 0.95 0.97 0.93 1.03 0.95	R09095375926 F 3701808822

#### EDGES FOR STATOR 12

# (c) 100 Percent of design speed; reading number 1145

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	38.0 36.4 36.3 - 38.1 - 39.2 - 40.2 - 42.5 - 44.5 - 41.0 -	TAM RELL DUT IN 1.7 38.0 1.6 36.4 1.0 36.3 -0.9 38.1 1.0 39.2 -1.3 40.2 -1.3 42.5 -1.2 44.5 -0.6 41.0 -1.6 39.6 2.3 40.5	BETAM OUT 1.7 1.6 -1.0 -0.9 -1.3 -1.3 -1.3 -1.2 -0.6 -1.6 -2.3	TOTAL IN 354.5 349.0 340.2 337.8 337.7 338.9 338.6 535.5 332.8 332.8	TEMP RATIO 0.998 1.000 0.999 0.999 0.997 0.995 0.995 0.999 1.001	TOTAL IN 17.34 17.20 16.60 16.30 16.17 16.04 15.80 15.65 16.09 16.10	PRESS RATIO 0.961 0.980 0.980 0.959 0.963 0.973 0.963 0.963 0.963
RP 1 23 4 5 6 7 8 9 10 11	ABS VEL IN OUT 259.9 203.0 256.3 207.4 239.4 190.2 238.9 173.8 236.2 171.3 233.8 169.4 228.9 167.2 225.9 167.1 244.4 173.9 260.4 184.6 264.2 169.4	225.9 167 244.4 173 260.4 184	IN 204.9 7.4 206.3 0.2 192.9 5.8 188.1 1.3 183.1 9.4 178.5 7.2 168.8	ID VEL 0UT 202.9 207.4 190.2 173.8 171.3 169.4 167.2 167.1 173.9 184.5 169.3	TANC IN 159.9 152.0 141.7 147.3 149.3 150.9 154.6 158.4 160.5 165.8 171.6	5 VEL 00T 5.9 5.7 -3.3 -2.7 -3.1 -3.8 -3.8 -3.4 -1.7 -5.2 6.7	WHEEL IN 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.	SPEED OUT
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.724 0.555 0.719 0.572 0.676 0.529 0.677 0.483 0.670 0.476 0.662 0.471 0.646 0.464 0.637 0.464 0.697 0.485 0.751 0.518 0.764 0.473	0.724 0.5 0.719 0.5 0.676 0.5 0.677 0.4 0.670 0.4 0.662 0.4 0.646 0.4 0.637 0.4 0.697 0.4 0.751 0.5	NO MERID! UT IN 555 0.571 572 0.579 529 0.545 483 0.539 476 0.519 471 0.505 464 0.476 464 0.454 485 0.526 518 0.579 473 0.581	MACH NO 0.554 0.552 0.529 0.485 0.471 0.464 0.463 0.485 0.518 0.473				PEAK SS MACH NO 1.090 1.048 0.970 0.989 0.999 1.006 1.026 1.051 1.035 1.007
RP 1 234567891011	PERCENT INCI SPAN MEAN 5.00 10.3 10.00 9.3 30.00 9.1 45.00 9.9 47.50 10.9 50.00 11.7 52.50 13.7 55.00 15.5 70.00 10.6 90.00 6.1 95.00 5.9	SS 3.8 10 2.8 2.7 3.5 6 4.5 6 5.3 7.3 6 4.2 6	DEV D-FACT 0.2 0.448 9.7 0.407 6.5 0.416 6.5 0.476 6.1 0.485 6.1 0.485 6.2 0.481 6.8 0.477 6.0 0.457 9.9 0.511	0. 0. 0. 0. 0. 0. 0. 0. 0.	LOSS CO TOT 0.131 0.068 0.075 0.156 0.153 0.144 0.109 0.080 0.119 0.126 0.219	DEFF PROF 0.131 0.068 0.075 0.156 0.153 0.144 0.109 0.080 0.119 0.126 0.219	LOSS P TOT 0.051 0.026 0.025 0.049 0.046 0.034 0.025 0.034	ARAM PROF 0.051 0.026 0.025 0.051 0.049 0.034 0.034 0.034 0.035 0.034

#### **EDGES FOR STATOR 12**

# (d) 100 Percent of design speed; reading number 1146

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BETAM IN OUT 40.6 3.4 39.3 2.9 38.6 0.6 41.9 -0.4 43.2 -0.4 44.9 -0.7 48.8 -0.7 48.8 -0.7 40.8 0.2 40.8 3.7	39.3 2.9 38.6 0.6 41.9 -0.4 43.2 -0.4 44.9 -0.6 46.9 -0.7 48.8 -0.5 44.0 0.1 40.8 0.2	TOTAL TEMP IN RATIO 356.7 1.003 352.0 1.001 342.1 0.999 340.4 0.996 340.5 0.997 341.0 0.995 341.8 0.993 337.1 0.997 334.2 0.999 333.4 1.000	TOTAL PRESS IN RATIO 17.54 0.954 17.44 0.968 16.70 0.969 16.26 0.958 16.17 0.961 16.05 0.963 15.83 0.975 15.76 0.967 16.37 0.944 16.25 0.922
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL 1N OUT 260.2 202.2 257.5 205.2 237.6 183.3 232.5 166.2 230.4 164.4 228.6 161.8 224.7 161.0 224.1 160.5 237.3 162.5 262.3 175.4 264.7 160.3	REL VEL 1N 0UT 260.2 202.2 257.5 205.2 237.6 183.3 252.5 166.2 230.4 164.4 228.6 161.8 224.7 161.0 224.1 160.5 237.3 162.5 262.3 175.4 264.7 160.3	MERID VEL IN OUT 197.4 201.8 199.4 205.0 185.8 183.3 173.1 166.2 168.1 164.4 161.9 161.8 153.7 161.0 147.6 160.5 170.5 162.5 198.6 175.4 200.5 159.9	TANG VEL 1N OUT 169.5 11.9 163.0 10.4 148.1 1.9 155.3 -1.0 157.6 -1.1 161.4 -1.7 163.9 -1.8 168.6 -1.5 164.9 0.1 171.4 0.7 172.8 10.4	WHEEL SPEED IN OUT 0.
RP 1 23 4 5 6 7 8 9 1 0 1 1 1	ABS MACH NO IN OUT 0.722 0.549 0.719 0.562 0.669 0.507 0.655 0.460 0.648 0.454 0.643 0.447 0.631 0.445 0.628 0.443 0.673 0.451 0.756 0.490 0.764 0.446	REL MACH NO IN OUT 0.722 0.549 0.719 0.562 0.669 0.567 0.655 0.447 0.631 0.445 0.628 0.443 0.673 0.451 0.756 0.490 0.764 0.446	MERID MACH NO 1N OUT 0.548 0.548 0.557 0.562 0.523 0.507 0.465 0.447 0.431 0.445 0.414 0.443 0.484 0.451 0.572 0.496		MERID PEAK SS VEL R MACH NO 1.022 1.153 1.028 1.119 0.986 1.011 0.960 1.042 0.978 1.055 0.999 1.079 1.048 1.096 1.087 1.129 0.953 1.068 0.884 1.046 0.798 1.032
RP 1 2 3 4 5 6 7 8 9 10 11	PERCENT INCI SPAN MEAN 5.00 12.9 10.00 12.1 30.00 11.4 45.00 13.8 47.50 14.9 50.00 16.4 52.50 18.1 55.00 19.8 70.00 13.6 90.00 7.3 95.00 6.2	DENCE DEV SS 6.5 11.9 5.7 11.0 5.0 8.1 7.4 7.0 8.4 7.0 10.0 6.7 13.4 6.8 7.2 7.4 1.0 7.8 -0.1 11.4	D-FACT EFF  0.457 0. 0.427 0. 0.443 0. 0.503 0. 0.507 0. 0.518 0. 0.514 0. 0.513 0. 0.513 0. 0.495 0. 0.544 0.	LOSS COEFF TOT PROF 0.156 0.156 0.109 0.109 0.120 0.120 0.167 0.167 0.158 0.158 0.152 0.152 0.108 0.108 0.101 0.101 0.124 0.124 0.177 0.177 0.243 0.243	LOSS PARAM TOT PROF 0.060 0.060 0.041 0.041 0.042 0.042 0.054 0.054 0.051 0.051 0.048 0.048 0.034 0.034 0.031 0.031 0.036 0.036 0.045 0.045 0.060 0.060

#### **EDGES FOR STATOR 12**

### (e) 100 Percent of design speed; reading number 1147

	,,	.,		01 4051	<b>P</b>	-,			- <b>- ·</b>	
RP 1 2 3 4 5 6 7 8 9 10	RAD IN 23.904 23.437 21.542 20.096 19.855 19.614 19.373 19.131 17.676 15.735 15.253	OUT 23.884 23.426 21.603 20.234 20.005 19.779 19.550 19.324 17.973 16.215	ABS IN 43.5 41.1 45.5 46.9 48.8 50.6 52.3 45.8 40.6	BETAM OUT 4.6 4.7 1.1 -0.5 -0.9 -1.1 -1.4 -0.1 0.9 4.6	REL IN 43.5 41.3 41.1 45.5 46.9 48.8 50.6 52.3 45.2 40.6	BETAM OUT 4.6 4.7 1.1 -0.5 -0.9 -1.1 -1.4 -0.1 0.9 4.6	TOTAL IN 361.8 356.0 343.7 342.9 342.3 342.6 342.4 337.7 333.6 332.8	L TEMP RAT 10 0.993 0.997 1.001 0.995 0.994 0.994 0.995 0.998 1.002	TOTAL IN 17,46 17,45 16.45 16.10 16.00 15.84 15.70 15.60 16.24	PRESS RAT10 0.941 0.952 0.967 0.956 0.959 0.964 0.972 0.975 0.972
RP 1 2 3 4 5 6 7 8 9 11 1	ABS IN 259.7 258.8 232.0 229.2 227.1 223.6 221.1 233.6 263.3 264.1	VEL 0UT 196.6 201.0 176.9 160.9 159.1 157.0 156.5 161.1 173.2 160.6	REL IN 259.7 258.8 232.0 229.2 227.1 223.6 221.1 220.1 233.6 263.3 264.1	VEL 0UT 196.6 201.0 176.9 160.9 159.1 157.0 156.5 155.2 161.1 173.2 160.6	MER II IN 188.2 194.5 175.0 160.7 155.2 147.4 140.5 134.6 162.7 201.0 200.6	VEL 0UT 196.0 200.3 176.9 159.1 156.9 156.5 155.2 161.1 173.2 160.1	TAN IN 178.8 170.7 152.4 163.4 165.8 168.2 170.8 174.1 167.6 170.0	VEL 0UT 15.8 16.5 -1.4 -3.7 -3.8 -0.3 2.9	WHEEL IN 0 0 0 0 0 0 0 0 0 0	SPEED OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 23 4 5 6 7 8 9 10	ABS M IN 0.715 0.719 0.650 0.642 0.636 0.618 0.615 / 0.661 0.759 0.763	0.438 0.438 0.431 0.427 0.444 0.447	REL M IN 0.715 0.650 0.652 0.636 0.626 0.618 0.615 0.665 0.763	ACH NO OUT 0.532 0.548 0.487 0.443 0.438 0.432 0.431 0.427 0.446 0.447	MERID M 1N 0.518 0.540 0.450 0.450 0.435 0.393 0.377 0.461 0.580 0.580	ACH NO OUT 0.530 0.546 0.487 0.443 0.438 0.432 0.431 0.427 0.446 0.446				PEAK SS MACH NO 1,211 1,167 1,038 1,100 1,116 1,134 1,154 1,181 1,089 1,026
RP 1 2 3 4 5 6 7 8 9 10	PERCENT SPAN 5.00 10.00 30.00 47.50 50.00 52.50 55.00 70.00 90.00 95.00	MEAN 15.8 14.1 13.9 17.4 18.6 20.2 21.8 23.3 15.4 6.7 6.0	DENCE SS 9.4 7.7 7.5 11.0 12.2 13.8 15.4 16.9 9.0 0.4 -0.3	DEV 13.1 12.8 8.6 6.8 6.5 6.0 7.2 8.4 12.3	D-FACT 0.485 0.449 0.461 0.531 0.537 0.541 0.539 0.544 0.515 0.502 0.539	EFF 0. 0. 0. 0. 0. 0. 0. 0. 0.	LOSS C TOT 0.203 0.163 0.135 0.173 0.173 0.153 0.107 0.108 0.205 0.248	0EFF PR0F 0.203 0.163 0.155 0.183 0.173 0.153 0.153 0.107 0.108 0.205 0.248	LOSS P TOT 0.078 0.062 0.047 0.060 0.056 0.049 0.033 0.031 0.053 0.061	ARAM PROF 0.078 0.062 0.047 0.060 0.056 0.049 0.033 0.031 0.053

#### EDGES FOR STATOR 12

# (f) 100 Percent of design speed; reading number 1148

RP 1 2 3 4 5 6 7 8 9 10 11	RADII 1N 0UT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BETAM IN OUT 45.2 4.8 42.5 4.4 42.7 0.9 47.2 -0.9 48.4 -1.3 49.9 -1.8 52.2 -1.5 53.6 -1.8 46.1 0.0 40.2 1.2 40.5 4.5	45.2 4.8 42.5 4.4 42.7 0.9 47.2 -0.9 48.4 -1.3 49.9 -1.8 52.2 -1.5 53.6 -1.6 46.1 0.0 40.2 1.2	TOTAL TEMP 1N RATIO 362.3 0.994 356.8 0.997 344.9 1.000 343.7 0.995 343.3 0.994 341.8 0.994 341.8 0.996 342.7 0.994 337.5 0.999 333.7 1.000 333.0 1.002	TOTAL PRESS IN RATIO 17.27 0.937 17.35 0.943 16.40 0.962 16.05 0.953 15.91 0.962 15.58 0.972 15.50 0.975 16.67 0.975 16.46 0.936 16.27 0.919
RP 1 2 3 4 5 6 7 8 9 10	ABS VEL IN OUT 255.1 191.2 256.7 195.5 231.3 175.3 228.7 158.6 225.2 155.4 218.7 153.8 218.0 152.7 232.7 161.6 264.0 176.1 265.0 161.8	REL VEL IN 0UT 255.1 191.2 256.7 195.5 231.3 175.3 228.7 158.6 225.8 157.0 223.2 155.4 218.7 153.8 218.0 152.7 232.7 161.6 264.0 176.1 265.0 161.8	MERID VEL IN OUT 179.7 190.5 189.3 194.9 170.0 175.3 155.4 158.6 149.9 156.9 133.9 153.7 129.2 152.6 161.5 161.6 201.7 176.0 201.6 161.3	TANG VEL IN OUT 181.1 15.9 173.3 15.0 156.8 2.7 167.8 -2.6 168.8 -3.5 170.9 -4.0 175.6 -4.8 167.5 0.1 170.4 3.5 172.1 12.7	MHEEL SPEED IN OUT 0.
RP 1 2 5 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.701 0.516 0.711 0.532 0.647 0.482 0.640 0.436 0.632 0.432 0.625 0.428 0.612 0.423 0.609 0.420 0.659 0.448 0.762 0.492 0.766 0.451	REL MACH NO IN OUT 0.701 0.516 0.711 0.532 0.647 0.482 0.640 0.436 0.632 0.432 0.602 0.423 0.609 0.420 0.659 0.448 0.762 0.492 0.766 0.451	MERID MACH NO IN OUT 0.493 0.514 0.525 0.530 0.475 0.482 0.435 0.436 0.420 0.427 0.375 0.423 0.361 0.420 0.457 0.488 0.582 0.449		MERID PEAK SS VEL R MACH NO 1.060 1.227 1.029 1.184 1.031 1.068 1.020 1.134 1.047 1.141 1.082 1.157 1.148 1.177 1.181 1.198 1.001 1.089 0.873 1.038 0.800 1.027
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 17.5 10.00 15.4 30.00 15.5 45.00 19.1 47.50 20.1 50.00 21.4 52.50 24.7 70.00 15.6	DENCE DEV SS 11.1 13.3 8.9 12.5 9.1 8.4 12.7 6.5 13.7 6.1 15.0 5.6 17.1 5.9 18.3 5.6 9.2 7.4	D-FACT EFF  0.500 0. 0.472 0. 0.474 0. 0.548 0. 0.549 0. 0.553 0. 0.549 0. 0.555 0. 0.510 0.	LOSS COEFF TOT PROF 0.225 0.225 0.199 0.199 0.154 0.154 0.194 0.194 0.177 0.177 0.165 0.165 0.123 0.123 0.111 0.111	LOSS PARAM TOT PROF 0.087 0.087 0.075 0.075 0.054 0.054 0.063 0.063 0.057 0.057 0.052 0.052 0.035 0.035 0.035 0.035

### EDGES FOR STATOR 12

# (g) 100 Percent of design speed; reading number 1179

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BETAM IN OUT 29.5 2. 28.6 1. 27.6 -2.5 31.5 -0.6 32.4 -1. 35.7 -1. 35.1 -2.5 36.0 -2.8 32.5 -2.0 33.4 -0.8 34.9 2.0	7 28.6 1.7 5 27.6 -2.5 6 31.5 -0.6 2 32.4 -1.2 9 33.7 -1.9 6 35.1 -2.5 8 36.0 -2.8 9 32.5 -2.0 8 33.4 -0.8	TOTAL TEMP IN RATIO 338.5 0.999 335.4 0.998 329.0 0.998 331.5 0.991 331.1 0.992 329.1 0.996 328.3 0.997 329.3 0.995 327.8 0.996 328.7 0.999 329.3 0.999	TOTAL PRESS IN RATIO 15.36 0.959 15.36 0.980 15.10 0.987 15.23 0.958 14.88 0.967 14.24 0.999 13.87 1.018 14.08 1.002 14.92 0.980 15.65 0.964 15.41 0.931
RP 12334567891011	ABS VEL IN OUT 256.8 210.7 256.6 218.0 247.9 214.2 257.6 209.5 249.1 203.9 229.6 199.2 218.2 196.3 225.9 196.9 259.4 217.3 290.8 244.3 285.0 227.3	REL VEL IN OUT 256.8 210.7 256.6 218.0 247.9 214.2 257.6 209.5 249.1 203.9 229.6 196.3 225.9 196.9 259.4 217.3 290.8 244.3 285.0 227.3	MERID VEL IN OUT 223.5 210.5 225.3 217.9 219.6 214.0 219.7 209.5 210.3 203.9 191.0 199.1 178.5 196.1 182.7 196.7 218.8 217.2 242.9 244.3 233.7 227.2	TANG VEL IN OUT 126.4 9.8 122.9 6.5 115.0 -9.2 134.6 -2.0 135.5 -4.1 127.4 -6.5 125.5 -8.6 132.9 -9.6 139.3 -7.7 159.9 -3.3 163.2 7.9	WHEEL SPEED IN OUT O.
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.733 0.591 0.736 0.617 0.716 0.612 0.744 0.597 0.717 0.580 0.658 0.566 0.624 0.558 0.647 0.560 0.754 0.623 0.857 0.705 0.837 0.651	REL MACH NO IN OUT 0.733 0.591 0.736 0.617 0.716 0.612 0.744 0.580 0.668 0.624 0.558 0.647 0.560 0.754 0.623 0.857 0.705 0.837 0.651	MERID MACH NO		MERID PEAK SS YEL R MACH NO 0.942 0.880 0.967 0.863 0.974 0.787 0.954 0.911 0.969 0.898 1.043 0.851 1.099 0.833 1.076 0.880 0.993 0.884 1.006 0.932 0.972 0.937
RP 1 2 3 4 5 6 7 8 9 10	PERCENT INCI SPAN MEAN 5.00 1.8 10.00 1.5 30.00 0.5 45.00 3.4 47.50 4.1 50.00 5.2 52.50 6.3 55.00 7.0 70.00 2.0 90.00 -0.1 95.00 0.3	DENCE DEV SS -4.6 11.2 -4.9 9.8 -6.0 5.1 -3.0 6.8 -2.3 6.2 -1.2 5.5 -0.1 4.6 -4.4 5.3 -6.5 6.8 -6.0 9.7	0.355 0. 0.322 0. 0.310 0. 0.359 0. 0.358 0. 0.317 0. 0.292 0. 0.323 0. 0.324 0. 0.301 0.	LOSS COEFF TOT PROF 0.136 0.136 0.066 0.066 0.044 0.044 0.138 0.138 0.115 0.115 0.005 0.005 -0.078 -0.078 -0.009 -0.009 0.063 0.063 0.094 0.094 0.187 0.187	LOSS PARAM TOT PROF 0.052 0.052 0.025 0.025 0.015 0.015 0.045 0.045 0.037 0.037, 0.001 0.001 -0.024 -0.024 -0.034 0.018 0.024 0.024 0.046 0.046

### EDGES FOR STATOR 12

# (h) 90 Percent of design speed; reading number 1151

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BETAM IN OUT 20.9 -0.4 20.6 -1.8 21.2 -4.2 24.5 -3.8 25.0 -3.9 25.4 -3.9 26.4 -4.1 27.3 -4.0 26.1 -4.3 29.4 -2.4 30.8 1.3	29.4 -2.4	TOTAL TEMP IN RATIO 320.3 1.002 318.5 1.000 315.4 0.999 318.1 0.995 318.2 0.995 317.6 0.998 317.6 0.998 317.1 1.000 319.0 0.999 319.8 0.997	TOTAL PRESS IN RATIO 13.40 0.945 13.34 0.975 13.25 0.979 13.57 0.966 13.32 0.978 13.08 0.995 13.09 0.996 13.78 0.971 14.13 0.964 14.27 0.909
RP 1 2 3 4 5 6 7 8 9 11 11	ABS VEL 1N OUT 242.8 215.4 239.8 225.0 230.9 224.2 241.9 232.8 241.6 232.4 233.7 231.0 225.3 231.6 226.0 232.6 251.9 248.4 271.7 277.6 273.9 264.2	REL VEL IN OUT 242.8 215.4 239.8 225.0 230.9 224.2 241.9 232.8 241.6 232.4 233.7 231.6 226.0 232.6 251.9 248.4 271.7 277.6 273.9 264.2	MERID VEL IN OUT 226.7 215.4 224.4 224.9 215.3 223.6 220.1 232.3 218.9 231.9 211.1 230.5 201.9 231.0 200.7 232.1 226.2 247.7 236.7 277.4 235.3 264.1	TANG VEL IN OUT 86.8 -1.5 84.5 -6.9 83.5 -16.5 100.5 -15.2 102.1 -15.9 100.0 -16.5 103.8 -16.1 110.9 -18.5 133.4 -11.6 140.2 6.1	WHEEL SPEED IN OUT 0.
RP 1 2 3 4 5 6 7 8 9 10	ABS MACH NO IN OUT 0.710 0.623 0.703 0.655 0.678 0.657 0.710 0.682 0.709 0.681 0.685 0.677 0.657 0.659 0.681 0.744 0.732 0.807 0.827 0.813 0.782	REL MACH NO 1N OUT 0.710 0.623 0.705 0.655 0.657 0.710 0.682 0.710 0.682 0.677 0.657 0.659 0.681 0.744 0.732 0.807 0.827 0.813 0.782	MERID MACH NO 1N 0UT 0.663 0.623 0.655 0.655 0.655 0.664 0.668 0.675 0.589 0.676 0.585 0.680 0.668 0.730 0.703 0.826 0.698 0.782		MERID PEAK SS VEL R MACH NO 0.950 0.710 1.002 0.703 1.039 0.678 1.059 0.709 1.092 0.685 1.144 0.657 1.156 0.659 1.095 0.744 1.172 0.807 1.122 0.813
RP 1 2 3 4 5 6 7 8 9 10 11	PERCENT INCI SPAN MEAN 5.00 -6.8 10.00 -6.5 30.00 -6.0 45.00 -3.6 47.50 -3.3 50.00 -3.1 52.50 -2.4 55.00 -1.6 70.00 -4.4 90.00 -4.1 95.00 -3.8	DENCE SS -13.2 8.1 -12.9 6.3 -12.4 3.3 -10.0 3.6 -9.7 3.5 -9.5 3.5 -8.8 3.3 -8.0 3.4 -10.8 3.1 -10.4 5.2 -10.1 9.0	D-FACT EFF  0.253	LOSS COEFF TOT PROF 0.194 0.194 0.088 0.080 0.080 0.080 0.112 0.112 0.119 0.119 0.081 0.081 0.015 0.015 0.094 0.094 0.103 0.103 0.258 0.258	LOSS PARAM TOT PROF 0.075 0.075 0.033 0.033 0.028 0.028 0.036 0.036 0.038 0.038 0.026 0.026 0.005 0.005 0.005 0.005 0.027 0.027 0.026 0.026

# TABLE VIII. - Continued. BLADE-ELEMENT DATA AT BLADE EDGES FOR STATOR 12

# (i) 90 Percent of design speed; reading number 1152

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS.BETAM IN OUT 29.0 0.7 27.4 0.4 28.2 -3.3 29.2 -3.0 29.6 -2.5 31.2 -2.5 32.6 -2.4 33.7 -2.5 31.8 -3.5 34.7 -2.8 35.6 -0.2	28.2 -3.3 29.2 -3.0 29.6 -2.8 31.2 -2.5 32.6 -2.4 33.7 -2.5 34.7 -2.8	TOTAL TEMP IN RATIO 330.5 0.996 327.2 0.998 322.0 0.998 322.5 0.997 322.7 0.998 323.4 0.996 323.1 0.997 321.0 0.999 322.5 0.999	TOTAL PRESS IN RATIO 14.63 0.970 14.62 0.978 14.28 0.987 14.38 0.987 14.34 0.979 14.13 0.992 14.05 0.994 14.28 0.987 14.67 0.938
RP 1 2 3 4 5 6 7 8 9 5 5 1 1	ABS VEL 1N OUT 236.2 194.7 234.8 197.1 221.1 191.0 228.6 193.4 229.8 193.1 228.4 193.1 222.5 192.5 220.7 191.6 234.5 198.5 250.9 214.3 257.5 205.0	REL VEL IN OUT 236.2 194.7 234.8 197.1 221.1 191.0 228.6 193.4 229.8 193.1 228.4 193.1 222.5 192.5 220.7 191.6 234.5 198.5 250.9 214.3 257.5 205.0	208.5 197.1 194.8 190.6 199.6 193.2 199.7 192.9 195.3 192.9 187.4 192.4 183.7 191.4 199.2 198.1 206.3 214.1 209.5 205.0	TANG VEL IN OUT 114.3 2.3 108.0 1.2 104.6 -10.9 111.4 -10.2 113.7 -9.3 118.3 -8.5 120.0 -8.0 122.3 -8.4 123.7 -12.2 142.7 -10.6 149.7 -0.6	WHEEL SPEED IN OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3	ABS MACH NO IN OUT 0.677 0.551 0.677 0.561	REL MACH NO IN OUT 0.677 0.551 0.677 0.561 0.639 0.547	MERID MACH NO IN OUT 0.592 0.551 0.601 0.561	·	MERID PEAK SS VEL R MACH NO 0.942 0.798 0.946 0.756
5 6 7 8 9 10	0.639 0.547 0.663 0.554 0.666 0.553 0.661 0.553 0.642 0.551 0.637 0.548 0.683 0.571 0.734 0.619 0.755 0.589	0.663 0.554 0.666 0.553 0.661 0.553 0.642 0.551 0.637 0.548 0.683 0.571 0.734 0.619 0.755 0.589	0.563 0.546 0.579 0.554 0.579 0.555 0.566 0.552 0.541 0.551 0.530 0.548 0.580 0.570 0.604 0.618 0.614 0.589		0.979 0.722 0.968 0.746 0.966 0.759 0.987 0.791 1.027 0.799 1.042 0.812 0.995 0.781 1.038 0.845 0.978 0.867

### EDGES FOR STATOR 12

# (j) 90 Percent of design speed; reading number 1153

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786		IN OUT 33.1 1.4 32.7 1.4 32.1 -2.2 33.2 -2.1 34.2 -2.0 35.2 -1.9 37.6 -1.8 38.9 7 -2.6	325.7 0.996 324.2 0.997 323.8 0.999	TOTAL PRESS IN RATIO 15.35 0.959 15.32 0.976 14.82 0.985 14.67 0.979 14.63 0.974 14.38 0.986 14.27 0.990 14.32 0.983 14.90 0.966 14.84 0.945
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL IN OUT 236.6 183.7 234.8 190.6 217.3 178.8 218.4 173.3 218.2 172.4 218.7 170.8 212.1 168.5 209.6 166.9 218.7 169.4 245.7 190.7 247.6 180.2	REL VEL IN OUT 236.6 183.7 234.8 190.6 217.3 178.8 218.4 173.3 218.2 172.4 218.7 170.8 212.1 168.5 209.6 166.9 218.7 169.4 245.7 190.7 247.6 180.2	MERID VEL IN OUT 198.2 183.6 197.6 190.5 184.1 178.7 182.8 175.2 180.4 172.3 170.7 170.7 168.1 168.4 163.0 166.8 170.6 169.2 194.3 190.6 194.9 180.2	TANG VEL IN OUT 129.2 4.5 126.9 4.6 115.5 -6.7 119.5 -6.4 122.8 -6.2 126.1 -5.8 129.3 -5.4 131.8 -5.5 136.8 -7.8 150.3 -6.8 152.6 1.0	WHEEL SPEED IN OUT 0.
RP 1 2 3 4 5	ABS MACH NO IN OUT 0.670 0.514 0.669 0.536 0.624 0.507 0.628 0.492	REL MACH NO IN OUT 0.670 0.514 0.669 0.536 0.624 0.507 0.628 0.492	MERID MACH NO IN OUT 0.561 0.514 0.563 0.536 0.528 0.507 0.526 0.491		MERID PEAK SS VEL R MACH NO 0.927 0.897 0.964 0.890 0.971 0.802 0.948 0.809
5 6 7 8 9 10	0.628 0.489 0.628 0.484 0.608 0.477 0.600 0.472 0.629 0.481 0.715 0.545 0.721 0.512	0.628 0.489 0.628 0.484 0.608 0.477 0.600 0.472 0.629 0.481 0.715 0.545 0.721 0.512	0.519 0.489 0.513 0.484 0.482 0.477 0.467 0.472 0.491 0.480 0.566 0.544 0.568 0.512		0.955 0.828 0.955 0.847 1.002 0.864 1.023 0.877 0.991 0.884 0.981 0.911 0.924 0.902

#### EDGES FOR STATOR 12

# (k) 90 Percent of design speed; reading number 1154

RP 1 2 3 4 5 6 7 8 9 10	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BE: IN	TAM REL OUT IN 2.0 36.8 2.1 35.8 -1.2 34.9 -1.0 39.3 -1.0 40.3 -1.0 42.0 -0.6 44.0 -0.9 46.0 -1.2 42.3 -1.0 38.1	BETAM OUT 2.0 2.1 -1.2 -1.0 -1.0 -0.6 -0.9 -1.2 -1.0	TOTAL TEMP IN RATIO 342.0 0.992 338.1 0.993 328.2 0.999 327.8 0.996 327.9 0.996 328.4 0.995 328.6 0.994 328.7 0.994 328.7 0.994 326.4 0.997 324.0 1.000 323.6 1.002	TOTAL PRESS IN RATIO 15.57 0.964 15.64 0.970 14.84 0.983 14.62 0.971 14.59 0.969 14.51 0.973 14.35 0.981 14.26 0.985 14.46 0.978 15.04 0.962 14.88 0.941
RP 1 2 3 4 5 6 7 8 9 11 11	ABS VEL IN OUT 235.2 182.4 236.0 187.4 211.6 167.3 209.6 154.9 209.3 153.2 207.6 152.4 203.6 151.1 202.2 150.6 216.1 157.9 243.6 179.5 243.4 163.7	235.2 18. 236.0 18 211.6 16 209.6 15 209.3 15 207.6 15 203.6 15 202.2 15 216.1 15 243.6 17	L MERI UT IN 2.4 188.2 7.4 191.5 7.3 173.7 4.9 162.2 3.2 159.7 1.1 146.5 0.6 140.6 7.9 159.7 9.5 191.7 3.7 190.0	D VEL OUT 182.3 187.3 167.2 154.9 153.2 152.4 151.1 150.6 157.9 179.5 163.7	TANG VEL IN OUT 141.0 6.4 138.0 6.8 121.0 -3.6 132.8 -2.7 135.2 -2.7 139.0 -2.6 141.3 -1.6 145.4 -2.5 145.6 -3.4 150.3 -3.1 152.1 4.2	0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.662 0.507 0.668 0.524 0.604 0.471 0.598 0.436 0.597 0.428 0.579 0.424 0.575 0.423 0.619 0.445 0.708 0.510 0.708 0.463	0.662 0. 0.668 0. 0.598 0. 0.597 0. 0.591 0. 0.579 0. 0.575 0. 0.619 0. 0.708 0.	NO MERID M 507 0.530 524 0.542 471 0.495 436 0.463 431 0.455 428 0.439 424 0.417 423 0.399 445 0.458 510 0.553	MACH NO OUT 0.506 0.524 0.471 0.436 0.431 0.428 0.424 0.423 0.445 0.510 0.463		MERID PEAK SS VEL R MACH NO 0.968 0.970 0.978 0.960 0.963 0.836 0.955 0.899 0.959 0.912 0.988 0.935 1.031 0.949 1.072 0.977 0.989 0.946 0.937 0.912 0.862 0.903
RP 1 23 4 5 6 7 8 9 10 11	PERCENT INC SPAN MEAN 5.00 9.1 10.00 8.7 30.00 7.7 45.00 11.2 47.50 12.0 50.00 13.5 52.50 17.0 70.00 11.9 90.00 4.6 95.00 4.1	SS 2.7 1 2.2 1 1.3 4.8 5.5 7.1 8.8 10.6 5.5 -1.7	DEV D-FACT  0.5 0.445  0.2 0.416  6.3 0.415  6.4 0.471  6.4 0.487  6.4 0.481  6.1 0.465  6.6 0.422  9.2 0.475	0. 0. 0. 0. 0. 0. 0. 0.	LOSS COEFF TOT PROF 0.142 0.142 0.115 0.115 0.080 0.080 0.136 0.136 0.144 0.144 0.129 0.129 0.094 0.094 0.074 0.074 0.096 0.096 0.134 0.134 0.208 0.208	LOSS PARAM TOT PROF 0.055 0.055 0.044 0.044 0.028 0.028 0.044 0.044 0.046 0.046 0.041 0.041 0.029 0.029 0.023 0.023 0.027 0.027 0.034 0.034 0.052 0.052

# TABLE VIII. - Continued. BLADE-ELEMENT DATA AT BLADE EDGES FOR STATOR 12

# (1) 90 Percent of design speed; reading number 1155

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RP 1 2 3 4 5 6 7 8 9 10	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BETAM (N OUT 40.3 3.2 38.8 3.6 39.5 0.2 44.3 -0.2 45.6 -0.3 47.0 -0.5 48.7 -0.7 50.4 -0.8 44.0 0.1 38.4 -0.3 39.2 2.7	38.8 3.6 39.5 0.2 44.3 -0.2 45.6 -0.3 47.0 -0.5 48.7 -0.7 50.4 -0.1 38.4 -0.3	TOTAL TEMP (N RAT(0) 342.5 0.998 339.5 0.997 331.0 0.995 330.9 0.995 330.6 0.995 330.4 0.995 324.1 1.001 323.6 1.002	TOTAL PRESS IN RATIO 15.47 0.961 15.52 0.976 14.66 0.965 14.61 0.966 14.47 0.973 14.35 0.979 14.28 0.982 14.52 0.980 15.11 0.955 14.91 0.942
RP 1 2 3 4 5 6 7 8 9 10 11	202.1 144.1 199.2 143.0 198.4 142.6 214.4 154.6 241.4 171.1	REL VEL IN OUT 228.2 173.7 228.9 179.5 206.5 157.5 207.1 146.6 205.9 144.9 202.1 144.1 199.2 143.0 198.4 142.6 214.4 154.6 241.4 171.1 240.2 158.2	MERID VEL IN OUT 174.0 173.4 178.5 179.1 159.4 157.5 148.2 146.6 144.0 144.9 137.8 144.1 131.6 145.0 126.6 142.6 154.3 154.6 189.2 171.1 186.2 158.0	TANG VEL IN OUT 147.7 9.8 143.3 11.2 131.2 0.7 144.7 -0.5 147.2 -0.8 147.8 -1.2 149.5 -1.7 152.8 -2.1 148.9 0.2 150.0 -0.8 151.8 7.5	WHEEL SPEED IN OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH NO 1N OUT 0.640 0.479 0.645 0.499 0.585 0.440 0.587 0.405 0.572 0.403 0.564 0.400 0.561 0.399 0.613 0.485 0.701 0.485 0.698 0.447	REL MACH NO IN OUT 0.640 0.479 0.645 0.499 0.585 0.440 0.584 0.405 0.572 0.403 0.564 0.400 0.561 0.399 0.613 0.435 0.701 0.485 0.698 0.447	MERID MACH NO IN OUT 0.488 0.478 0.503 0.498 0.452 0.440 0.420 0.440 0.390 0.403 0.372 0.400 0.358 0.399 0.441 0.435 0.549 0.446		MERID PEAK SS VEL R MACH NO 0.996 1.014 1.003 0.992 0.988 0.903 0.989 0.982 1.006 0.998 1.046 1.002 1.087 1.014 1.127 1.039 1.002 0.970 0.904 0.911 0.848 0.903
RP 1 2 3 4 5 6 7 8 9 10	PERCENT INC SPAN MEAN 5.00 12.6 10.00 11.3 45.00 16.2 47.50 17.3 50.00 18.5 52.50 19.9 70.00 13.5 90.00 4.9	6.2 11.8 5.2 11.7 5.9 7.7 9.8 7.2 10.9 7.1 12.1 6.9 13.5 6.7 15.0 6.5 7.1 7.4 -1.4 7.3	D-FACT EFF  0.472 0. 0.434 0. 0.457 0. 0.519 0. 0.526 0. 0.520 0. 0.519 0. 0.522 0. 0.476 0. 0.449 0. 0.488 0.	LOSS COEFF TOT PROF 0.163 0.163 0.124 0.124 0.115 0.115 0.166 0.166 0.167 0.167 0.134 0.134 0.108 0.108 0.092 0.092 0.088 0.088 0.162 0.162 0.208 0.208	LOSS PARAM TOT PROF 0.063 0.063 0.047 0.047 0.040 0.054 0.054 0.054 0.054 0.054 0.034 0.034 0.028 0.028 0.025 0.025 0.042 0.042 0.052 0.052

#### **EDGES FOR STATOR 12**

### (m) 90 Percent of design speed; reading number 1156

RP 1 2 3 4 5 6 7 8 9 10 11	RAD I IN 23, 904 2: 23, 437 2: 21, 542 2: 20, 096 2: 19, 855 2: 19, 614 1: 19, 373 1: 19, 131 1: 17, 676 1: 15, 735 1: 15, 253 1:	OUT 3.884 3.426 1.603 0.234 0.005 9.779 9.550 9.324 7.973 6.215	ABS IN 50.6 45.7 46.2 48.8 49.5 50.7 52.2 53.7 45.4 38.7 39.5	BETAM OUT 4.2 4.0 0.8 -1.0 -1.3 -1.8 -2.0 -2.2 0.4 0.5 3.9	REL IN 50.6 45.7 46.2 48.8 49.5 50.7 52.2 53.7 45.4 38.7 39.5	BETAM OUT 4.2 4.0 0.8 -1.0 -1.3 -1.8 -2.0 -2.2 0.4 0.6 3.9	TOTA IN 349.6 345.4 335.3 333.8 333.5 333.5 333.0 332.5 331.9 328.8 324.3	L TEMP RATIO 0.989 0.994 1.000 0.996 0.996 0.997 0.997 0.997 1.003	TOTAL IN 15.33 15.36 14.83 14.72 14.63 14.31 14.66 15.00	PRESS RATIO 0.944 0.946 0.969 0.964 0.970 0.977 0.982 0.951 0.951
RP 1 2 3 4 5 6 7 8 9 10	220.2 204.2 206.2 203.7 200.4 197.3 195.8 216.0 239.2	0UT 158.9 160.7 153.3 142.5	REL IN 219.2 220.2 204.2 206.2 203.7 200.4 197.3 195.8 216.0 239.2 239.3	VEL 0UT 158.9 160.7 153.3 142.5 141.2 140.6 140.2 139.9 155.0 166.7	MER II IN 139.2 153.8 141.4 135.7 132.2 126.9 116.0 151.6 186.6 184.6	0UT 158.4 160.3 153.3 142.5 141.2 140.5 140.1 139.8 155.0 166.7 154.2	IN 169.3 157.6 147.3 155.2 155.0 155.1 155.9 157.8 153.8 149.6 152.3	1.1 1.6 10.6	IN 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.	0. 0.
	ABS MA			ACH NO	MERID M	ACH NO			MERID	PEAK SS
RP 1 23 4 5 6 7 8 9 10	0.613 0.574 0.582 0.575 0.565 0.556 0.552 0.616 0.694	0UT 0.434 0.441 0.425 0.396 0.593 0.391 0.390 0.435 0.471 0.436	IN 0.606 0.613 0.574 0.582 0.575 0.565 0.5552 0.695	OUT 0.434 0.441 0.425 0.395 0.391 0.390 0.389 0.435 0.436	IN 0.385 0.428 0.398 0.373 0.373 0.358 0.341 0.341 0.536	0.433 0.440 0.425 0.396 0.592 0.391 0.390 0.389 0.435 0.435		·	VEL R 1,138 1,042 1,084 1,050 1,068 1,108 1,159 1,203 0,835	MACH NO 1.174 1.087 1.017 1.063 1.060 1.062 1.070 1.087 1.087 1.006 0.909

#### **EDGES FOR STATOR 12**

#### (n) 80 Percent of design speed; reading number 1157

RP 1 2 3 4 5 6 7 8 9 10 11	RAD11 IN 0UT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BETAM IN OUT 50.7 3.8 45.0 3.6 44.6 1.0 46.1 -0.0 47.8 -0.3 49.1 -0.8 50.6 -1.0 44.6 0.0 38.9 -0.2 39.5 2.9	REL BETAM IN OUT 50.7 3.8 45.0 3.6 44.6 1.0 46.1 -0.0 46.6 -0.4 47.8 -0.3 49.1 -0.8 50.6 -1.0 44.6 38.9 -0.2 39.5 2.9	TOTAL TEMP IN RATIO 336.7 0.989 332.5 0.995 324.6 1.000 323.4 0.997 322.9 0.998 322.5 0.998 322.4 0.998 320.3 0.998 317.0 1.001 316.6 1.002	TOTAL PRESS IN RATIO 14.06 0.956 14.05 0.962 13.73 0.981 13.73 0.967 13.58 0.978 13.51 0.981 13.44 0.985 13.69 0.979 14.03 0.960 13.86 0.953
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL IN OUT 192.8 137.6 192.2 141.3 181.1 138.0 185.4 131.3 184.4 130.5 180.9 130.0 178.9 129.2 177.6 129.0 193.9 138.8 213.2 149.5 211.7 139.2	REL VEL IN OUT 192.8 137.6 192.2 141.3 181.1 158.0 185.4 131.5 184.4 130.5 186.9 130.0 178.9 129.2 177.6 129.0 193.9 138.8 213.2 149.5 211.7 139.2	MERID VEL IN OUT 122.1 137.3 135.8 141.0 129.0 137.9 128.5 131.3 126.8 130.5 121.7 130.0 117.1 129.2 112.7 129.0 138.0 138.8 166.0 149.5 163.2 139.1	TANG VEL IN OUT 149.2 9.2 136.0 8.8 127.1 2.3 133.6 -0.1 133.9 -1.0 135.3 -1.9 137.3 -2.3 136.2 0.1 133.8 -0.6 134.8 7.0	HHEEL SPEED IN OUT 0.
RP 1 2 5 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.539 0.382 0.541 0.388 0.528 0.370 0.525 0.367 0.516 0.366 0.510 0.364 0.506 0.564 0.556 0.426 0.426 0.616 0.396	REL MACH NO IN OUT 0.539 0.382 0.541 0.394 0.514 0.370 0.525 0.367 0.516 0.366 0.510 0.364 0.557 0.364 0.557 0.396 0.616 0.396	MERID MACH NO IN OUT 0.341 0.381 0.382 0.595 0.366 0.370 0.361 0.367 0.347 0.366 0.334 0.364 0.321 0.363 0.396 0.426 0.475 0.396		MERID PEAK SS VEL R MACH NO 1.125 1.048 1.038 0.948 1.070 0.883 1.021 0.916 1.069 0.915 1.104 0.925 1.145 0.941 1.006 0.895 0.900 0.816 0.852 0.804
RP 1 2 5 4 5 6 7 8 9 10 11	PERCENT INCI SPAN MEAN 5.00 23.0 10.00 17.9 30.00 17.4 45.00 18.0 47.50 18.3 50.00 19.2 52.50 20.4 55.00 21.6 70.00 14.2 90.00 5.4	DENCE DEV SS 16.6 12.4 11.5 11.7 11.0 8.5 11.6 7.4 11.9 7.0 12.8 7.1 14.0 6.5 15.2 6.4 7.8 7.4 -0.9 7.3	D-FACT EFF  0.566 0. 0.516 0. 0.478 0. 0.526 0. 0.527 0. 0.517 0. 0.517 0. 0.516 0. 0.484 0. 0.458 0.	LOSS COEFF TOT PROF 0.244 0.244 0.209 0.209 0.116 0.116 0.178 0.178 0.171 0.171 0.134 0.134 0.114 0.114 0.092 0.092 0.110 0.110 0.175 0.175	LOSS PARAM TOT PROF 0.094 0.094 0.079 0.079 0.040 0.040 0.058 0.058 0.055 0.055 0.043 0.043 0.036 0.036 0.028 0.036 0.032 0.032 0.035 0.032

#### EDGES FOR STATOR 12

### (o) 70 Percent of design speed; reading number 1173

RP 1 2 3 4 5 6 7 8 9 10	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BETAM IN OUT 18.1 0.5 17.8 -0.6 18.7 -3.0 20.4 -3.5 20.7 -3.3 21.9 -2.9 23.5 -2.4 24.9 -2.0 24.5 -2.9 28.5 -1.4	REL BETAM IN OUT 18.1 0.5 17.8 -0.6 18.7 -3.0 20.4 -3.5 20.7 -3.3 21.9 -2.9 23.5 -2.4 24.9 -2.0 24.5 -2.0 28.5 -1.4 29.8 0.5	TOTAL TEMP IN RATIO 304.4 1.001 303.2 1.001 301.9 1.000 302.5 1.000 302.9 1.000 303.3 1.000 303.9 0.999 304.9 0.997 304.4 1.000 306.8 1.000 307.5 0.999	TOTAL PRESS IN RATIO 11.75 0.968 11.76 0.991 11.79 0.990 11.84 0.991 11.86 0.991 11.81 0.995 11.85 0.997 12.12 0.991 12.20 0.995
RP 1 23 4 5 6 7 8 9 10 11	ABS VEL 1N OUT 185.3 161.9 184.8 173.2 179.7 174.3 181.6 178.7 182.9 179.8 182.6 180.1 181.0 180.2 182.8 183.5 195.8 194.2 213.0 216.4 219.3 214.7	REL VEL 1N OUT 185.3 161.9 184.8 173.2 179.7 174.3 181.6 178.7 182.9 179.8 182.6 180.1 181.0 180.2 182.8 183.5 195.8 194.2 213.0 216.4 219.3 214.7	169.4 179.8	TANG VEL IN OUT 57.4 1.4 56.4 -1.7 57.7 -9.1 63.3 -10.9 64.6 -10.2 68.2 -9.2 72.1 -7.7 76.9 -6.3 81.1 -9.9 101.7 -5.3 109.2 2.0	WHEEL SPEED IN OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
	ABS MACH NO	REL MACH NO	MERID MACH NO		MERID PEAK SS
RP 1234 567 8 9	IN OUT 0.545 0.473 0.545 0.509 0.530 0.513 0.536 0.527 0.539 0.530 0.538 0.530 0.532 0.530 0.537 0.540 0.578 0.573 0.630 0.641 0.650 0.635	IN 0UT 0.545 0.475 0.545 0.509 0.530 0.513 0.536 0.527 0.539 0.530 0.532 0.530 0.537 0.540 0.578 0.573 0.630 0.641 0.650 0.635	IN OUT 0.518 0.473 0.519 0.509 0.502 0.513 0.502 0.526 0.505 0.529 0.499 0.529 0.488 0.530 0.487 0.540 0.526 0.572 0.554 0.641 0.564 0.635		VEL R MACH NO 0.919

#### EDGES FOR STATOR 12

#### (p) 70 Percent of design speed; reading number 1174

RP 1 23 4 5 6 7 8 9 10 11	RADII 1N 0UT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BETAM IN OUT 25.6 1.6 24.7 0.6 25.3 -2.4 27.4 -2.6 27.9 -2.5 29.0 -2.0 31.0 -1.7 32.6 -1.3 31.6 -2.2 33.6 -0.6	24.7 0.6 25.3 -2.4 27.4 -2.6 27.9 -2.5 29.0 -2.0 31.0 -1.7 32.6 -1.3 31.6 -2.2 33.6 -0.6	TOTAL TEMP IN RATIO 309.9 1.001 308.4 1.001 306.4 1.000 306.2 1.000 307.4 0.998 307.3 0.998 308.1 0.997 307.3 0.998 308.6 1.000 308.7 1.000	TOTAL PRESS IN RATIO 12.40 0.993 12.32 0.994 12.25 0.995 12.26 0.995 12.33 0.988 12.20 0.996 12.17 0.998 12.33 0.992 12.72 0.989 12.73 0.965
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL IN OUT 179.2 149.7 177.8 158.0 171.7 155.2 169.9 154.2 170.8 154.2 174.4 155.0 169.3 154.6 168.2 154.3 179.3 160.9 201.5 183.6 203.7 175.0	REL VEL IN OUT 179.2 149.7 177.8 158.0 171.7 155.2 169.9 154.2 170.8 154.2 174.4 155.0 169.3 154.6 168.2 154.3 179.3 160.9 201.5 183.6 203.7 175.0	MERID VEL IN OUT 161.6 149.6 161.6 157.9 155.2 155.1 150.9 154.1 151.0 154.1 152.5 154.9 145.1 154.5 141.8 154.2 152.7 160.7 167.7 183.6 168.1 175.0	TANG VEL IN OUT 77.4 4.2 74.3 1.6 73.5 -6.4 78.1 -6.9 79.8 -6.6 84.6 -5.3 87.2 -4.5 90.6 -3.6 94.0 -6.2 111.6 -2.0 115.0 3.4	WHEEL SPEED IN OUT 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.
RP 1	ABS MACH NO IN OUT 0.521 0.432	REL MACH NO IN OUT 0.521 0.432	MERID MACH NO IN OUT 0.470 0.432		MERID PEAK SS VEL R MACH NO
2545678911	0.519 0.458 0.502 0.451 0.496 0.448 0.499 0.450 0.493 0.449 0.489 0.449 0.524 0.468 0.592 0.536 0.599 0.510	0.519 0.458 0.502 0.451 0.496 0.448 0.499 0.450 0.493 0.449 0.493 0.449 0.489 0.448 0.524 0.468 0.592 0.536 0.599 0.510	0.471 0.458 0.453 0.451 0.441 0.448 0.441 0.450 0.445 0.450 0.423 0.449 0.412 0.448		0.926 0.521 0.978 0.519 0.999 0.502 1.021 0.502 1.020 0.518 1.015 0.558 1.065 0.582 1.088 0.604 1.053 0.594 1.095 0.652 1.041 0.654

#### EDGES FOR STATOR 12

### (q) 70 Percent of design speed; reading number 1175

RP 1 23 4 5 6 7 8 9 10 11	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BETAM IN OUT 35.4 2.9 34.3 2.5 36.1 -0.6 38.0 0.5 38.6 0.3 40.5 0.2 42.3 0.1 44.2 0.1 40.7 -0.2 37.8 -0.2 38.5 2.4	REL BETAM 1N OUT 35.4 2.9 34.3 2.5 36.1 -0.6 38.0 0.5 38.6 0.3 40.5 0.2 42.3 0.1 44.2 0.1 40.7 -0.2 37.8 -0.2 38.5 2.4	TOTAL TEMP IN RATIO 316.5 0.998 314.3 1.000 311.8 0.999 311.7 0.998 311.4 0.999 311.8 0.998 311.9 0.997 312.0 0.997 311.0 0.999 310.0 1.000	TOTAL PRESS IN RATIO 12.85 0.981 12.86 0.990 12.64 0.993 12.66 0.984 12.65 0.986 12.50 0.992 12.48 0.993 12.65 0.987 12.95 0.979 12.87 0.966
R 1 2345678901	ABS VEL 1N OUT 169.9 133.4 170.2 139.3 160.3 131.1 163.9 127.9 163.9 127.3 162.4 126.6 158.8 125.8 158.5 125.2 170.9 132.9 190.2 149.5 190.3 139.1	REL VEL IN OUT 169.9 133.4 170.2 139.3 160.3 131.1 163.9 127.9 163.9 127.3 162.4 126.6 158.8 125.8 158.5 125.2 170.9 132.9 190.2 149.5 190.3 139.1	MERID VEL 1N OUT 138.5 133.2 140.6 139.2 129.5 131.1 129.1 127.9 128.1 127.9 123.6 126.6 117.4 125.8 113.7 125.2 129.5 132.9 150.2 149.5 148.9 138.9	TANG VEL 1N OUT 98.5 6.7 96.0 6.2 94.4 -1.3 101.0 1.1 102.3 0.5 106.9 0.3 110.5 0.3 111.5 -0.4 116.7 -0.4 118.5 5.8	WHEEL SPEED IN OUT 0.
RP 1 2 3 4	ABS MACH NO IN OUT 0.488 0.380 0.490 0.398 0.462 0.376	REL MACH NO IN OUT 0.488 0.380 0.490 0.398 0.462 0.376 0.473 0.366	MERID MACH NO IN OUT 0.397 0.379 0.405 0.398 0.374 0.376 0.373 0.366	•	MERID PEAK SS VEL-R MACH NO 0.962 0.692 0.990 0.680 1.012 0.660
5 6 7 8 9	0.473 0.366 0.474 0.365 0.469 0.363 0.458 0.360 0.457 0.359 0.495 0.381 0.555 0.431 0.556 0.400	0.474 0.365 0.469 0.363 0.458 0.360 0.457 0.359 0.495 0.381 0.555 0.431 0.556 0.400	0.370 0.365 0.357 0.363 0.339 0.360 0.328 0.359 0.375 0.381 0.439 0.431 0.435 0.400		0.990 0.691 0.994 0.697 1.024 0.716 1.071 0.725 1.101 0.749 1.026 0.729 0.995 0.710 0.933 0.705

#### EDGES FOR STATOR 12

# (r) 70 Percent of design speed; reading number 1176

4 5 6 7 8	RADII IN 0UT 23.904 23.884 23.437 23.426 21.542 21.602 20.096 20.23 19.855 20.009 19.614 19.779 19.373 19.559 19.131 19.32 17.676 17.979 15.735 16.211 15.253 15.786	IN 56.5 48.9 41.9 44.6 44.2 45.9 0 48.5 45.5 45.9 39.4	BETAM OUT 4.8 4.6 1.9 1.5 1.1 0.9 0.7 0.6 1.2 0.7	REL	BETAM OUT 4.8 4.6 1.5 1.1 0.9 0.7 0.6 1.2	TOTAL TEMP IN RATIO	TOTAL PRESS IN RATIO 13.11 0.959 13.00 0.970 12.78 0.988 12.80 0.980 12.79 0.978 12.64 0.988 12.58 0.991 12.71 0.989 12.58 0.991 12.71 0.989 12.97 0.974 12.88 0.965
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL 1N OUT 168.8 118. 164.9 120. 157.7 122. 161.2 118. 161.5 116. 159.3 116. 155.8 115. 164.5 115. 166.3 123. 183.7 133. 184.0 124.	4 168.8 5 164.9 5 157.7 0 161.2 8 161.5 0 159.8 155.8 2 154.5 7 166.3 183.7	OUT 118.4 120.5 122.5 118.0 116.8 116.0 115.3	MER II IN 93.1 108.3 117.5 116.8 115.7 110.7 103.2 98.3 115.6 141.9	OUT 118.0 120.1 122.5 118.0 116.8 116.0 115.3 115.2 123.7 133.6	TANG VEL 1N OUT 140:9 9.9 124.3 9.8 105.3 4.1 111.1 3.0 112.6 2.3 114.5 1.5 119.2 1.3 119.5 2.5 116.6 1.7 118.0 9.2	0. 0. 0. 0. 0. 0.
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH N IN OUT 0.477 0.33 0.468 0.34 0.452 0.34 0.463 0.33 0.464 0.33 0.458 0.33 0.447 0.32 0.443 0.32 0.480 0.35 0.535 0.38 0.537 0.35	IN 3 0.477 0 0.468 9 0.452 6 0.463 3 0.463 0 0.458 9 0.447 8 0.443 4 0.480 4 0.535		MERID M IN 0.263 0.307 0.336 0.338 0.318 0.296 0.282 0.334 0.413	ACH NO OUT 0.332 0.339 0.348 0.336 0.333 0.330 0.329 0.328 0.354 0.355		MERID PEAK SS VEL R MACH NO 1,268 1,028 1,109 0.882 1,043 0.735 1,010 0.762 1,047 0.784 1,118 0.803 1,171 0.823 1,070 0.792 0,941 0.714 0.875 0.708
RP 1 2 3 4 5 6 7 8 9 10 11	SPAN ME 5.00 28 10.00 14 45.00 15 47.50 15 50.00 17 52.50 19 55.00 21 70.00 15 90.00	NCIDENCE AN SS .8 22.4 .7 8.3 .4 9.0 .9 9.5 .4 11.0 .8 13.4 .5 15.1 .5 9.1 .5 -0.4	DEV 13.3 12.8 9.4 8.9 8.5 8.3 8.1 8.0 8.5 8.3	D-FACT 0.598 0.532 0.446 0.485 0.495 0.491 0.490 0.456 0.430	EFF 0. 0. 0. 0. 0. 0. 0. 0.	LOSS COEFF TOT PROF 0.283 0.283 0.216 0.216 0.088 0.088 0.146 0.146 0.157 0.157 0.157 0.157 0.074 0.074 0.074 0.074 0.148 0.148 0.198 0.198	LOSS PARAM TOT PROF 0.109 0.109 0.082 0.082 0.031 0.031 0.047 0.047 0.051 0.051 0.044 0.044 0.030 0.023 0.023 0.023 0.021 0.021 0.038 0.038 0.049 0.049

#### EDGES FOR STATOR 12

# (s) 60 Percent of design speed; reading number 1177

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RP 1 23 4 5 6 7 8 9 10 11	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BETAM IN OUT 57.8 5. 50.0 5. 41.3 2. 42.9 2. 43.6 1. 45.3 1. 47.7 0. 49.9 0. 46.2 1. 39.4 0.	IN OUT 57.8 5.3 57.8 5.3 50.0 5.1 41.3 2.2 642.9 2.0 643.6 1.6 45.3 1.3 45.3 1.3 47.7 0.9 47.9 0.7 49.9 0.7 49.9 0.7 49.9 0.7 49.9 0.7 3 46.2 1.3 9 39.4 0.9	TOTAL TEMP IN RAT [0 316.3 0.989 313.5 0.994 307.4 1.001 307.0 0.998 306.9 0.998 306.9 0.998 306.9 0.998 306.9 0.998 306.4 0.998 304.4 1.001 304.3 1.001	TOTAL PRESS IN RAT10 12.29 0.968 12.19 0.978 12.05 0.995 12.05 0.985 12.05 0.985 12.00 0.987 11.95 0.989 11.91 0.992 11.98 0.991 12.18 0.981 12.13 0.972
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL IN OUT 144.3 101.4 140.0 103.7 135.2 104.6 137.8 101.5 138.1 100.3 135.9 100.0 133.6 98.7 132.1 98.3 141.1 104.8 157.1 114.9 158.0 105.7	REL VEL IN OUT 144.3 101.4 140.0 103.7 135.2 104.6 137.8 101.5 135.9 100.0 133.6 98.7 132.1 98.3 141.1 104.8 157.1 114.9 158.0 105.7	90.0 103.3 101.7 104.6 100.9 101.5 100.0 100.2 95.6 100.0 90.0 98.6 85.1 98.3 97.7 104.8 121.4 114.9	TANG VEL IN OUT 122.2 9.4 107.3 9.3 89.2 4.0 93.9 3.6 95.3 2.9 96.7 2.2 98.8 1.5 101.0 1.1 101.9 2.4 99.7 1.9 101.2 7.6	WHEEL SPEED IN OUT 0.
RP 1 2 3 4 5 6 7 8 9 10 11	ABS MACH NO IN OUT 0.412 0.288 0.401 0.296 0.391 0.300 0.399 0.292 0.399 0.288 0.393 0.287 0.386 0.283 0.382 0.282 0.409 0.302 0.459 0.332 0.461 0.305	REL MACH NO IN OUT 0.412 0.288 0.401 0.296 0.391 0.300 0.399 0.298 0.393 0.287 0.386 0.283 0.382 0.282 0.409 0.302 0.459 0.332 0.461 0.305	IN OUT 0.219 0.287 0.258 0.294 0.294 0.300 0.292 0.289 0.276 0.287 0.260 0.283 0.246 0.282 0.283 0.302 0.354 0.332		MERID PEAK SS VEL R MACH NO 1.314 0.909 1.147 0.771 1.029 0.626 1.006 0.648 1.003 0.656 1.046 0.666 1.096 0.682 1.154 0.700 1.073 0.678 0.946 0.612 0.869 0.608
RP 1 2 3 4 5 6 7 8 9	PERCENT INCI SPAN MEAN 5.00 30.1 10.00 22.9 30.00 14.1 45.00 14.8 47.50 15.3 50.00 16.8 52.50 18.9 55.00 20.9 70.00 15.7 90.00 5.9 95.00 5.3	DENCE SS 23.7 13.8 16.4 13.3 7.7 9.7 8.4 9.4 8.9 9.0 10.4 8.6 12.5 8.3 14.5 8.0 9.3 8.7 -0.4 8.5 -1.0 11.8	0.599 0. 0.524 0. 0.445 0. 0.476 0. 0.488 0. 0.484 0. 0.489 0. 0.489 0. 0.458 0. 0.426 0.	LOSS COEFF TOT PROF 0.289 0.289 0.211 0.211 0.102 0.102 0.147 0.147 0.159 0.159 0.131 0.131 0.110 0.110 0.079 0.079 0.079 0.079 0.140 0.140 0.205 0.205	LOSS PARAM TOT PROF 0.111 0.111 0.080 0.080 0.036 0.036 0.048 0.048 0.051 0.051 0.041 0.041 0.035 0.025 0.023 0.025 0.023 0.023 0.036 0.036

#### EDGES FOR STATOR 12

(t) 50 Percent of design speed; reading number 1178

RP 1 2 3 4 5 6 7 8 9 10 11	RADII IN OUT 23.904 23.884 23.437 23.426 21.542 21.603 20.096 20.234 19.855 20.005 19.614 19.779 19.373 19.550 19.131 19.324 17.676 17.973 15.735 16.215 15.253 15.786	ABS BETAM IN OUT 56.5 5.8 49.0 5.2 41.4 1.9 43.0 1.9 45.7 1.6 45.7 1.1 48.3 0.5 50.3 0.5 46.0 1.3 39.5 0.9 40.2 5.0	REL BETAM 1N OUT 56.5 5.8 49.0 5.2 41.4 1.9 43.0 1.9 43.7 1.6 45.7 1.1 48.3 0.9 50.3 0.5 46.0 1.3 39.5 0.9 40.2 5.0	TOTAL TEMP IN RATIO 308.0 0.992 306.0 0.996 301.4 0.999 301.4 0.999 301.5 0.999 301.5 0.999 301.5 0.999 300.9 0.999 299.5 1.001 299.4 1.001	TOTAL PRESS IN RATIO 11.63 0.978 11.57 0.985 11.47 0.993 11.48 0.988 11.47 0.988 11.44 0.990 11.40 0.993 11.38 0.994 11.43 0.993
RP 1 2 3 4 5 6 7 8 9 10 11	ABS VEL IN OUT 120.6 85.6 117.5 87.7 113.4 88.5 116.1 85.3 115.6 84.4 114.5 83.7 111.8 83.0 111.0 82.7 119.4 88.4 131.4 97.1 131.7 89.2	REL VEL IN OUT 120.6 85.6 117.5 87.7 113.4 88.5 116.1 85.3 115.6 84.4 114.3 83.0 111.0 82.7 119.4 88.4 131.4 97.1 131.7 89.2	MERID VEL IN OUT 66.5 85.2 77.1 87.4 85.0 88.4 84.9 85.3 83.6 84.3 79.8 83.7 74.3 83.0 71.0 82.7 82.9 88.4 101.4 97.1	7.9 75.1 7.9 79.3 2.8 79.9 2.3 81.9 1.6 83.5 1.4 85.4 0.8 85.9 2.0 83.5 1.5 85.0 7.7	WHEEL SPEED IN OUT 0.
RP 1 23 4 5 6 7 8 9 10 11	0.347 0.246 0.339 0.252 0.329 0.256 0.338 0.247 0.336 0.244 0.332 0.242 0.324 0.240 0.322 0.239	REL MACH NO IN OUT 0.347 0.246 0.359 0.256 0.358 0.247 0.356 0.244 0.332 0.242 0.324 0.322 0.256 0.384 0.282 0.385 0.259	MERID MACH NO IN OUT 0.191 0.244 0.222 0.251 0.247 0.256 0.247 0.243 0.244 0.232 0.216 0.240 0.206 0.239 0.241 0.256 0.297 0.282 0.294 0.258		MERID PEAK SS VEL R MACH NO 1.280 0.749 1.134 0.640 1.040 0.550 1.004 0.553 1.048 0.567 1.118 0.580 1.166 0.596 1.066 0.574 0.957 0.514 0.883 0.512
RP 1 2 3 4 5 6 7 8 9 10	PERCENT INCI SPAN MEAN 5.00 28.8 10.00 21.9 30.00 14.3 45.00 14.9 47.50 15.4 50.00 17.2 52.50 19.6 55.00 21.3 70.00 15.6 90.00 6.0 95.00 5.6	DENCE DEV SS 22.4 14.3 15.5 13.3 7.9 9.4 8.5 9.3 9.0 9.0 10.8 8.5 13.2 8.5 14.9 7.9 9.2 8.7 -0.3 8.5 -0.7 12.6	D-FACT EFF  0.585	LOSS COEFF TOT PROF 0.280 0.280 0.202 0.202 0.091 0.091 0.154 0.154 0.155 0.155 0.141 0.141 0.102 0.102 0.083 0.083 0.087 0.087 0.132 0.132 0.194 0.194	LOSS PARAM TOT PROF 0.107 0.107 0.076 0.076 0.032 0.032 0.050 0.050 0.050 0.050 0.050 0.050 0.032 0.032 0.026 0.026 0.025 0.025 0.034 0.034 0.048 0.048

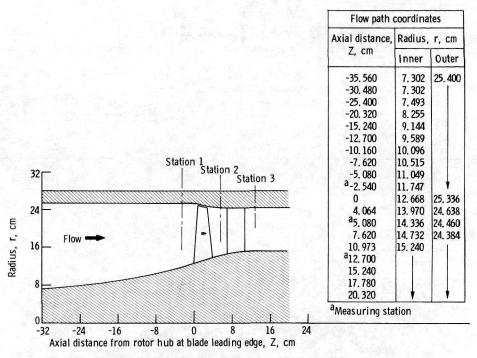


Figure 1. - Flow path for stage showing axial location of instrumentation.

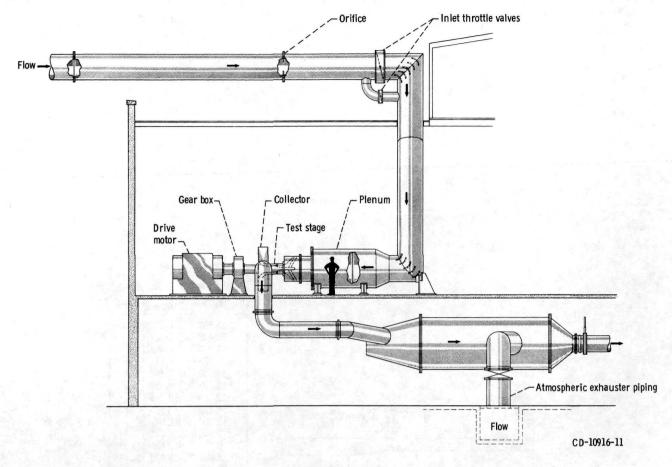


Figure 2. - Test facility.

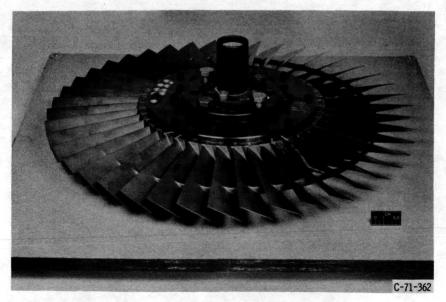


Figure 3. - Test rotor (rotor 14).

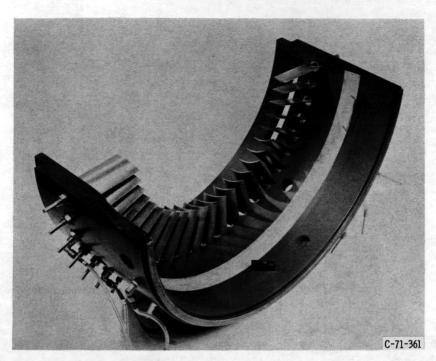
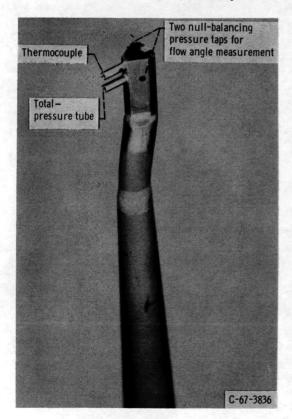


Figure 4. - Test stator (stator 10).



(a) Combination total pressure, total temperature, and flow angle probe (double barrel).



(b) Static-pressure probe.

Figure 5. - Survey probes.

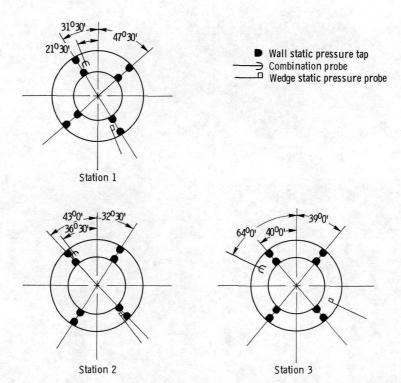


Figure 6. - Circumferential location of measurements (looking downstream; clockwise rotation).

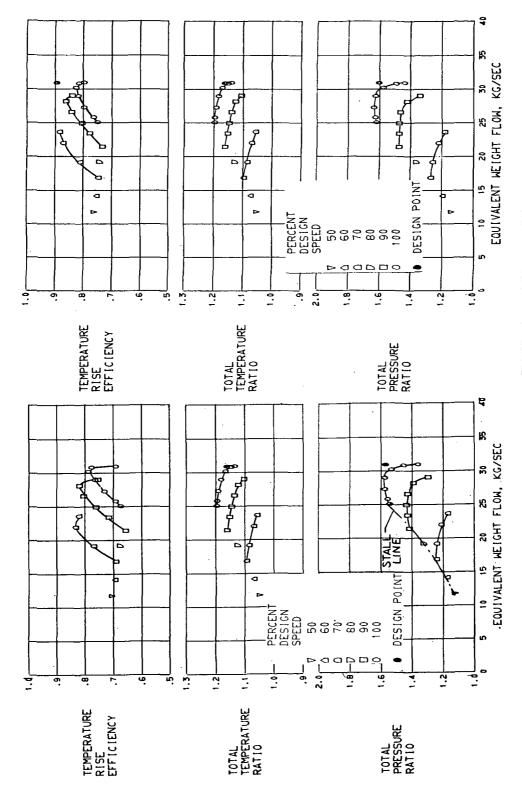
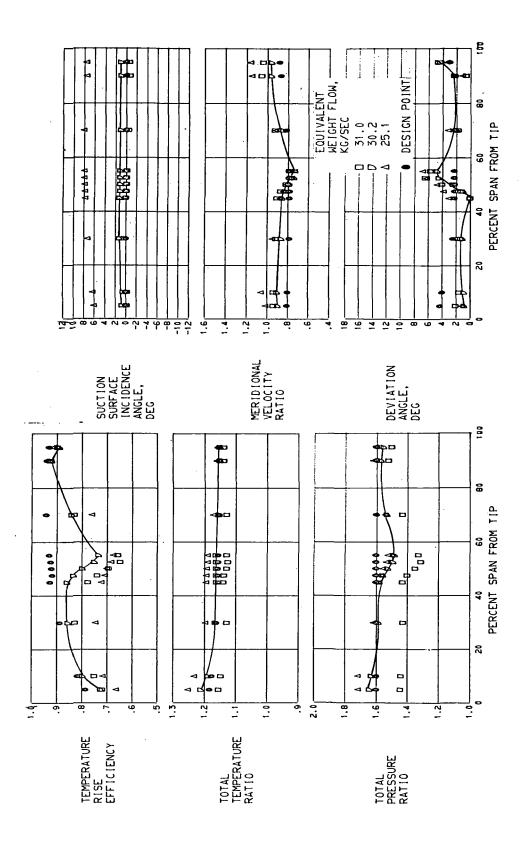


FIGURE 7. - OVERALL PERFORMANCE FOR STAGE 17-12.

FIGURE 8 - OVERALL PERFORMANCE FOR ROTOR 17.



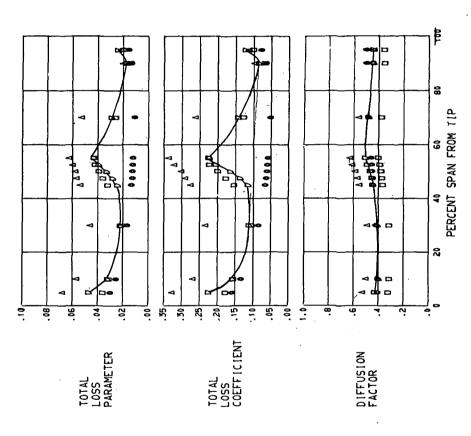


FIGURE 9. - RADIAL DISTRIBUTION OF PERFORMANCE FOR ROTOR 17. 100 PERCENT DESIGN SPEED.

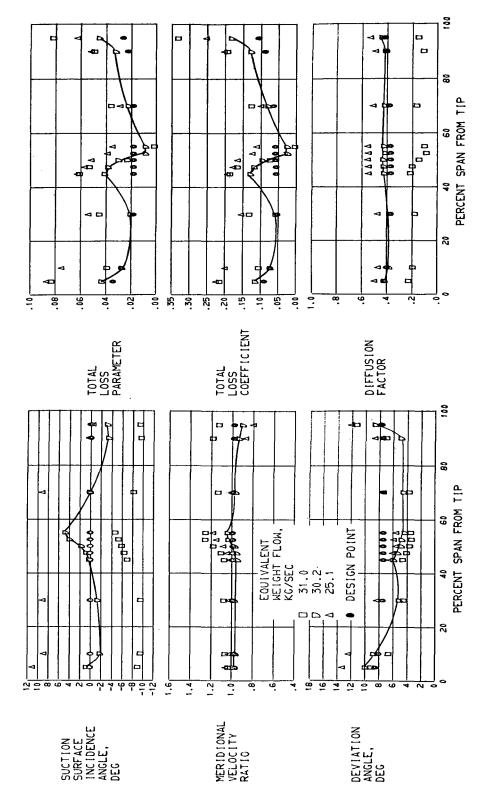
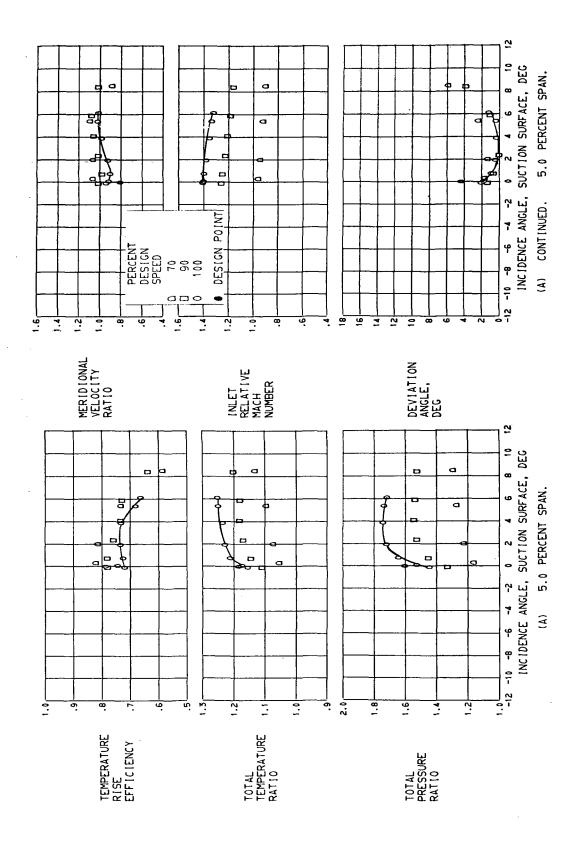


FIGURE 10. - RADIAL DISTRIBUTION OF PERFORMANCE FOR STATOR 12. 100 PERCENT DESIGN SPEED.





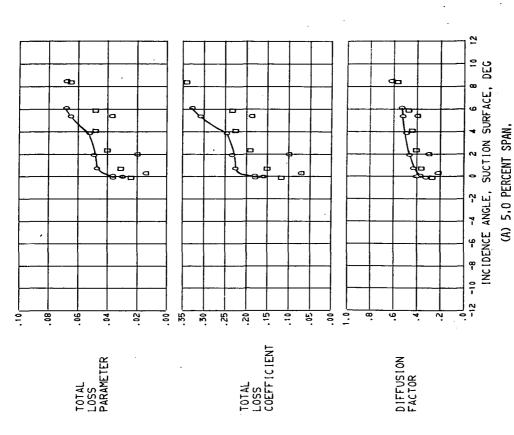
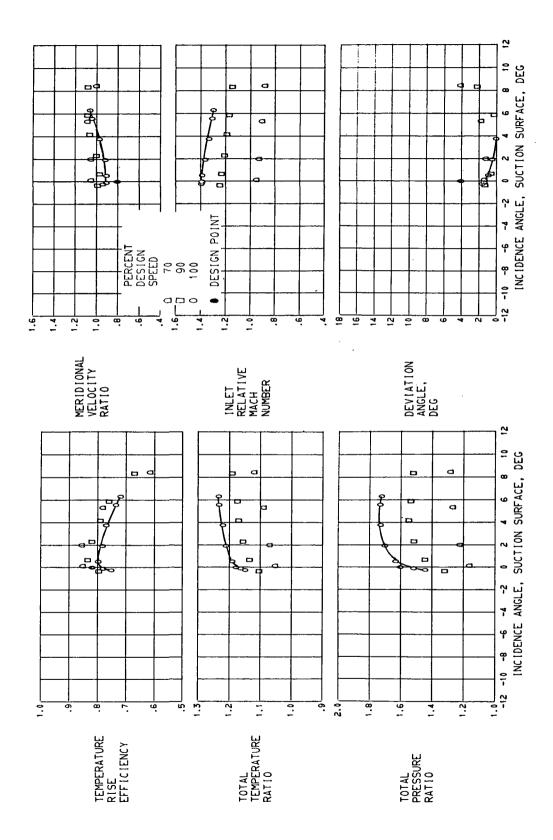


FIGURE 11. - BLADE ELEMENT PERFORMANCE FOR ROTOR 17.



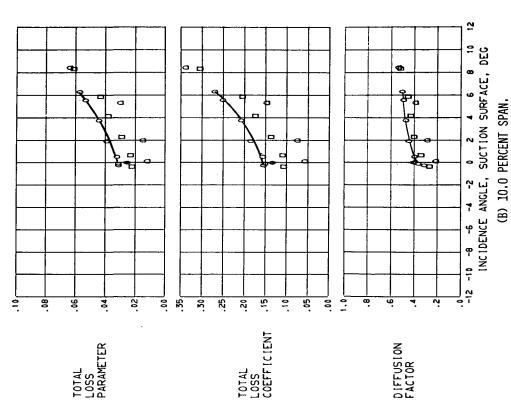
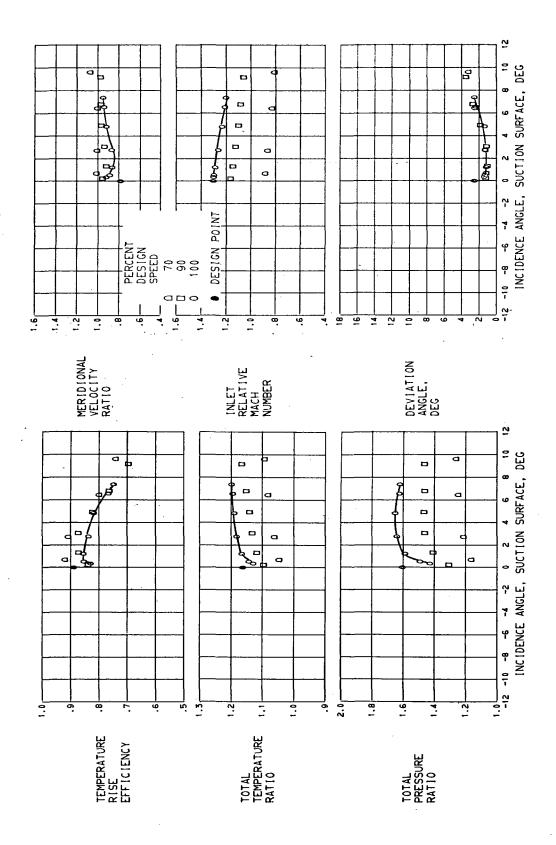


FIGURE 11. - BLADE ELEMENT PERFORMANCE FOR ROTOR 17.



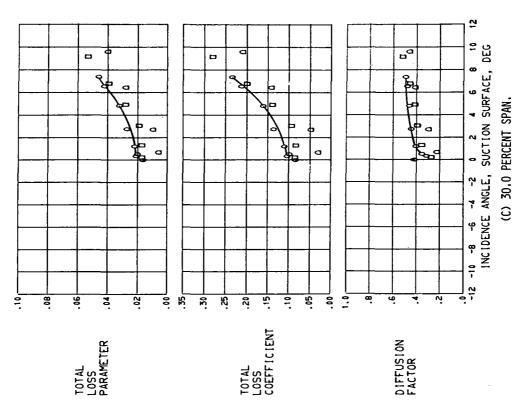
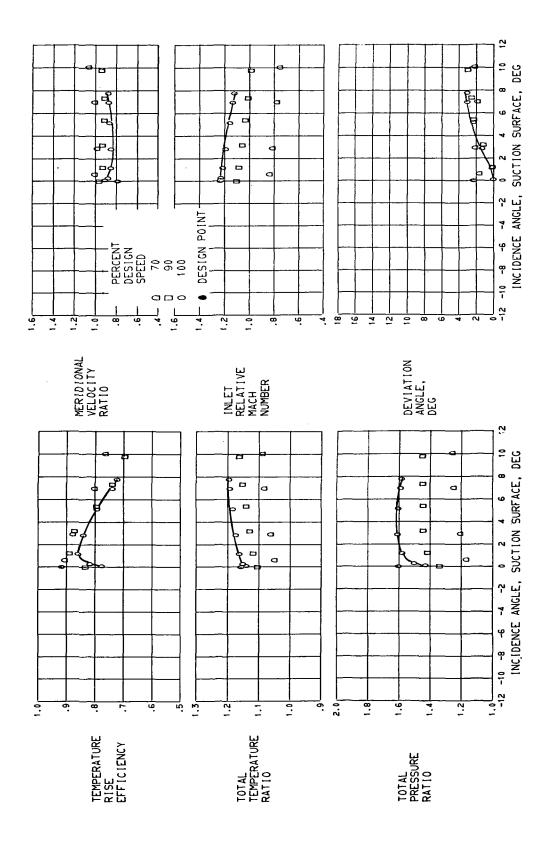
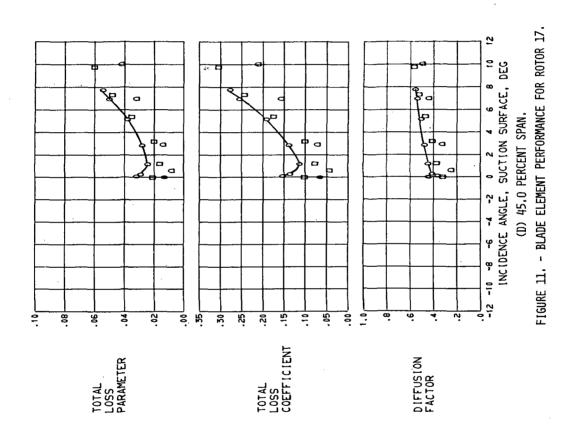
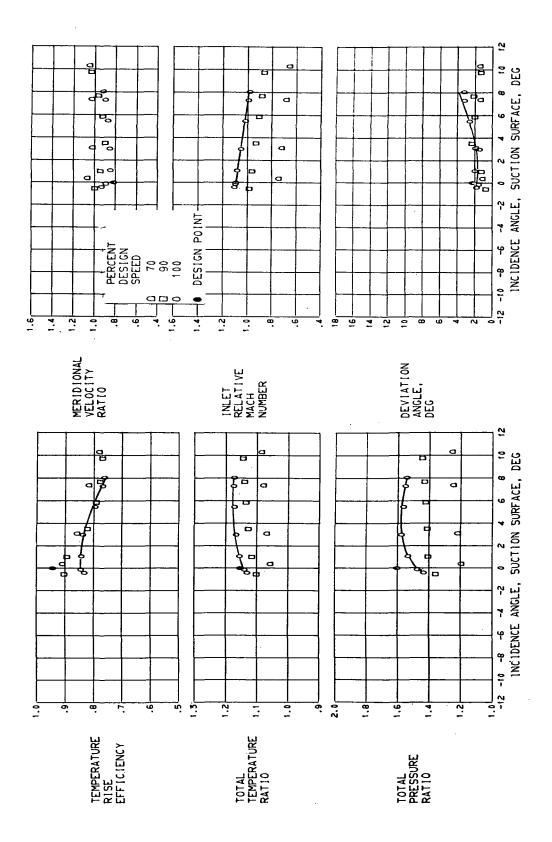
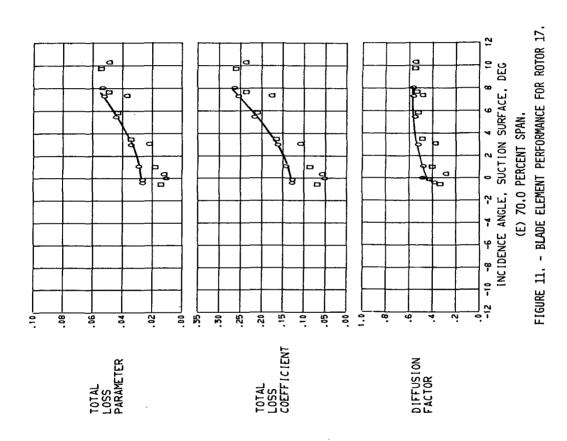


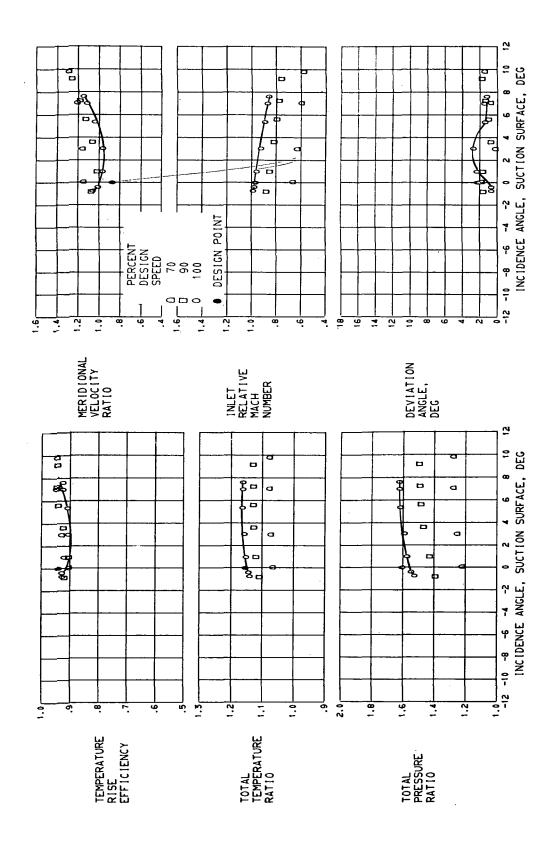
FIGURE 11. - BLADE ELEMENT PERFORMANCE FOR ROTOR 17.

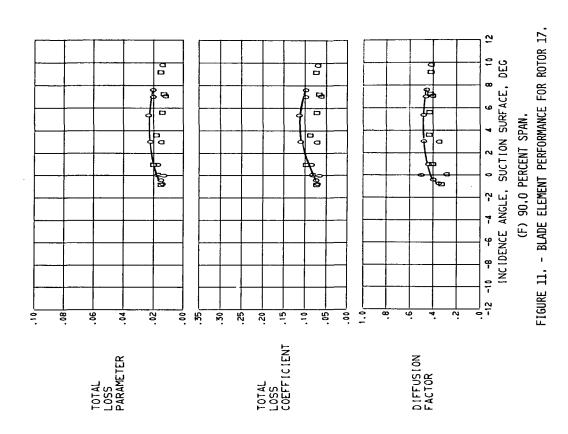


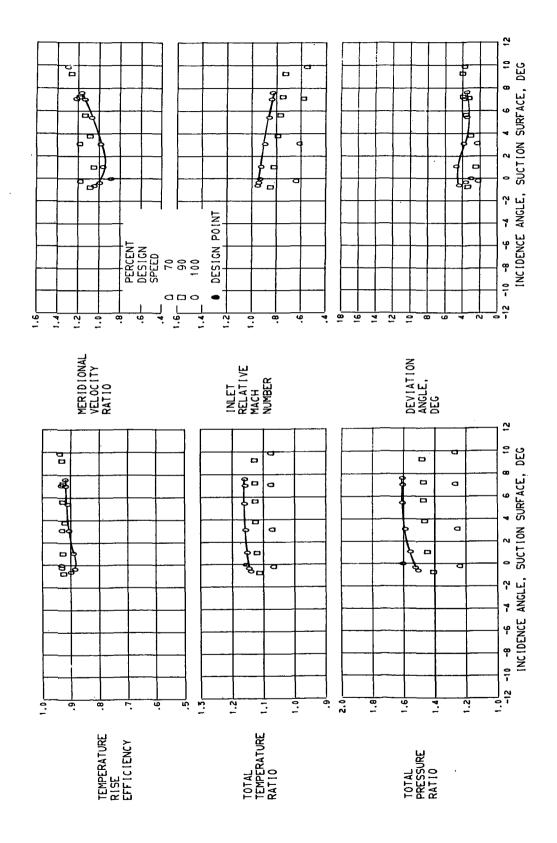


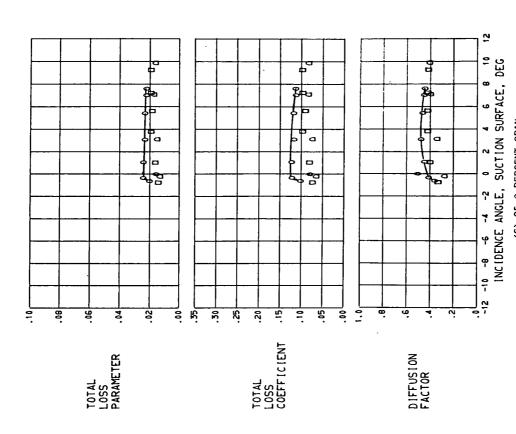












(G) 95.0 PERCENT SPAN. FIGURE 11. - BLADE ELEMENT PERFORMANCE FOR ROTOR 17.

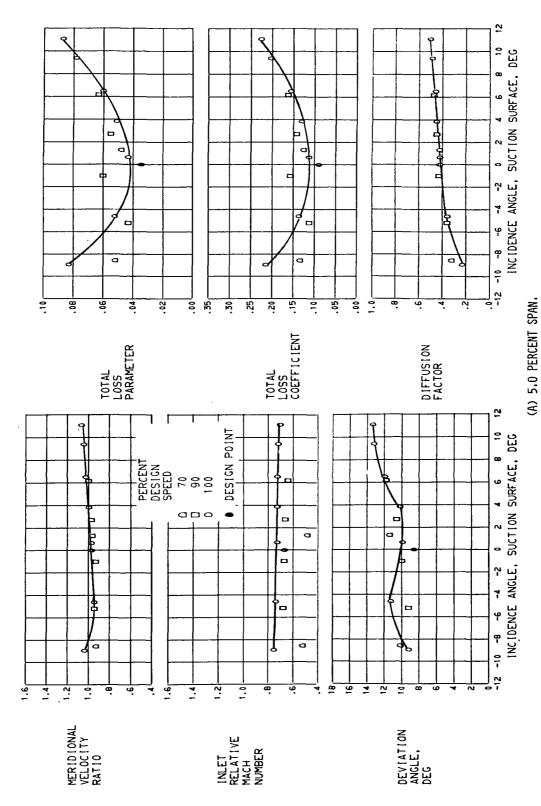


FIGURE 12, - BLADE ELEMENT PERFORMANCE FOR STATOR 12,

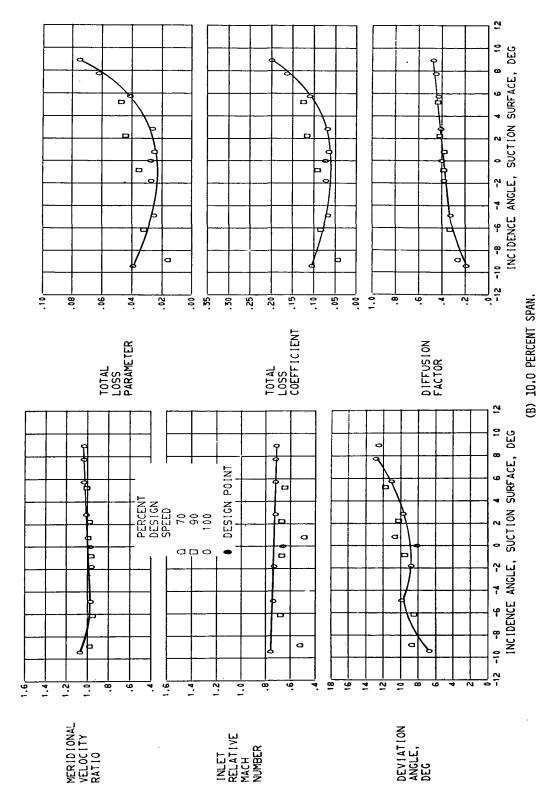


FIGURE 12. - BLADE ELEMENT PERFORMANCE FOR STATOR 12.

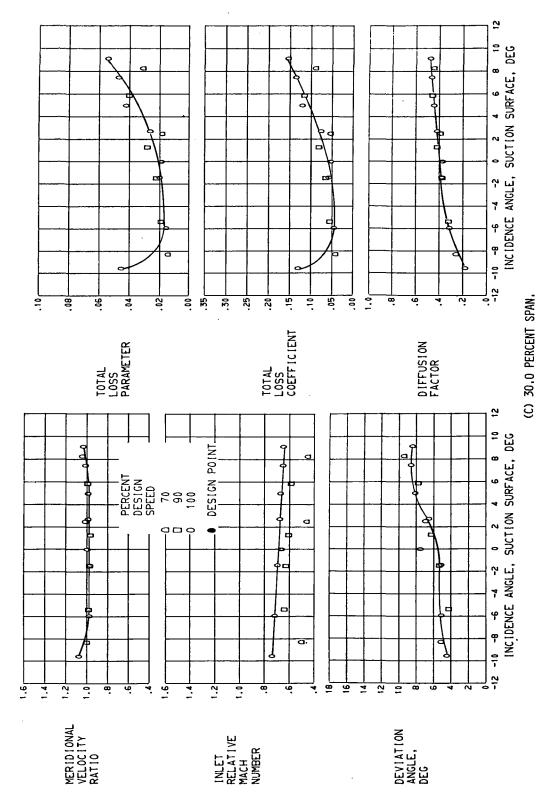


FIGURE 12, - BLADE ELEMENT PERFORMANCE FOR STATOR 12,

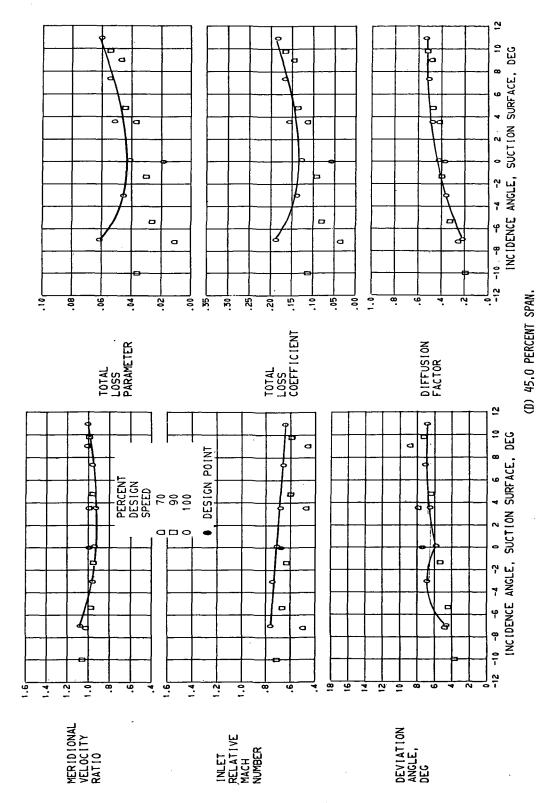


FIGURE 12. - BLADE ELEMENT PERFORMANCE FOR STATOR 12.

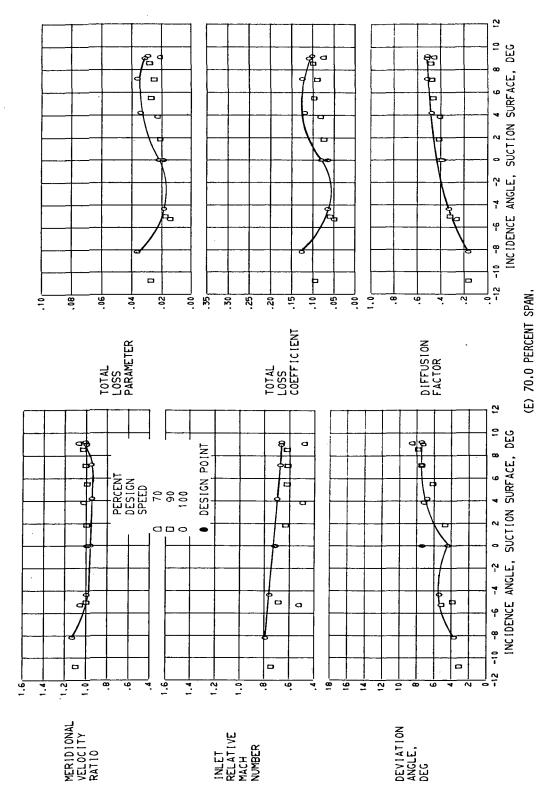


FIGURE 12. - BLADE ELEMENT PERFORMANCE FOR STATOR 12.

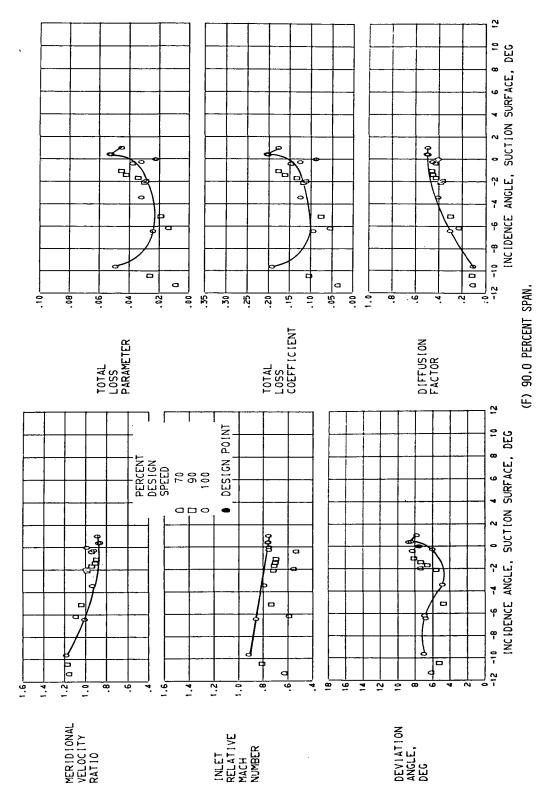


FIGURE 12. - BLADE ELEMENT PERFORMANCE FOR STATOR 12.

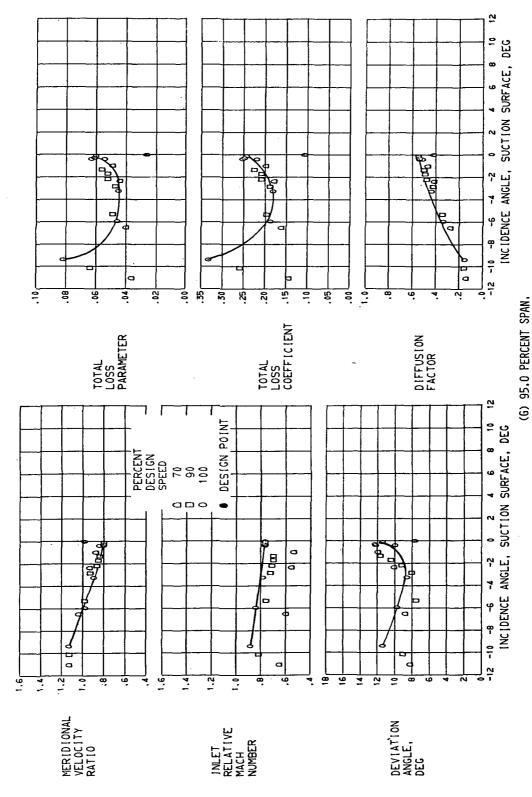


FIGURE 12, - BLADE ELEMENT PERFORMANCE FOR STATOR 12.

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